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Appendix A

COMMON BLOCK AND VARIABLE DEFINITION

Table A.1 presents common block definition and use information for all common blocks in the ABC-ART program. For each common block, the routines where variables in the block are initially defined, redefined, and used are shown.

In Tables A.2, A.3, and A.4 alphabetical lists of all variables in the Fleet Accounting, Airframe Manufacturer, and Air Carrier Modules, respectively, are presented. For each variable, the mode type (real, integer, etc.), the means by which values are assigned (input, calculation, etc.), the dimension if an array, the parameters referenced by each array subscript (aircraft type, year, etc.), the common block membership, the routine where a value was initially assigned, and the definition of the variable are given. The codes used in these tables are given in footnotes to the tables.

Table A.1
COMMON BLOCK DEFINITION AND USAGE

BLOCK NAME	ROUTINES WHERE COMMON BLOCK IS		
	DEFINED	REDEFINED	USED
ACCASHF	CASHFLW		CASHFLW, INRR, INTROR
ACCINCO	REVENUE		REVENUE, CASHFLW, INTROR
ACCOST	CASHFLW		CASHFLW
ACCUMS	SHARE		BET, SHARE, CURVES
CASHFLO	CASHFLW		CASHFLW, INRR, INTROR
COMPCST	COMPCOS		COMPCOS, CASHFLW, INTROR
COSTOT	ACCOST, COMPCOS		ACCOST, COMPCOS, COSTPR
CSHFLO	ACCOST		ACCOST, COMPCOS
CUMOUT	ACCOST		ACCOST, COSTPR
DELAY	RDTE		RDTE
DELIVER	PLANT		PLANT, REVENUE, COMPCOS (inactive)
DEMAND	INPLANT		INPLANT, PLANT
IN			OPLIFE, REPAY, DEPSUB, NETSUB, SUM, CFSUB, DCFSUB, OUTPUT, TAX
INCOME	REVENUE		REVENUE, CASHFLW, INTROR
INDICES	BET		BET, AMORITZ, MODS, BUYS, PLOTSGL, INPLANT
LIFETIM	INPLANT		PLANT, INPLANT
MARKET	3ET		BET, BUYS
MAXP	PLANT		PLANT
MODS	BET		BET, MODS
ORDER	PLANT		PLANT, REVENUE, COMPCOS (inactive)
PERIOD	RDTE		RDTE
PLOTDAT	BET		BET, CURVES, PLOTTER
POP	BET		BET, AMORITZ, BUYS, PLOTSGL, INPLANT
PRDSCHL	PLANT		PLANT, COMPCOS (inactive)
PRICEO	ACPRICE	INPLANT	ACPRICE, INPLANT, REVENUE
PRINT	INPLANT		INPLANT, INTROR
PRODIDS	COMPCOS		COMPCOS, INTROR
PRODUCT	PLANT		PLANT, COMPCOS
RDTECMP	RDTE		RDTE, CASHFLW, INTROR

Table A.1
COMMON BLOCK DEFINITION AND USAGE (Concluded)

BLOCK NAME	ROUTINES WHERE COMMON BLOCK IS		
	DEFINED	REDEFINED	USED
RDTELBL	RDTE		RDTE, INTROR
RESULTS	BET		BET, MODS, BUYS, PLOTSGL
SHARES	SHARE		BET, SHARE
STARTER	BET		BET, CURVES, PLOTTER, PLOTSGL
STARTUP	INPLANT		INPLANT, PLANT
STATLIST	BET	MODS	BET, MODS, AMORTIZ, PLOTSGL
TOTAL	PLANT		PLANT, COMPCOS, REVENUE
TOTALS	BET	BET	BET, SHARE
TRDTEC	ACOST		ACOST, RDTE, COSTPR, COMPCOS
TTLCMP	ACOST		ACOST, COMPCOS, COSTPR
TTS	PLANT	PLANT	PLANT, RDTE, COMPCOS (inactive)

Table A.2
DEFINITION OF FLEET ACCOUNTING MODULE VARIABLES

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
BUY	R	C				BET	Cumulative number of aircraft brought up through a given year of a particular type or in a market or in all markets.
CARD	A	I	8	DI		UNIT04	Data on an input card.
CURVE	R	C	33	C		PLOTTER	Storage array for RPM or fuel consumption data to be plotted next.
D1	R				STARTER		Not defined.
D2	R				STARTER		Not defined.
DUMMY	R	A	15	DI	STARTER	CURVES	Temporary designation for unused variables in common starter.
DYEAR	I	C				AMORTIZ	Past year such that aircraft bought in that year are of retirement age in the current year.
FBURNED	R	C			TOTALS	BET	Fuel consumed in given year by aircraft of a given type; or cumulative fuel consumed up through a given year by aircraft of a particular type or in a market or in all markets.
FBURNED	R	C	31	Y		PLOTSGL	Fuel consumed in a year by aircraft of a specific type.
FCTFUEL	R	C				BET	Fraction of fuel burned in a given year in a specific market relative to total fuel burned in the year.
FCTPOPL	R	C				BET	Fraction of the number of aircraft in service in a given year in a specific market relative to total aircraft in service in the year.
FCTRPMS	R	C				BET	Fraction of the RPMs generated in a given year in a specific market relative to the total RPMs generated in the year.
FCTSMIS	R	C				BET	Fraction of the seat miles flown in a given year in a specific market relative to the total seat miles flown in the year.

Table A.2 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
FUELRN	R	C	10, 31	A, Y	RESULTS	BET	Fuel consumed per aircraft of each type in each year.
FUELRN	R	C	30, 31	C, Y	PLOTDAT	CURVES	Array for storing fuel consumption data for plotting aircraft market share curves.
GROWTH	R	I	31	Y		BET	Growth rate (in percent) of RPMs for each year.
I	I	C				BET	Various uses: Do loop index, array index, etc.
IN	I	C			INDICES	BET	Index equal to number of existing aircraft types plus one.
IT	I	C				BET	Array element index.
J	I	C				BET	Various uses: Do loop index, array index, etc.
LP	R	I	31	Y	STATLIST	BET	Load factor for flights in a market for each year.
LIFETIM	R	I	10	A	STATLIST	BET	Lifetime or nominal retirement age in years for each type of aircraft.
LOADPCT	R	C				BUYS	Computed load factor for flights in a market in a year when seat miles supplied exceeds number needed to maintain input load factor (decimal).
LUN	I	A				BET	Equal to 9, a logical unit number for the plotter.
LUN11	I	A				BET	Equal to 11, a logical unit number for the plotter.
MARKET	R	C	31	Y	MARKET	BET	Projected RPMs demanded in each year.
MAXFBRN	I	C				PLOTSGL	Maximum value of fuel consumption in hundreds of millions of barrels.
MAXRPMS	I	C				PLOTSGL	Maximum value of RPMs flown in hundreds of billions of miles.
MLIFET	R	I	10	A	MONS	BET	Lifetime or nominal retirement age in years for modified aircraft.
MODATA	A, R	I	7	MD		BET	Temporary storage for seven data items pertaining to modified aircraft: MTYPE, MODYR, MSEATS, MSPEC, MSPEED, MUTILIZ, MLIFET.

Table A.2 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
MODYR	R	I	10	A	MODS	BET	Year in which aircraft modification to begin.
MRKT	I	A				BET	Number indicator for each market.
MRKTYPE	A2	I	3	MA	STARTER	BET	Alphanumeric name given each market.
MSEATS	R	I	10	A	MODS	BET	Average number of seats in modified aircraft.
MSFC	R	I	10	A	MODS	BET	Average fuel consumption per seat mile for modified aircraft.
MSPEED	R	I	10	A	MODS	BET	Average speed in miles per hour for modified aircraft.
MTYPE	A	I	10	A	MODS	BET	Alpha descriptor of aircraft type to be modified.
MUTILIZ	R	I	10	A	MODS	BET	Utilization in hours per year for modified aircraft.
NAME	A	C				BET	Alpha descriptor of aircraft type from MODATA(1).
NOBUY5	R	I	10,46	A,Y	POP	BET	Number of aircraft of each type placed in service (bought) in each year.
NOBYS	R	C			TOTALS	BET	Number of aircraft of a nspecific type bought in a given year.
NOCRVS	I	C			STARTER	CURVES	Number of curves to be plotted.
NOEXPLS	I	I			INDICES	BET	Number of existing aircraft types in a market.
NOMODS	I	I				BET	Number of aircraft types to be modified.
NOMRKSF	I	I,A				PLOTSGL	Number of tick marks on the fuel consumption axis (Y-axis) of plots.
NOMRKSX	I	I,A				PLOTSGL	Number of tick marks on the RPMs axis (Y-axis) of plots.
NONEW	I	I				BET	Number of new aircraft types in a market.
NORETIR	R	I	10,46	A,Y	POP	BET	Number of aircraft retired from service of each type in each year.
NORTR	R	C			TOTALS	BET	Number of aircraft of a given type retired in a given year.

Table A.2 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
OUT	I	C			INDICES	BET	Index equal to total number of existing and new types of aircraft in a market.
PASS	I					AMORTIZ	Variable not used.
PERCENT	R	C			STARTER	BET	RPM growth rate for a market in the base year.
PLOTS	A	I	10	A	STATLIST	BET	Flag indicating whether plots of fuel consumed and RPMs flown versus time are to be provided for an aircraft type.
PLTFUEL	R	C	3	MA	STARTER	BET	Total fuel consumed by all aircraft in a market in 2005.
PLTRPMS	R	C	3	MA	STARTER	BET	Total revenue passenger miles flown by all aircraft in a market in 2005.
POPUL	R	C	10,31	A,Y	POP	BET	Number of aircraft in service of each type in each year.
POPULNO	R	C			TOTALS	BET	Number of aircraft of a given type in service in a specific year.
RETIR	R	C				BET	Cumulative number of aircraft retired up through a given year of a particular type or in a market or in all markets.
RPM	R	C	10,31	A,Y	RESULTS	BET	Revenue passenger statute miles flown per aircraft of each type in each year.
RPMDIFF	R	C				BUYS	Difference between RPMs demanded and those available in a market before aircraft buys in a given year.
RPMS	R	C			TOTALS	BET	RPMs flown in a given year by aircraft of a given type; or cumulative RPMs flown up through a given year by aircraft of a particular type or in a market or in all markets.
RPMS	R	C	31	Y		PLOTSGL	RPMs flown in a year by aircraft of a specific type.
RPME	R	C	30,31	C,Y	PLOTDAT	CURVES	Array for storing RPMs flown for plotting market share curves.

Table A.2 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
SEATHI	R	C	10,31	A,Y	RESULTS	BET	Number of seat miles flown per aircraft of each type for each year.
SEATS	R	I	10	A	STATLIST	BET	Average number of seats per aircraft of each type.
SFC	R	I	10	A	STATLIST	BET	Average fuel consumption per seat mile flown for each aircraft type.
SHRBUYS	R	C	3,31	MA,Y	SHARES	SHARE	Number of aircraft placed in service in a market in each year.
SHRETIR	R	C	3,31	MA,Y	SHARES	SHARE	Number of aircraft retired from service in a market in each year.
SHRFUEL	R	C	3,31	MA,Y	SHARES	SHARE	Fuel consumption in a market in each year.
SHRPOPL	R	C	3,31	MA,Y	SHARES	SHARE	Number of aircraft in service in a market for each year.
SHRPPMS	R	C	3,31	MA,Y	SHARES	SHARE	RPMs flown in a market in each year.
SHRSMIS	R	C	3,31	MA,Y	SHARES	SHARE	Seat miles flown in a market in each year.
SMILES	R	C			TOTALS	BET	Seat miles flown in a given year by aircraft of a given type; or cumulative seat miles flown up through a given year by aircraft of a particular type or in a market or in all markets.
SPEED	R	I	10	A	STATLIST	BET	Average block-to-block speed for each type of aircraft.
TOPFERN	R	I,C				PLOTSGL	Maximum value on the fuel consumption axis (Y-axis) of plots.
TOPRPMS	R	I,C				PLOTSGL	Maximum value on the RPMs axis (Y-axis) of plots.
TOTAL	R	C				BUYS	Total RPMs available in a market in a given year before aircraft buys in the year.
TOTLBUY	R	C	31	Y	ACCUMS	SHARE	Total number of aircraft bought in all markets for each year.
TOTLFLB	R	C	31	Y	ACCUMS	SHARE	Total fuel consumed in all markets in each year.

Table A.2 (Concluded)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
TOTLMIS	R	C	31	Y	ACCUMS	SHARE	Total seat miles flown in all markets in each year.
TOTLPOP	R	C	31	Y	ACCUMS	SHARE	Total number of aircraft in service for all markets in each year.
TOTLRPM	R	C	31	Y	ACCUMS	SHARE	Total revenue passenger miles flown in all markets in each year.
TOTLRTR	R	C	31	Y	ACCUMS	SHARE	Total number of aircraft retired from service in all markets for each year.
TYPE	A	I	10	A	STATLIST	BET	Alpha name given to each aircraft type.
UTILIZ	R	I	10	A	STATLIST	BET	Utilization in hours per year per aircraft of each type.
XO	R	A				PLOTSGL	X-coordinate starting value for plots.
XSC	R	C				PLOTSGL	X-coordinate scaling factor for plots.
YEAR	I	C				BET	Index denoting the calendar year.
YEARS	R	C	31	C		PLOTSGL	Calendar year from 1975 through 2005.
YR	I	C				BET	Index indicating the number of the year.
YRINTRO	R	I	10	A	STATLIST	BET	Year of introduction for each aircraft type.
YSC	R	C	31	Y		PLOTSGL	Y-coordinate scaling factor.

FOOTNOTES

¹Type of variable classification:

A = Alphanumeric
 A2 = Double precision alphanumeric
 I = Integer
 R = Real

²Value mode classifications:

C = Value calculated in program
 I = Read from input file
 A = Value assigned in program

³Indexed parameter for arrays:

A = Aircraft Type
 MA = Market type
 Y = Number of years
 C = Number of curves
 MD = Modification data
 DI = Data items

Table A.3
DEFINITION OF AIRFRAME MANUFACTURER MODULE VARIABLES

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED 3 PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
ACCASHF	R	C	372	MO		CASHFLW	Accumulated monthly cash flow.
ACCINCO	R	C	372	MO		REVENUE	Accumulated revenue (income) by month.
ACCOMP	R	C	11, 372	CM, MO		INTROR	Accumulated production costs by component and month.
ACCOST	R	C	372	MO		CASHFLW	Accumulated sum of RDT&E and production costs by month.
ACCUM	R	C				PLANT	Sum of demand at end of each demand year.
ACRDTE	R	C	5, 372	CM, MO		INTROR	Accumulated RDT&E costs by component and month.
AD	R	C			COSTOT	ACCOST	Avionics development cost.
ADDE	R	C			COSTOT	ACCOST	Total airframe design and development engineering cost, includes concept formulation and contract definition.
ADI	R	DSD, I				ACCOST	Avionics development cost.
AFSPA0	R	DSD				ACCOST	Airframe spares factor, production phase.
AGE0	R	C			COSTOT	ACCOST	Operational ground support equipment cost.
AGE01	R	DSD, I				ACCOST	Operational ground support equipment cost.
AGEP	R	C			COSTOT	ACCOST	Ground support equipment development cost.
AGEPI	R	DSD, I				ACCOST	Ground support equipment development cost.
AMFG	R	C			CSHFLO	ACCOST	Airframe manufacturing cost.
AP	R	C			CUMOUT	ACCOST	Average airplane price for Q units.
API	R	C				ACPRICE	Fractional form of average national price/cost index over period of interest.
AQ	R	C			COSTOT	ACCOST	Production total cost for NV airplanes.
AQFEE	R	C			COSTOT	ACCOST	Production phase contractor fee.
ASTOP	R	A				ACPRICE	Indicator for end of data.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
AVGDIF	R	C				PLANT	Months start of production to be offset to improve production schedule.
BBRKPT	I	C				PLANT	Temporary save for "b.st" period breakpoint.
BIMCI	R	C	31	Y		INRR	Cash flow for eac year.
BRKPT	I	C				PLANT	Loop indicator--year in demand schedule.
BSTOP	R	A				ACPRICE	Set equal to blank word.
C	R	C	37,5	CM,LC	COSTOT	ACOST	Production cost factors by learning curve step.
C	R	A				ACPRICE	Constant to convert to millions (1.0E-6).
CACS	R	C				ACOST	Air conditioning system cost.
CAERO	R	C				ACOST	Aerodynamic control system cost.
CAFCT	R	C				ACOST	Cumulative total airframe costs for Q units.
CAFCTI	R	C				ACOST	Cumulative total airframe costs for Q-1 units.
CAFFV	R	C				ACOST	Flight test vehicle airframe cost.
CAFO	R	C				ACOST	Production aircraft airframe cost.
CAFO	R	C				COMPCOS	Airframe cost for each unit based on learning curve.
CAFOG	R	C				COMPCOS	Airframe learning curve factor for Nth unit.
CAFOI	R	C				COMPCOS	Airframe learning curve factor for N+1st unit.
CAFUCA	R	C			CUMOUA	ACOST	Cumulative average unit airframe cost for Q units.
CANTIC	R	C				ACOST	Anti-icing cost.
CASHFLO	R	C	372	MO		CASHFLW	Cash flow by month.
CAVCT	R	C				ACOST	Cumulative total avionics costs for Q units.
CAVCTI	R	C				ACOST	Cumulative total avionics costs for Q-1 units.
CAVTV	R	C				ACOST	Flight test vehicle avionics cost.
CAVION	R	C			CSHFLO	ACOST	Avionics system cost.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
CAVO	R	C				ACCOST	Production aircraft avionics cost.
CAVO	R	C				COMPCOS	Avionics cost for each unit based on learning curve.
CAVOU	R	C				COMPCOS	Avionics learning curve factor for Nth unit.
CAVOI	R	C				COMPCOS	Avionics learning curve factor for N+1st unit.
CAVONE	R	C				ACCOST	Avionics equipment cost.
CAVONI	R	C				ACCOST	Avionics installation cost.
CAVONT	R	C				ACCOST	Total avionics equipment and installation cost.
CAVUCA	R	C			CUMOUT	ACCOST	Cumulative average unit avionics cost for Q units.
CBODY	R	C				ACCOST	Aircraft fuselage cost.
CD	R	C			COSTOT	ACCOST	Contract definition phase cost.
CELCAD	R	C				ACCOST	Electrical distribution system cost.
CEMP	R	C				ACCOST	Empennage cost.
CENACC	R	C				ACCOST	Engine accessories cost.
CENGs	R	C				ACCOST	Airplane engines cost.
CF	R	C				ACCOST	Concept formulation phase cost.
CFACS	R	DSD, I				ACCOST	Complexity factor air conditioning system.
CFAERO	R	DSD, I				ACCOST	Complexity factor aerodynamic control system.
CFANTC	R	DSD, I				ACCOST	Complexity factor anti-icing system.
CFASSY	R	C				ACCOST	Final assembly and check-out cost.
CFAVON	R	DSD, I				ACCOST	Complexity factor avionics system.
CFBODY	R	DSD, I				ACCOST	Complexity factor aircraft fuselage.
CFELCD	R	DSD, I				ACCOST	Complexity factor electrical distribution system.
CFEMP	R	DSD, I				ACCOST	Complexity factor empennage structure.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED ³ PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
CFENAC	R	DSD, I				ACCOST	Complexity factor engine accessories.
CFENG	R	DSD, I				ACCOST	Complexity factor airbreathing engines.
CFFUSY	R	DSD, I				ACCOST	Complexity factor fuel system.
CFHNDL	R	DSD, I				ACCOST	Complexity factor for loading and handling.
CFHYCD	R	DSD, I				ACCOST	Complexity factor hydraulic system.
CFINST	R	DSD, I				ACCOST	Complexity factor instrument system.
CFLG	R	DSD, I				ACCOST	Complexity factor alighting gear system.
CFNAC	R	DSD, I				ACCOST	Complexity factor engine nacelles.
CFPACC	R	DSD, I				ACCOST	Complexity factor passenger accommodations.
CFPNCD	R	DSD, I				ACCOST	Complexity factor pneumatic system.
CFPOW	R	DSD, I				ACCOST	Complexity factor auxiliary power system.
CFTRV	R	DSD, I				ACCOST	Complexity factor thrust reverser.
CFUSYS	R	C				ACCOST	Fuel system cost.
CFWING	R	DSD, I				ACCOST	Complexity factor wing structure.
CHANDL	R	C				ACCOST	Load and handling system cost.
CHYCAD	R	C				ACCOST	Hydraulic system cost.
CINST	R	C				ACCOST	Instrument system cost.
CINSTE	R	C				ACCOST	Instrument equipment cost.
CINSTI	R	C				ACCOST	Instrument installation cost.
CLG	R	C				ACCOST	Alighting gear cost.
CNACEL	R	C				ACCOST	Engine nacelles cost.
COMENT	R	I	8	DI		ACPRICE	Input card comments.
COMPCST	R	C	11,372	CM, MO	COMPCST	COMPCOS	Summary array of production costs by component and month.
CONFIG	R	DSD, I				ACCOST	Engineering complexity factor.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
COST	R	C	374	MO		CASHFLW	Sum of RDT&E and production costs each month.
CPACCO	R	C				ACCOST	Passenger accommodations and furnishing cost.
CPCT	R	C				ACCOST	Cumulative total propulsion costs for Q units.
CPCTI	R	C				ACCOST	Cumulative total propulsion costs for Q-1 units.
CPFC	R	C				ACCOST	Flight test vehicle propulsion system cost.
CPWCAD	R	C				ACCOST	Pneumatic system cost.
CPO	R	C				ACCOST	Production aircraft propulsion system cost.
CPO	R	C				ACCOST	Propulsion cost for each unit based on learning curve.
CPOO	R	C					Propulsion learning curve factor for Nth unit.
CPOI	R	C					Propulsion learning curve factor for N+1st unit.
CPOWER	R	C				ACCOST	Auxiliary power system cost.
CPUCA	R	C			CINOUT	ACCOST	Cumulative average unit propulsion cost for Q units.
CSTRUC	R	C				ACCOST	Assembled airplane cost.
CTJ	R	C				ACCOST	Airplane engine unit cost.
CTJI	R	DSD, I				A.COST	Input value for airplane engine unit cost.
CTREVS	R	C				ACCOST	Thrust reverser cost.
CV	R	C				ACCOST	Total aircraft manufacturing cost.
CWING	R	C				ACCOST	Wing cost.
DCF	R	C	31	Y		INRR	Discounted cash flow for each year.
DEEL	R	C			COSTOT	ACCOST	Airframe design and development engineering cost.
DEITHPL	I	C				RDT&E	Last month before delivery month.
DELAY	I	A	5	CH		RDT&E	Delay before RDT&E component cost starts (5 factors).

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED ³ PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
DELIVER	I	C	4000	NA	DELIVER	PLANT	Month in which each unit will be delivered (same as PRODUCT).
DELTAP	R	C				INPLANT	Increment for aircraft price.
DELTAR	R	A				INRR	Change in rate for present value calculations.
DELSCHL	I	C	396	MO		PLANT	Cumulative number of units delivered each month (same as PRDSCHL).
DEMAND	R	C	31	Y	DEMAND	INPLANT	No. of one new aircraft type demanded each year.
DISCREP	I	C	29	Y		PLANT	Discrepancy between production possible and demanded.
DS	R	C			COSTOT	ACCOST	Research, development, test and evaluation support.
EAIRFP	R	C				ACPRICE	Estimated airframe price.
EAIRPR	R	C				ACPRICE	Estimated airplane price.
EDEVIC	R	C				ACPRICE	Estimated airplane development cost.
EENGPR	R	C				ACPRICE	Estimated engines total price.
EN	R	I				ACPRICE	Number of main engines
EN	R	DSD, I				ACCOST	Number of main engines.
END1	I	C				PLANT	Last month in production schedule for period 1.
END1	I					COMPCOS	Not used.
END2	I	C				PLANT	Last month in production schedule for period 2.
ENSPAO	R	DSD, I			CSHPLO	ACCOST	Main engine spares factor, production phase.
ENSPAR	R	DSD, I				ACCOST	Main engine spares factor, RDT&E phase.
EPRICE	R	C			PRICED	ACPRICE	Estimated airplane market place price.
ESEPRI	R	C	3	A		ACPRICE	Estimated airplane price by seat cost, millions

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED ³ PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
FAC	R	C			COSTOT	ACCOST	Production facilities cost.
FACI	R	DSD, I				ACCOST	Input value of production facilities cost.
FEE	R	DSD, I			CUMOUT	ACCOST	Manufacturer fee factor.
FEE	R	I				ACPRICE	Not used.
FIRSTPR	I	C				PLANT	No. of years in first period.
FP1	I	C				PLANT	Pointer to first year of period 2.
FPROD1	R	C				PLANT	Actual monthly production rate for period 1.
FPROD2	R	C				PLANT	Actual monthly production rate for period 2.
FTO	R	C			COSTOT	ACCOST	Flight test operation cost.
FTOI	R	DSD, I				ACCOST	Input value for flight test operation cost.
FTS	R	C			COSTOT	ACCOST	Flight test aircraft spares cost.
FV	R	C			COSTOT	ACCOST	Flight test vehicles cost.
FVCT	R	C			CUMOUT	ACCOST	Cumulative total airplane costs for Q units.
FVCTI	R	C				ACCOST	Cumulative total airplane costs for Q-1 units.
FVSPAR	R	DSD, I				ACCOST	Flight test vehicle spares.
FVUC	R	C			CUMOUT	ACCOST	Unit airplane cost of Q-th unit.
FVUCA	R	C			CUMOUT	ACCOST	Cumulative average unit airplane cost for Q units.
GTS	R	C			CUMOUT	ACCOST	Ground test vehicle spares cost.
GTS PAR	R	DSD, I				ACCOST	Ground test vehicle spares factor.
GTV	R	C			COSTOT	ACCOST	Ground test vehicles cost.
HP	R	I				ACPRICE	Engine design shaft horsepower.
I	I	LI				INPLANT	Loop index (multiple uses).
IA	I	C	3	DI		ACPRICE	Indicator used in estimating airplane cost.
LACCUM	I	C				PLANT	Rounded (up by .67) sum of units demanded.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
IAIRPL	I	I				ACPRICE	Indicator for airplane type. IAIRPL=1 for conventional jet transports. IAIRPL=2 for small jet transports. IAIRPL=3 for wide body jet transports. IAIRPL=4 for turbo-prop transports. IAIRPL=5 for general aviation types. IAIRPL=6 for supersonic transports.
IB	I	LI				PLANT	Loop index - starts at 2.
ICNT	I	C				INRR	Counter to find every 10th cycle.
ICONPG	I	DSD,I				ACCOST	Indicator for aircraft type, (6 = subsonic production, 7 = prototype, 8 = supersonic production).
ICUM	I	C			CUMOUT	ACCOST	Indicator for cumulative quantity.
IDATA	I	DSD,I				ACCOST	Not used.
IDIFF	I	C				PLANT	Difference between last month of each demand year and month last unit scheduled for production.
IENG	I	I				ACPRICE	Indicator for type of engines. IENG=1 for turbojet and turbofan. IENG=2 for turboprop. IENG=3 for reciprocating. IENG=4 for airbreathing.
II	I	LI				ACPRICE	Loop index.
IL	I	C				INRR	Last value of loop index.
IMAX1	I	C				PLANT	Temporary to find largest differences in period 1.
IMAX2	I	C				PLANT	Temporary to find largest differences in period 2.
IMIN1	I	C				PLANT	Temporary to find largest differences in period 1.
IMIN2	I	C				PLANT	Temporary to find largest differences in period 2.

Table 4.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED 3 PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
IMNTH	I	C			STARTUP	INPLANT	Month in which demand begins (years since 1974 x 12).
IN	I	C				RDTE	Starting month for each RDTE component.
INCOME	R	C	372	MO		REVENUE	Total revenue (income) each separate month.
IOPS	I	DSD, I				ACCOST	Indicator for type operational program, 1 = commercial airline, 0 = other.
IP	I	C			PRINT	INPLANT	Print flag for monthly cost, income, and cash-flow tables in INTROR (0=print, 1=no print).
IPOWER	I	DSD, I				ACCOST	Not used.
IPROO	I	DSD, I			CUMOUT	ACCOST	Indicator for prototype or production tooling; 1 = production, 0 = prototype.
IQ	I	DSD, I	33	NA	CUMOUT	ACCOST	Indicator for aircraft quantity matrix.
IR	I	LI				INTROR	Pointer to month within loops.
IS	I	C				INRR	Starting value of loop index.
IT	R	C				ACCOST	Initial flight crew training cost for NV airplanes.
ITH	I	LI				CASHFLW	Component loop index.
ITHPL	I	LI				PLANT	Unit loop index pointer.
ITOTAL	I	C			TOTAL	PLANT	Number of units in period 1.
ITOTAL1	I	C				PLANT	Upward rounded (by .67) number of whole aircraft demanded in period 1.
ITOTAL2	I	C				PLANT	Upward rounded (by .67) number of whole aircraft demanded in period 2.
IV	R	C				ACCOST	Total aircraft production costs for NV airplanes.
IMLC	I	DSD, I				ACCOST	Indicator for landing gear component breakdown.
IYEAR	I	C				PLANT	Loop index for each year of demand.
IYR	I	C				INPLANT	Indicator for first year of demand (since 1974).

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED ³ PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
J	I	LI				INPLANT	Loop index.
JMTHPL	I	C				PLANT	Offset pointer to month of production for each unit.
K	R	C				INRR	Temporary for accumulated discounted cash flow.
KICKOFF	I	C	29	Y		PLANT	Startup date of production for each period.
KK	R	C				INRR	Temporary for accumulated discounted cash flow.
KKK	R	C				INRR	Temporary for accumulated discounted cash flow.
LB	I	C				PLANT	Loop index.
LEARN	R	DSD, I			CSHFLO	ACCOST	Airframe learning curve.
LEARNA	R	DSD, I			CSHFLO	ACCOST	Avionics learning curve.
LEARNP	R	DSD, I			CSHFLO	ACCOST	Engine learning curve.
LN	I	C				RDTE	Months before production that RDTE component cost starts.
LTIME	I	DSD	372	MO		PLANT	Lead time in months on delivery from order times.
MACH	R	DSD, I			CUMOUT	ACCOST	Maximum design flight mach number for engines.
MAXP	I	C			MAXP	PLANT	Maximum monthly production rate.
MEQ	R	C			COSTOT	ACCOST	Miscellaneous equipment cost.
MIN	I	A				PLANT	Not used.
MINIMAX	I	C				PLANT	Largest difference (in absolute value) for each subdivision.
MNTH	I	LI				CASHFLW	Month loop index.
MNTHPAY	I	C				REVENUE	Monthly payments between order and delivery.
N	I	LI				ACCOST	Loop index.
NCREW	R	DSD, I				ACCOST	Number in flight crew per airplane.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
NDATA	I	DSD,I			COSTOT	ACCOST	Number of positions (from 1 to 5) on learning curve.
NFV	R	DSD,I			COSTOT	ACCOST	Number flight test vehicles.
NG	R	DSD,I			COSTOT	ACCOST	Number of ground test vehicles.
NL	I	C				INRR	Calculated last value of loop index.
NOCOMP	I	A				COMPCDS	Number of production cost components.
NOCON	R	DSD,I				ACCOST	Number of concept formulation contractors.
NOCON1	R	DSD,I				ACCOST	Number of contract definition contractors.
NOENC	R	DSD,I				ACCOST	Number of concept formulation engineers.
NLENC1	R	DSD,I				ACCOST	Number of contract definitions engineers.
NOYES	R	DSD,I				ACCOST	Number of years for concept formulation.
NOYES1	R	DSD,I				ACCOST	Number of years for contract definitions.
NPL	R	C				ACCOST	Total number flight crew personnel to be trained.
NTRY	I	C				PLANT	Program control index.
NV	R	DSD,I,C			COSTOT	ACCOST	Number of operational vehicles.
NVEII	R	DSD,I	5	LC	COSTOT	ACCOST	Number of vehicles for which costs to be computed.
NVHF	R	C			COSTOT	ACCOST	Total number flight test and operational vehicles.
NVHF	I	I				ACPRICE	Total number flight test & operational vehicles.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
ORDER	I	C	4000	NA	ORDER	PLANT	Month in which each unit is ordered.
ORITHPL	I	C				REVENUE	Next month after order month.
OS	R	C			COSTOT	ACCOST	Operational vehicles spares cost.
OSA	R	C				ACCOST	Production airframe spares.
OSP	R	C				ACCOST	Production engine spares.
OSPO	R	C				COMP COS	Engine spares learning cost factor for Nth unit.
OSP1	R	C				COMP COS	Engine spares learning cost factor for N+1st unit.
OT	R	C			COSTOT	ACCOST	Training equipment cost.
OUT	I	C				RDTE	Ending month for each RDTE component.
OV	R	C			COSTOT	ACCOST	Operational vehicles costs.
PDTJ	R	C			COSTOT	ACCOST	Propulsion development cost turbo-jet engines.
PDTJ1	R	DSD, I				ACCOST	Input value for propulsion development cost turbo-jet engines.
PERIOD	I	C			LIFETIM	INPLANT	No. of years new aircraft is demanded.
PERIOD	I	A	5	CM		RDTE	Duration of each RDTE cost component.
PN	R	DSD, I				ACCOST	Total number of passengers.
PN	R	I, A				ACPRICE	Total passenger capacity.
PO	R	I				INPLANT	Base market price of aircraft type.
PP	R	C	20	PE		INPLANT	Array of aircraft price estimates used.
PRODIDS	R	A			PRODIDS	COMP COS	Array of alphabetic titles for production cost factors.
PRI	R	C				COSTPR	Indicator for vehicle type.
PRICEO	R	C				INPLANT	Price of aircraft.
PRODUCT	I	C	4000	NA		PLANT	Month in which each unit will be produced.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED 3 PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
PROD1	I	C				PLANT	Upward rounded (by .67) monthly production rate in first period.
PROD2	I	C				PLANT	Upward rounded (by .67) monthly production rate in first period.
PROFU	R	C			CSHFLO	ACCOST	Total propulsion system cost.
PROSCHL	I	C	396	MO	PRDSCHL	PLANT	Cumulative number of units produced each month.
PV	R	C	31	Y		INRR	Present value factor.
Q	R	C				ACCOST	Total number of vehicles manufactured.
R	R	C				INRR	Rate of return difference.
RATE	R	DSD, I			CUMOUT	ACCOST	Vehicle production rate, number/month.
RDPEE	R	C			COSTOT	ACCOST	Contractor fee, RDTE phase.
RDTE	R	C				ACCOST	Research, development, testing and engineering cost.
RDTECMP	R	C	5,372	CM, MO		RDTE	Monthly breakdown of RDTE costs by component.
RDTELBL	R	A	5	CM		RDTE	Alphabetic descriptor of RDTE cost components.
RE	R	DSD, I				ACCOST	Engineering labor rate.
RR	R	C	20	PE		INPLANT	Array of rate of return values based on aircraft price.
RR	R	C				INRR	Rate of return difference.
RT	R	DSD, I				ACCOST	Tooling labor rate.
RRR	R	C				INRR	Rate of return difference.
SC	R	C			CUMOUT	ACCOST	Total manufacturing sustaining costs for Q units.
SE	R	C			COSTOT	ACCOST	Sustaining engineering costs.
SEO	R	C				COMPCOS	Sustaining engineering cost factor for Nth unit.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED ³ PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
SEL	R	C				COMCOS	Sustaining engineering cost fact for N+1st unit.
SPARES	R	C				COMCOS	Total spares cost based on learning curve (same as OS).
ST	R	C			COSTOT	ACOST	Sustaining tooling cost.
STO	R	C				COMCOS	Tooling and special equipment learning cost factor for Nth unit.
ST1	R	C				COMCOS	Tooling and special equipment learning cost factor for N+1st unit.
SUBSYS	R	C			COSTOT	ACOST	Subsystem development cost.
SUBSYI	R	DSD, I				ACOST	Input value for subsystem development cost.
SUM	R	C				CASHFLW	Used for totaling RDT&E and production costs.
SUM	R	C				INRR	Temporary sum of discounted cash flow.
SUM	R	C				REVENUE	Temporary to accumulate sum by month.
SUM	R	C				INTROR	Temporary to accumulate sum by month.
SUM1	R	C				CASHFLW	Used for totaling RDT&E costs.
SUM2	R	C				CASHFLW	Used for totaling production costs.
SUMDIF	R	C				ACOST	Sum of differences for each period subdivision.
T	R	I				ACOST	Thrust per engine at sea level.
TDO	R	C			COSTOT	ACOST	Production aircraft technical data cost.
TDP	R	C			COSTOT	ACOST	RDT&E technical data cost.
TITLE	R	I	10	DI		COSTPR	Input title for printouts.
TMC	R	C			CUMOUT	ACOST	Total of all manufacturing costs for Q units.
TOOLC	R	DSD, I				ACOST	Complexity factor tooling.
TOTAL	R	C				PLANT	Actual total number of aircraft demanded in both periods.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
TOTAL1	R	C				PLANT	Actual number of craft demanded in 1st period.
TOTAL2	R	C				PLANT	Actual number of craft demanded in 2nd period.
TOVERW	R	C				ACCOST	Total engine thrust over airplane takeoff gross weight.
TPEREN	R	C				ACCOST	Thrust in pounds per engine.
TRDTE	R	C			COSTOT	ACCOST	Total research, development, tooling & engineering cost.
TRDTEC	R	C	5	CM		ACCOST	Array of RDT&E cost factors.
TRI	R	C			COSTOT	ACCOST	Initial transportation cost.
TS	I	C			STARTUP	INPLANT	Same as IMNTH in INPLANT. Month demand for aircraft type starts.
TST	R	C			COSTOT	ACCOST	Tooling and special equipment cost.
TTLCHP	R	C	5	CM		ACCOST	Array of production cost factors.
TTS	I	C			TTS	PLANT	Modified start of production month.
UMC	R	C				COMPCOS	Operational vehicles cost (same as UV).
VMAX	R	C				ACCOST	Maximum vehicle speed, knots.
WA	R	C			CUMOUT	ACCOST	Vehicle AMPR weight.
WACS	R	DSD, I				ACCOST	Air conditioning system weight.
WAERO	R	DSD, I				ACCOST	Aerodynamic control system weight.
WAIRFR	R	C				ACPRICE	Airframe weight.
WANTIC	R	DSD, I				ACCOST	Anti-icing system weight.
WAVION	R	DSD, I				ACCOST	Avionics system weight.
WAVIOT	R	C				ACCOST	Total avionics and instrument weight.
WBODY	R	DSD, I				ACCOST	Fuselage weight.
WE	R	C				ACCOST	Aircraft empty weight.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
WE	R	I				ACPRICE	Aircraft empty weight.
WELCAD	R	DSD, I				ACCOST	Electric power conversion and distribution system weight.
WT TP	R	DSD, I				ACCOST	Empennage weight.
WENACC	R	DSD, I				ACCOST	Engine accessories weight.
WENG3	R	DSD, I				ACCOST	Engines total weight.
WENG5	R	I				ACPRICE	Engines total weight.
WFUSYS	R	DSD, I				ACCOST	Fuel system weight.
WFUIOT	R	DSD, I				ACCOST	Total fuel weight.
WCROSS	R	DSD, I			CUMOUT	ACCOST	Aircraft gross take-off weight (= WC + WTO).
WHANDL	R	DSD, I				ACCOST	Load and handling system weight.
WHYCAD	R	DSD, I				ACCOST	Hydraulic power conversion and distribution system weight.
WINST	R	DSD, I				ACCOST	Instrument system weight.
WLG	R	DSD, I				ACCOST	Alighting gear system weight.
WLGCON	R	DSD, I				ACCOST	Alighting gear controls weight.
WLGSTR	R	DSD, I				ACCOST	Alighting gear structure weight.
WLGIRS	R	DSD, I				ACCOST	Tire weight.
WLGW4L	R	DSD, I				ACCOST	Wheels and brake weight.
WNACEL	R	DSD, I				ACCOST	Engine nacelle weight.
WPACCO	R	DSD, I				ACCOST	Passenger accommodations (and equipment) weight.
WPAYL	R	DSD, I				ACCOST	Payload weight.
WPNCAD	R	DSD, I				ACCOST	Pneumatic power and distribution system weight.
WPOWER	R	DSD, I				ACCOST	Auxiliary power system weight.
WPPROV	R	C				ACCOST	Crew size related subsystem development cost factor.

Table A.3 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED ³ PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
WTREVS	R	DSD, I				ACCOST	Thrust reverser weight.
WING	R	DSD, I				ACCOST	Wing weight.
SAVD	R	DSD, I				ACCOST	Avionics development factor.
XFASSY	R	DSD, I			CSHFLO	ACCOST	Final assembly-check out cost fraction.
XNEW	R	DSD, I				ACCOST	Miscellaneous equipment development factor.
YEAR	I	LI				INPLANT	Loop index.
YR	I	LI				INPLANT	Loop index.
Z	R	C				ACCOST	Airframe production learning curve cost factor.
ZA	R	C				ACCOST	Avionics production learning curve cost factor.
ZAF	R	C				COMPCOS	Airframe learning curve factor.
ZAV	R	C				COMPCOS	Avionics learning curve factor.
ZETA	R	C				ACCOST	Airframe learning curve exponent.
ZETAA	R	C				ACCOST	Avionics learning curve exponent.
ZETAP	R	C				ACCOST	Engine learning curve exponent.
ZP	R	C				ACCOST	Engine production learning curve cost factor.

FOOTNOTES

¹Type of variable classification:

I - Integer
R - Real

²Value mode classifications:

C - Value calculated in program
I - Read from input file
DSD - Data statement definition
A - Value assigned in program
LI - Loop index

³Indexed parameter for arrays:

Y - Number of Years
DI - Data items
CM - Component number
MO - Number of months
NA - Number of aircraft
PE - Price estimates
LC - Learning curve step

Table A.4
DEFINITION OF AIR CARRIER MODULE VARIABLES

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
ADDC	R	I				DIRECT	Costs of added flight crew (over 2), dollars.
A11	R	I				INDIR	Flight operations expense (less rentals) in dollars per block hour.
A12	R	I				INDIR	Maintenance expense for flight equipment in dollars per block hour.
A14	R	I				INDIR	Flight operations expense for rentals in dollars per block hour.
A15	R	I				INDIR	Cost per stewardess per block hour, dollars.
A16	R	I				INDIR	Food expense in dollars per passenger per block hour.
A17	R	I				INDIR	Cost of other passenger in-flight expenses in dollars per passenger-mile.
A18	R	I				INDIR	Aircraft line servicing expense in dollars per departure.
A19	R	I				INDIR	Aircraft control servicing expense in dollars per block hour.
A110	R	I				INDIR	Landing fee per departure in dollars.
A111	R	I				INDIR	Passenger traffic servicing expense in dollars per passenger.
A112	R	I				INDIR	Baggage traffic servicing expense in dollars per ton.
A112A	R	I				INDIR	Cargo traffic servicing expense in dollars per ton.
A113	R	I				INDIR	Reservation and sales expense per passenger in dollars.
A114	R	I				INDIR	Reservation and sales expense per passenger-mile in dollars.
A115	R	I				INDIR	Reservation and sales expense for property in dollars per ton-mile.

Table A.4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER 3	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
A116	R	I				INDIR	Advertising and publicity expense per passenger-mile in dollars.
A117	R	I				INDIR	Advertising and publicity expense for property in dollars per ton-mile.
A118	R	I				INDIR	Maintenance expense for ground property and equipment per departure in dollars.
A119	R	I				INDIR	Expense for depreciation and amortization of general ground property and equipment in dollars per departure.
A120	R	I				INDIR	Maintenance equipment depreciation factor.
A121	R	I				INDIR	General and administrative expense factor.
ALOSS	R	C				TAX	Yearly total earnings before taxes (where there is a loss), dollars.
AMT	R	A/I				DIRECT/INDIR	Air maneuver time, hours.
AREV	R	A	25,100	Y,A	IN	INPUTS	Annual revenue, dollars.
BAG	R	I				INDIR	Baggage per passenger in pounds.
BF	R	C	12	ST		DIRECT	Block fuel, pounds.
BS	R	C	12	ST		DIRECT	Block speed, mph.
BT	R	C	12	ST		DIRECT	Block time, hours.
BVALUE	R	C	25,100	Y,A	IN	DEPSUB	Book value of aircraft in specific year, dollars.
C1	R	C	17	ST		INDIR	Flying operations (less rentals) expense, dollars.
C2	R	C	17	ST		INDIR	Maintenance flight equipment expense, dollars.
C4	R	C	17	ST		INDIR	Rentals of flight equipment, dollars.
C6	R	C	17	ST		INDIR	Stewardess expense (first class), dollars.
C7	R	C	17	ST		INDIR	Stewardess expense (coach), dollars.

Table A.4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED ³ PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
C8	R	C	17	ST		INDIR	Stewardess expense (total), dollars.
C9	R	C	17	ST		INDIR	Food expense (first class), dollars.
C10	R	C	17	ST		INDIR	Food expense (coach), dollars.
C11	R	C	17	ST		INDIR	Passenger in flight food expense (total), dollars.
C12	R	C	17	ST		INDIR	Other passenger in-flight expenses (first class), dollars.
C13	R	C	17	ST		INDIR	Other passenger in-flight expenses (coach), dollars.
C14	R	C	17	ST		INDIR	Other passenger in-flight expenses (total), dollars.
C15	R	C	17	ST		INDIR	Aircraft line servicing expense, dollars.
C16	R	C	17	ST		INDIR	Aircraft control servicing expense, dollars.
C17	R	C	17	ST		INDIR	Landing fees, dollars.
C18	R	C	17	ST		INDIR	Passenger traffic servicing expense (first class), dollars.
C19	R	C	17	ST		INDIR	Passenger traffic servicing expense (coach), dollars.
C20	R	C	17	ST		INDIR	Baggage traffic servicing expense (first class), dollars.
C21	R	C	17	ST		INDIR	Baggage traffic servicing expense (coach), dollars.
C22	R	C	17	ST		INDIR	Traffic servicing expense (passengers and baggage), dollars.
C23	R	C	17	ST		INDIR	Traffic servicing expense (cargo), dollars.
C24	R	C	17	ST		INDIR	Reservation and sales expense (first class), dollars.
C25	R	C	17	ST		INDIR	Reservation and sales expense (coach), dollars.

Table A.4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
C26	R	C	17	ST		INDIR	Reservation and sales expense (passenger total), dollars.
C27	R	C	17	ST		INDIR	Reservation and sales expense (property), dollars.
C28	R	C	17	ST		INDIR	Advertising and publicity expense (first class), dollars.
C29	R	C	17	ST		INDIR	Advertising and publicity expense (coach), dollars.
C30	R	C	17	ST		INDIR	Advertising and publicity expense (passenger total), dollars.
C31	R	C	17	ST		INDIR	Advertising and publicity expense (property), dollars.
C32	R	C	17	ST		INDIR	Maintenance expense (ground property and equipment), dollars.
C33	R	C	17	ST		INDIR	Depreciated general ground property and equipment, dollars.
C34	R	C	17	ST		INDIR	Depreciation of maintenance equipment, dollars.
C35	R	C	17	ST		INDIR	General and administrative expense, dollars.
C36	R	C	17	ST			Total indirect operating expense, dollars.
CACH	R	C				DIRECT/ INDIR	Total indirect operating expense, dollars.
CAM	R	C	12/17	ST		DIRECT/ INDIR	Cost per aircraft mile, dollars.
CASH	R	C	12/17	ST		DIRECT/ INDIR	Operating cost per available seat mile, dollars.
CBH	R	C	12/17	ST		DIRECT/ INDIR	Cost per block hour, dollars.
CD	R	A	12	ST		DIRECT	Depreciation on flight equipment, dollars.
CF	R	I	17	ST		INDIR	Passenger trip circuitry factor.

Table A.4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED 3 PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
CF	R	C	25	Y	IN	CFSUB	Net cash flow, dollars.
CFC	R	C	12	ST		DIRECT	Flight crew costs, dollars.
CFH	R	C	12/17	ST		DIRECT/INDIR	Cost per flight hour, dollars.
CFO	R	C	12	ST		DIRECT	Fuel and oil costs, dollars.
CFOP	R	C	12	ST		DIRECT	Total flying operations cost, dollars.
CG	R	C	25,100	Y,A		TAX	Capital gains, dollars.
CGTAX	R	A				TAX	Capital gains tax rate.
CGTX	R	C	25,100	Y,A		TAX	Capital gains tax, dollars.
CI	R	C	12	ST		DIRECT	Insurance costs, dollars.
CLA	R	C	12	ST		DIRECT	Airframe labor costs, dollars.
CLE	R	I	12	ST		DIRECT	Engine labor costs, dollars.
CLF	R	I	17	ST		INDIR	Coach load factor (decimal).
CLS	R	I				DIRECT	Climb speed, mph.
CLT	R	C				DIRECT/INDIR	Climb time, hours.
CM	R	C	12	ST		DIRECT	Total direct maintenance costs, dollars.
CMA	R	C	12	ST		DIRECT	Airframe material costs, dollars.
CMB	R	C	12	ST		DIRECT	Maintenance burden, dollars.
CME	R	C	12	ST		DIRECT	Engines material cost, dollars.
COFL	R	I				DIRECT	Cost of fuel, dollars per pound
COIL	R	I				DIRECT	Cost of oil, dollars per gallon.
COST	R	A	25,100	Y,A	IN	OPLIFE INPUTS	Amount of aircraft price financed, dollars.
CRS	R	I				DIRECT/INDIR	Cruise speed, mph.

Table A.4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED ³ PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
CRT	R	C	12/17	ST		DIREC./INDIR	Cruise time, hours.
CS	R	I				INDIR	Number of coach seats.
CSAREV	R	C			IN	SUM	Cumulative total annual revenue, dollars.
CSCF	R	C			IN	CFSUB	Cumulative total net cash flow, dollars.
CSCM	R	C				DIRECT INDIRECT	Cost per cruise mile per seat, dollars.
CSDCF	R	C			IN	DCFSUB	Cumulative total discounted net cash flow, dollars.
CSDEPR	R	C			IN	SUM	Cumulative total depreciation (double-declining), dollars.
CSEBIT	R	C			IN	SUM	Cumulative total earnings before taxes and interest, dollars.
CSEBT	R	C			IN	SUM TAX	Cumulative total earnings before taxes, dollars.
CSINTI	R	C			IN	SUM	Cumulative total initial investment, dollars.
CSINTX	R	C			IN	SUM, TAX	Cumulative income tax sum, dollars.
CSNTRN	R	C			IN	SUM, TAX	Cumulative total net earnings, dollars.
CSOPCT	R	C			IN	SUM	Cumulative total operating cost, dollars.
CSPRIN	R	C			IN	SUM	Cumulative total yearly principal payment, dollars.
CSTEW	R	I				INDIR	Average number of stewardses in coach.
CSTO	R	C				DIRECT INDIRECT	Cost per takeoff per seat, dollars.
CSYNTR	R	C			IN	SUM	Cumulative total yearly interest payment, dollars.
CTO	R	C				DIRECT INDIRECT	Cost per takeoff, dollars.

Table A.4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE CODE ²	ARRAY DIMENSION	INDEXED 3 PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
D1	R	I				DIRECT	Distance at maximum payload point on range-payload diagram in miles.
D2	R	I				DIRECT	Distance at maximum fuel point on range-payload diagram in miles.
DIS	R	I	17	ST		INDIR	Passenger trip distance, miles.
DCF	R	C	25	Y	IN	DCFSUB	Discounted cashflow, dollars.
DEPR	R	C	25,100	Y,A	IN	DEPSUB	Depreciation (double-declining), dollars.
DEPREC	R	C	25	Y	IN	SUM,TAX	Total depreciation (straight-line and double-declining), dollars.
DESS	R	I				DIRECT	Descent speed, mph.
DEST	R	C				INDIR	Descent time, hours.
DOC	R	C	25,100	Y,A		DIRECT	Direct operating costs per plane per year, dollars.
DOC1	R	C	100	A		DIRECT	Direct operating costs per plane, dollars.
DPT	R	I	17	ST		INDIR	Departures per passenger trip (flight basis).
E8T	R	C	25,100	Y,A	IN	NETSUB	Earnings before taxes, dollars.
EBIAT	R	C	25,100	Y,A	IN	NETSUB	Earnings before interest and taxes, dollars.
ECLIFE	R	A	100	A	IN	INPUTS	Economic life, years.
EXP	R	I	17	ST		INDIR	Fraction of RTP that is express cargo.
F1	R	I				DIRECT	Fuel at maximum payload point on range-payload diagram in pounds.
F2	R	I				DIRECT	Fuel at maximum fuel point on range-payload diagram in pounds.
FCAL	R	C				DIRECT	Coefficient in maintenance labor cost equations.
FCAM	R	C				DIRECT	Coefficient in airframe maintenance material cost equations.

Table A.4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE NODE ²	ARRAY DIMENSION	INDEXED ³ PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
FCEL	R	C				DIRECT	Coefficient in engine maintenance labor cost equation.
FCEN	R	C				DIRECT	Coefficient in engine maintenance materials cost equation.
FCK	R	I				DIRECT	Flight crew cost factor (for a crew of 2) in dollars.
FHAL	R	C				DIRECT	Coefficient in airframe maintenance labor cost equations.
FHAM	R	C				DIRECT	Coefficient in airframe maintenance material cost equation.
FHEL	R	C				DIRECT	Coefficient in engine labor cost equations.
FHEM	R	C				DIRECT	Coefficient in engine maintenance materials cost equation.
FLP	R	I	17	ST		INDIR	First class load factor (decimal).
FLTT	R	C				DIRECT INDIR	Flight time, hours.
FOODR	R	I				INDIR	Food expense factor for first class.
FS	R	I				INDIR	Number of first class seats.
FSTEW	R	I				INDIR	Average number of stewardesses in first class.
GMT	R	I,A				DIRECT INDIR	Ground maneuver time, hours.
H	R	I				DIRECT INDIR	Cruise altitude, feet.
HINFL	R	A	25	Y		INPUTS	Inflation rate applied to price of aircraft.
I	I	LI			IN	OPLIFE	Pointer indicating year number; also miscellaneous loop index.
IECLIF	I	C				DEPSUB	Midyear of economic life.

Table A.4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
II	I	C				DEPSUB	Loop indicator for year value.
INCTAX	R	C	25,100	Y,A	IN	NETSUB	Income tax, dollars.
INTINV	R	A	25,100	Y,A	IN	OPLIFE INPUTS	Initial investment, dollars.
IOC	I	C	25,100	Y,A		INPUTS	Indirect operating costs per plane per year, dollars.
IOC1	I	C	100	A		INDIR	Indirect operating costs per plane, dollars.
IYEAR	I	A			IN	INPUTS	Total number of years under consideration (set at 15).
J	I	LI				REPAY	Loop index.
K	I	A				DIRECT DEPSUB	Counter, loop index.
L	I	LI				DEPSUB	Loop index.
LL	I	C				DEPSUB	Beginning year of second half of economic life.
M	I	LI				OPLIFE INPUTS	Loop indicator for number of aircraft.
MM	I	C				DEPSUB	Loop indicator for year value.
MPLANE	I	I,A				DIRECT INDIR	Aircraft number for particular input data set.
N	I	LI				REPAY	Loop index.
NAC	I	A			IN	INPUTS	Number of aircraft being considered.
NAME	I		122			INPUTS	Not used.
NLIFE	I	C				DEPSUB	Economic life, years.
NLIFE	I	C				REPAY	Economic life, years.
NN	I	C				REPAY	Loop end indicator for number of years.
NRCREW	I	I				DIRECT	Number in crew.

Table A.4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
NRENGN	I					DIRECT	Number of engines.
NRSEAT	I	I				DIRECT INDIR	Number of seats.
NSL	I	I				DIRECT INDIR	Number of stage lengths (up to 12 for DIRECT, up to 17 for INDIR).
NTEARN	R	C	25,100	Y,A	IN	NETSUB	Net earnings, dollars.
MUSTAG	I,A					DIRECT	Flag to reread name list NSTAGE.
OPCOST	R	C	25,100	Y,A	IN	INPUTS	Operating cost, dollars.
PBCOST	R	C				REPAY	Amount of aircraft price financed, dollars.
PRICE	R	A,C	25,100	Y,A	IN	OPLIFE INPUTS	Price of new aircraft, dollars.
PRIN	R	C	25,100	Y,A	IN	REPAY	Yearly principal payment, dollars.
PV	R	C	25	Y	IN	DCFSUB	Present value discount factor
R	R	A			IN	DCFSUB	Interest rate for PV calculation (ROI).
RATE	R	C				DEPSUB	Rate of depreciation.
RCH	R	I				DIRECT INDIR	Rate of climb at cruise altitude, feet per minute.
RCSL	R	I				DIRECT INDIR	Rate of climb (sea level), feet per minute.
RES	R	A	100	A	IN	INPUTS	Residual fraction of aircraft price for salvage.
RL	R	I				DIRECT	Maintenance labor rate, dollars per hour.
RRATE	R	A	100	A	IN	INPUTS	Interest rate for repayment of purchase loan.
RTE	R	C				INDIR	Express tons enplaned.
RTF	R	C				INDIR	Freight tons enplaned.
RTM	R	I				INDIR	Tons of mail carried per flight.

Table A-4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED ³ PARAMETER	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
RTP	R	I				INDIR	Tons of property (cargo) carried per flight.
SALVAG	R	C	25,100	Y,A	IN	DEPSUB TAX	Value of aircraft at end of economic life, dollars.
SAREV	R	C	25	Y	IN	SUM	Yearly total annual revenue, dollars.
SB	R	C	17	ST		INDIR	Block speed, mph.
SCG	R	C	25	Y		TAX	Total yearly capital gains, dollars.
SCCTX	R	C	25	Y		TAX	Total yearly capital gains tax, dollars.
SDEPR	R	C	25	Y	IN	SUM	Yearly total depreciation (double-declining), dollars.
SEBIAT	R	C	25	Y	IN	SUM	Yearly total earnings before interest and taxes, dollars.
SEBT	R	C	25	Y	IN	SUM	Yearly total earnings before taxes, dollars.
SEXP	R	C	25	Y		TAX	Total yearly expenses (excluding taxes), dollars.
SINCOM	R	C	25	Y		TAX	Total yearly earnings before taxes, dollars.
SINCTX	R	C	25	Y	IN	SUM	Yearly income tax sum, dollars.
SL	R	I	12/17	ST		DIRECT/INDIR	Flight stage length in miles.
SNTERN	R	C	25	Y	IN	SUM	Yearly sum of net earnings, dollars.
SOPCST	R	C	25	Y	IN	SUM	Yearly sum of operating costs, dollars.
SPRIN	R	C	25	Y	IN	SUM	Yearly sum of principal payments, dollars.
SSALVG	R	C	25	Y	IN	SUM,TAX	Yearly sum of salvage value, dollars.
SSTDP	R	C	25	Y	IN	SUM	Yearly sum of straight-line depreciation, dollars.
STDEP	R	C	100	A	IN	DEPSUB	Straight line depreciation, dollars.
STLIFE	R	C				DEPSUB	Economic life for straight line depreciation, years.
SYNTRS	R	C			IN	SUM	Yearly sum of interest payments, dollars.
SUM	R	C				DCFSUB	Total accumulated present value of cash flow, dollars.

Table A.4 (Continued)

VARIABLE NAME	TYPE ¹	VALUE MODE ²	ARRAY DIMENSION	INDEXED PARAMETER ³	COMMON BLOCK	DEFINING ROUTINE	DESCRIPTION
T	R	I				DIRECT	Time factor in engine labor cost equations.
TAP	R	C	17	ST		INDIR	Advertising and publicity expense (total), dollars.
TAS	R	C	17	ST		INDIR	Total aircraft servicing expense, dollars.
TB	R	C	17	ST		INDIR	Block time, hours.
TRS	R	C	17	ST		INDIR	Total reservation and sales expense, dollars.
TTS	R	C	17	ST		INDIR	Total traffic servicing expense, dollars.
TXRATE	R	A				INPUTS	Tax rate for income tax.
U	R	I				DIRECT INDIR	Annual utilization per aircraft in block hours.
VA	R	I				DIRECT	Airframe cost, dollars.
VE	R	I				DIRECT	Unit engine cost, dollars.
VT	R	C				DIRECT	Aircraft total cost, dollars.
WA	R	C				DIRECT	Airframe weight, pounds.
WEH	R	C				DIRECT	Aircraft empty weight, pounds.
WEN	R	C				DIRECT	Unit engine weight, pounds.
WGR	R	I				DIRECT	Gross vehicle weight, pounds.
XNAME	A	I	20	DI		DIRECT INDIR	Comment card label for output table headings.
YINTRST	R	C	25,100	Y,A		REPAY	Yearly interest payments, dollars.
YRPPAY	R	C	25,100	Y,A		REPAY	Yearly principal payment, dollars.

Table A.4 (Concluded)

1 Type of variable classification:	2 Value mode classifications:	3 Indexed parameter for arrays:
A - Alphanumeric	C - Value calculated in program	ST - Stage length
I - Integer	I - Read from input file	A - Aircraft type
R - Real	A - Value assigned in program	Y - Number of years
	LI - Loop index	DI - Data items

Appendix B

DESCRIPTION OF PLOTTER SOFTWARE ROUTINES

Appendix B

DESCRIPTION OF PLOTTER SOFTWARE ROUTINES

The plotted output of the Fleet Accounting Module is generated on the ZETA PLOTTER 230, using a CDC 7600 computer. The plotter software used by the Fleet Accounting Module is described in this appendix. This software includes subroutines PLOT, SYMBOL, PLOTS, RSTR, LINAXS, and NUMBER. The descriptions provided here were extracted from internal NASA Ames documentation^{*}. Listings of the source code for the routines are provided in Appendix C.

SUBROUTINE PLOT(X,Y,IPEN)

PLOT performs the basic plotting function of moving the pen from its current location to a specified location with the pen either up or down.

X,Y - The coordinates (in inches) of the point to which the pen is to be moved. The positive X direction is defined as parallel to the margin of the plotting paper and toward the clean roll (package) of paper (i.e., from left to right in the normal fashion as one stands alongside the plotter in the indicated position).

^{*}L. C. Evans, "Memorandum for Systems Studies Division Staff on Plotter Software," NASA-Ames MS:202-8, Moffett Field, California (April 1974).

IPEN - The magnitude of IPEN specifies the operation to be performed:

IPEN = 0 No change in pen position; present pen location is redefined as (X,Y).

- 1 Pen is moved to (X,Y) without raising or lowering the pen.
- 2 Pen is moved to (X,Y) with pen down.
- 3 Pen is moved to (X,Y) with pen up.
- 4 No pen movement; the current location of the pen is returned in (X,Y).
- 5 Initialization call; used by PLOTS.
- 6 Pen is moved to a new "page", the plotter buffer is dumped, and the pertinent parameters are re-initialized, readying the plot software either for termination of the program or for a new plot.
- 7 Used to change the plotting factor (if $X > 0$) or the plotter grain (if $X < 0$ and $Y > 0$); no operation if $X < 0$ and $Y < 0$. This value of IPEN should not be used unless the user is familiar enough with the coding of PLOT to understand what effects changes in these values will have.
- 8 Pen is raised but not moved.
- 9 Pen is lowered but not moved.
- 10 (not currently used).
- 11,12,13,14 Same as IPEN = 1,2,3,4, except that plot is offset, scaled, and rotated.
- 20 Same as IPEN = 13, except that pen position is saved.
- 21,22,23 Same as IPEN = 11,12,13, except that pen movement is calculated relative to the pen position saved from the last call to PLOT with IPEN = 20.

In the case of IPEN = -1,-2,-3,-11,-12,-13,-21,-22,-23, the action is the same as that described above, but after the new pen location has been reached, it is redefined as the origin.

SUBROUTINE SYMBOL(XLLHC,YLLHC,HEIGHT,BCD,ORIENT,NCHAR)

SYMBOL causes a string of alphanumeric information to be plotted.

XLLHC,YLLHC - For the purposes of visualization, each character can be thought of as being drawn inside a rectangle whose size is determined by the value of HEIGHT. The size of the rectangle is the same for all characters involved in a given call to SYMBOL. The value of (XLLHC,YLLHC) gives the coordinates of the lower left-hand corner of the rectangle associated with the first character to be plotted.

HEIGHT - The height of the characters (in inches). The width of each character, for the purposes of determining the length of a plotted character string, is $0.8 \times \text{HEIGHT}$.

BCD - An array containing the character string to be plotted.

ORIENT - The orientation at which the string is to be plotted, measured CCW in degrees from the +X axis.

NCHAR - The number of characters to be plotted.
If NCHAR is less than zero, one of a series of plotting symbols is drawn centered on the point given by (XLLHC,YLLHC). Fifteen symbols are available, and the one drawn is determined by the absolute value of NCHAR (taken modulo 15) according to:

-NCHAR	Plotting Symbol
1	square
2	circle
3	triangle
4	cross ("plus" sign)
5	cross ("X")
6	diamond
7	tilted hourglass
8	upside-down teepee
9	Z
10	Y
11	lozenge
12	asterisk (comb. of 4 & 5)
13	hourglass
14	vertical line
15	star

The standard character set, as represented by the symbols on the 029 keyboard, is available through SYMBOL, with the exception of the following nine symbols: `_ , | , ' , @ , ~ , & , ¢ , # , 0-8-2`.

SUBROUTINE PLOTS(LUN)

PLOTS performs all of the necessary initialization for the plotting software. It should be called prior to calling any other routine in the package.

LUN - Logical unit number. In addition to transmitting this information to the plotting routines, a tape of this number must be declared in the user's PROGRAM statement.

SUBROUTINE RSTR(IARG)

RSTR generates the call CALL PLOT(0.,0.,6) in order to clear the plotting buffers and move to a new page.

IARG - A dummy argument; it is ignored.

SUBROUTINE LINAXS(XO,YO,X1,Y1,LABSIZ,WHICH,NTIC,NLFREQ,VALO,VALMAX,
NDIGIT,NLABEL,LABEL)

LINAXS plots and labels a linear axis.

XO - X coordinate of the left-hand edge of the plot in inches.

YO - Y coordinate of the lower edge of the plot in inches.

X1 - X coordinate of the right-hand edge of the plot in inches.

Y1 - Y coordinate of the upper edge of the plot in inches.

LABSIZ - (REAL) Size of the labels in inches.

WHICH - (INTEGER) Indicates whether a horizontal or vertical axis is
to be plotted:

WHICH = +1 vertical
= -1 horizontal

NTIC - Number of tick marks, including one at the end of the axis.
This is also, therefore the number of intervals into which
the tick marks divide the axis.

NLFREQ - Frequency with which tick marks are to be labeled.

VALO - Value of the axis variable at the beginning of the axis.

VALMAX - Value of the axis variable at the end of the axis.

NDIGIT - Number of significant digits to be used in writing tick mark
labels.

NLABEL - Number of characters in axis label.

LABEL - Array containing the axis label.

SUBROUTINE NUMBER(XLLHC,YLLHC,HEIGHT,A,ORIENT,N)

NUMBER converts a number to its EBCDIC representation and plots it.

XLLHC,YLLHC - The coordinates in inches of the lower left-hand corner of the character string to be plotted. (See the write-up for SYMBOL for a more complete explanation.)

HEIGHT - The height in inches of the number on the plot.

A - The number which is to be converted and plotted; is considered to be an array if N is a FORMAT.

ORIENT - The orientation at which the number is to be plotted.

N - If the absolute value of N is less than 20, then the value of N is the number of decimal places to be included. If the absolute value of N is not less than 20, then N is assumed to be an array containing the FORMAT with which the number(s) in A are to be plotted. If N contains a format, then the element of the array following the end of the format must contain all EBCDIC blanks, and the next word must contain the number of values in the array A which are to be written. The character string generated by the format must be no more than 1000 characters long and must be terminated by an "@" symbol.

If HEIGHT is less than zero, (XLLHC,YLLHC) is assumed to specify the lower right-hand corner of the last character to be plotted.

Appendix C

LISTING OF PROGRAM SOURCE CODE FOR THE
FLEET ACCOUNTING MODULE

```

1      PROGRAM RET (INPUT,OUTPUT,TAPE=INPUT,TAPE6=OUTPUT,
      1      V TAPE2,TAPE4,TAPE7,TAPE8,TAPE9,TAPE11)
      2
      3      RET IS THE MAJOR COMPONENT IN THE FLEET ACCOUNTING MODULE OF THE
      4      ARC-AT MODELS. THE BASIC FUNCTION OF THIS MODULE IS TO PROJECT
      5      THE FLEET COMPOSITION NECESSARY TO MEET TRAFFIC DEMAND. RET ALSO
      6      IS USED TO DRIVE THE AIRFRAME MANUFACTURE MODULE.
      7
      8      DIMENSION TYPE(10),YRINTPOL(1),SEATS(10),SFC(10),SPEED(10),
      9      UTILIZ(10),LIFETIM(10),PLOTS(10)
      10
      11      DIMENSION GROWTH(10),MARKET(10)
      12
      13      DIMENSION MODATA(10),MODYR(10),MSEATS(10),MSEED(10),
      14      MUTILIZ(10),MLIFET(10)
      15
      16      DIMENSION SEATN(10,31),FUELRN(10,31),OPN(10,31),POPUL(10,31)
      17
      18      DIMENSION TOTLMS(10,31),TOTLFL(10,31),TOTLRPM(10,31),TOTLPOP(10,31)
      19
      20      DIMENSION SHRSMIS(10,31),SHRFUEL(10,31),SHRPPMS(10,31),SHRPOP(10,31)
      21
      22      DIMENSION TOTLBUY(10,31),TOTLRT(10,31)
      23
      24      DIMENSION SHRBUY(10,31),SHRRT(10,31)
      25
      26      DIMENSION PLTRPMS(10,31),PLTFUEL(10,31)
      27
      28      DIMENSION MARKET/ MARKET
      29
      30      COMMON /MARKET/ MARKET
      31
      32      COMMON /STATIST/ TYPE,YRINTRO,SEATS,SFC,SPEED,UTILIZ,LIFETIM
      33
      34      *PLOTS
      35
      36      COMMON /MODS/ TYPE,MODYR,MSEATS,MSEFC,MSEED,MUTILIZ,MLIFET
      37
      38      COMMON /RESULTS/ SEATN,FUELRN,OPN
      39
      40      COMMON /POP/ POP,MODYR,MSEATS,MSEFC,MSEED,MUTILIZ,MLIFET
      41
      42      COMMON /INDICES/ NEXPLS,IN,OUT
      43
      44      COMMON /SHARES/ SHRSMIS,SHRFUEL,SHRPPMS,SHRPOP,SHRBUY,SHRRT
      45
      46      COMMON /TOTALS/ SMILES,FRUNED,PPMS,POPULMO,MODYR,NORTA
      47
      48      COMMON /ACCURS/ TOTLMS,TOTLFL,TOTLRPM,TOTLPOP,TOTLBUY,TOTLRT
      49
      49      COMMON /STARTER/ NOCPVS,MARKTYPE,D1,D2,PLTRPMS,PLTFUEL,PERCENT
      50
      50      INTEGER YEAR,YR
      51
      51      INTEGER IN,OUT
      52
      52      REAL LF,PLTFETIM,MARKET,MODATA
      53
      53      REAL MTYPE,MODYR,MSEATS,MSEFC,MSEED,MUTILIZ,MLIFET
      54
      54      SET UP IDENTIFICATION OF LOGICAL UNITS FOR PLOTTING ROUTINES
      55
      55      LUN=9
      56
      56      LUN1=11
      57
      57      REVIND LUN
      58
      58      CALL INIT/4
      59
      59      CALL SETUP(LUN1)
      60
      60      NOCPVS=1
      61
      61      MARKT=1
      62
      62      GO TO 96
      63
      63      CONTINUE
      64
      64      MARKT=MARKT+1
      65
      65      CONTINUE
      66
      66      NO 15 Y=1.1
      67
      67      MTYPE(1)=6.0
      68
      68      15 CONTINUE
      69
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C
C
C      50      READ (5,9) MKTYPE(MKMT)
C              FORMAT (2A,1)
C              IF (END(5)) 999,94
C      94      CONTINUE
C              WRITE (6,7) MKTYPE(MKMT)
C              FORMAT (14,15,16,MARKET = *,2A1J)
C      7        READ RPM GROWTH RATES FOR EACH YEAR FROM 1975-2015
C      70      READ (5,11) GROWTH
C              FORMAT (8F10.0)
C              IF (END(5)) 999,97
C      97      CONTINUE
C              WRITE (6,12)
C              FORMAT(14,15,16,GROWTH*)
C              WRITE (6,13) GROWTH
C              FORMAT(1X,8F10.4)
C              PERCENT=GROWTH(1)
C      11      READ LOAD FACTOR FOR EACH YEAR FROM 1975-2015
C      13      READ (5,21) LF
C              IF (END(5)) 999,93
C              CONTINUE
C              WRITE (6,13)
C              FORMAT(14,15,16,LF *)
C              WRITE (6,15) LF
C      93      READ IN PARAMETERS DEFINING EXISTING AIRCRAFT CURRENTLY IN SERVICE
C      13      READ (5,21) NOEXPLS
C              IF (END(5)) 999,98
C              CONTINUE
C              WRITE (6,12)
C              FORMAT(1X,15,16,EXISTING FLEET *)
C              WRITE (6,21) NOEXPLS
C              NO 10 1=1,NOEXPLS
C      100      READ (5,31) TYPE(1),VRINTD(1),SEATS(1),SFC(1),SPEED(1),UTILIZ(1),
C              LIFETIM(1),PILOTS(1)
C      3        FORMAT(10,6F10.0,4I0)
C      99      IF (END(5)) 999,99
C              CONTINUE
C              WRITE (6,6) TYPE(1),VRINTD(1),SEATS(1),SFC(1),SPEED(1),UTILIZ(1),
C              LIFETIM(1),PILOTS(1)
C      6        FORMAT(1X,11,6F10.4,4I0)
C      110      CONVERT FUEL BURNED FROM POUNDS PER SEAT MILE TO BARRELS PER
C              SEAT MILE
C              SFC(1)=SFC(1)/28.4
C
C      59      RET
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229 229

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41 WRITE (5,41)
 FORMAT(NEW AIRCRAFT TYPES *)
 WRITE (6,2) NNEW
 TNEWEXPLS=1
 OUT=NOEXPLS+NNEW
 IF (NNEW.EQ.0) GO TO 45
 DO 40 I=1,N,OUT,1
 READ (5,3) TYPE(I),YRINTQ(I),SEATS(I),SFC(I),SPEED(I),UTILI7(I),
 LIFETIM(I),PLOTS(I)
 IF (I.EQ.51) 999,104
 CONTINUE
 104 WRITE (5,6) TYPE(I),YRINTQ(I),SEATS(I),SFC(I),SPEED(I),UTILI7(I),
 LIFETIM(I),PLOTS(I)
 SFC(I)=SFC(I)/281.44
 DO 39 J=1,46
 NNOYS(I,J)=0.0
 NRET(I,J)=1.0
 CONTINUE
 38 CONTINUE
 40 CONTINUE
 45 CONTINUE
 FROM INPUT QRY AND RETIREMENT HISTORY, COMPUTE AIRCRAFT
 C POPULATION BY TYPE IN BASE YEAR
 C
 DO 55 I=1,NDEXPLS
 POPUL(I,1)=0.0
 DO 55 J=1,16,1
 POPUL(I,1)=POPUL(I,1)+NNOYS(I,J)-NRET(I,J)
 CONTINUE
 55
 C
 C
 C
 COMPUTE SEAT-MILES/YR/PLANE, REVENUE-PASSENGER MILES/YR/PLANE
 AND FUEL-BURNED/YR/PLANE FOR BASE YEAR 1975.
 DO 53 I=1,NDEXPLS
 SEATM(I,1)=SPEED(I)*UTILI7(I)*SEATS(I)
 REVEN(I,1)=SEATM(I,1)*LF(I)
 FUELRN(I,1)=SFC(I)*SEATM(I,1)
 CONTINUE
 53
 C
 C
 C
 COMPUTE RMS GENERATED FOR BASE YEAR 1975
 DO 57 I=1,N,OUT,1
 POPUL(I,1)=0.0
 CONTINUE
 MARKET(I)=0.0
 DO 60 I=1,NDEXPLS
 MARKET(I)=MARKET(I)+RPM(I,1)*POPUL(I,1)
 CONTINUE
 60
 C
 C
 C
 COMPUTE PROJECTED RMS DEMANDED IN FUTURE YEARS (1975-2051)
 DO 70 I=2,31
 MARKET(I)=(1.0+GROWTH(I)*0.01)*MARKET(I-1)
 CONTINUE
 70
 C
 C
 C
 WRITE (6,71)
 FORMAT(MARKET *)
 71
 C
 C
 C
 WRITE (6,72) MARKET

[illegible]

```

C
C SUBROUTINE IMPLANT IS THE INTERFACE WITH THE AIRFRAME
C MANUFACTURER WHOLFF WHICH ESTIMATES THE AIRCRAFT MANUFACTURING
C DATES OF RETURN ON INVESTMENT
C
C
C CALL IMPLANT
C
C
C CONTINUE
C
C PLTPRMS(MRKT)=SHRPPMS(MRKT,31)
C PLTFUEL(MRKT)=SHRFUEL(MRKT,31)
C CALL PLOTSGL (MRKT)
C
C
C FOR EACH AIRCRAFT TYPE BY YEAR, PRINT CUMULATIVE VALUES FOR
C SEAT MILES FLUM, FUEL CONSUMED, RPMs FLUM, BUY52 AND
C RETIREMENTS
C
C
C ON 450 I=1,OUT,1
C SMILES=0.0
C FURNED=0.0
C RPMs =0.0
C POPULM=0.0
C BUY=0.0
C RETIR=0.0
C WRITE (6,221) TYPE(I)
C WRITE (6,251)
C DO 435 YEAR=1975,2005,1
C VR=YEAR-1974
C SMILES = SMILES+SEATH(I,YR)*POPUL(I,YR)
C FURNED = FURNED+FUELRN(I,YR)*POPUL(I,YR)
C RPMs = RPMs+RPM(I,YR)*POPUL(I,YR)
C BUY=BUY+MRUY52(I,15,YR)
C RETIR=RETIR+MRETR(I,15,YR)
C WRITE (6,452) YEAR,SMILES,FURNED,RPMs,BUY,RETIR
C
C CONTINUE
C
C 435 CONTINUE
C 440 CONTINUE
C
C GO CONSOLE NEXT MARKET
C
C
C GO TO 95
C CONTINUE
C
C MRKT=MRKT-1
C
C PRINT CUMULATIVE FLEET COMPOSITION AND ACTIVITY PROJECTS FOR
C EACH MARKET
C
C
C ON 450 I=1,MRKT
C SMILES=0.0
C FURNED=0.0
C RPMs =0.0
C BUY=0.0
C RETIR=0.0
C WRITE (6,501) MRKT,TYPE(I)
C WRITE (6,451)
C FORMAT(//TI,2,*,YEAR,*,T22,*,ACCUMULATIVE,*,T44,*,ACCUMULATIVE,*,T65,*,
C *ACCUMULATIVE,*,T88,*,ACCUMULATIVE,*,T110,*,ACCUMULATIVE,*,
C *T22,*,SEAT-MILES,*,T44,*,FUEL-BURNED,*,T65,*, RPMs,*,T88,*, = BUY50
C *,T110,*, = RETIRED*)
C WRITE (6,449)

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297 RET
 298 RET
 299 RET
 300 RET
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 342 RET
 343 RET


```

400      CONTINUE
      CONTINUE
      C
      C
      PRINT YEARLY FLEET COMPOSITION AND ACTIVITY PROJECTIONS FOR
      FACH MARKET
      C
      C
      DO 700 I=1,MRKT
      WRITE (6,501) MRKTYP(I)
      WRITE (6,299)
      WRITE (6,298)
      DO 650 YEAR=1975,2005,1
      YEAR=1974
      WRITE (6,301) YEAR,SHRSMJ(I,YR),SHRDEL(I,YR),SHRPHM(I,YR),
      SHRPOP(I,YR),SHRNUV(I,YR),SHRSTIR(I,YR)
      C
      C
      650 CONTINUE
      700 CONTINUE
      WRITE (6,501)
      801 FORMAT(1H1,75L,4TOTALS FOR ALL MARKETS)
      WRITE (6,299)
      WRITE (6,298)
      C
      420      C
      C
      PRINT TOTAL YEARLY FLEET COMPOSITION AND ACTIVITY PROJECTIONS
      FOR ALL MARKETS
      C
      C
      C
      DO 800 YEAR=1975,2245,1
      YEAR=1974
      WRITE (6,301) YEAR,TOTLMS(YR),TOTLFL(YR),TOTLPH(YR),TOTLPOP(YR),
      TOTLNUV(YR),TOTLSTIR(YR)
      C
      C
      800 CONTINUE
      C
      430      C
      C
      CALL SUBROUTINE PLOTTER TO PLOT TOTALS FOR MARKETS
      C
      C
      OPLOT=1
      READ(5,2) OPLOT
      IF(OPLOT.EQ.0) STOP
      CALL PLOTTER (LUN)
      STOP
      END
      C
      C
      435

```

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14/27/79 19.12.83

CTN 4.5+400

5:00PM TIME AMORT17 75/75 OPT=2

1	C	SUBROUTINE AMORT17 (YR)	AMORT17	2
	C	THIS SUBROUTINE RETIRES AIRCRAFT THAT HAVE REACHED	AMORT17	3
	C	RETIREMENT AGE	AMORT17	4
5	C		AMORT17	5
		RTMENSTON TYPE(10),YRINTPO(10),SEATS(J),SFC(2,P),SPEED(10),	AMORT17	6
		X UTIL17(10),LIFETIM(10)	AMORT17	7
		RTMENSTON POPUL(10,31),NOBUYS(10,40),MORET(10,45)	AMORT17	8
10		RTMENSTON PLOTS(10)	AMORT17	9
		RTMENSTON LF(31)	AMORT17	10
		INTEGER YR,OUT,0YEAR,PASS	AMORT17	11
		REAL LF,LIFETIM,NOBUYS,MORET	AMORT17	12
		COMMON /STATIST/ TYPE,YRINTPO,SEATS,SEC,SFC,UTIL17,LF,LIFETIM	AMORT17	13
		9,PLOTS	AMORT17	14
15		COMMON /POP/ NOBUYS,MORET,POPUL	AMORT17	15
		COMMON /INDICES/ NOEXPLS,IN,OUT	AMORT17	16
	C	INITIALIZE BUY AND RETIREMENT COUNTS FOR CURRENT YEAR TO ZERO	AMORT17	17
20	C		AMORT17	18
		ON 5 I=1,OUT	AMORT17	19
		NOBUYS(I,15+YR)=0	AMORT17	20
		MORET(I,15+YR)=0	AMORT17	21
	5	CONTINUE	AMORT17	22
25		ON 30 I=1,OUT	AMORT17	23
	C		AMORT17	24
30	C	FTND THE PAST YEAR SUCH THAT AIRCRAFT BROUGHT IN THAT YEAR HAVE	AMORT17	25
	C	REACHED RETIREMENT AGE, RETIRE THOSE AIRCRAFT	AMORT17	26
	C		AMORT17	27
		0YEAR=15+YR-LIFETIM(I)	AMORT17	28
		IF (0YEAR.LE.0) GO TO 15	AMORT17	29
		MORET(I,15+YR)=NOBUYS(I,0YEAR)	AMORT17	30
35	15	CONTINUE	AMORT17	31
		POPUL(I,YR)=POPUL(I,YR-1)-MORET(I,15+YR)	AMORT17	32
		IF (POPUL(I,YR).LE.0.0) POPUL(I,YR)=0.0	AMORT17	33
		IF ((POPUL(I,YR-1).EQ.0.0).AND.(POPUL(I,YR).EQ.0.0))	AMORT17	34
		X MORET(I,15+YR)=0.0	AMORT17	35
	30	CONTINUE	AMORT17	36
		RETURN	AMORT17	37
		END	AMORT17	38
			AMORT17	39
			AMORT17	40

CIPODDITION. MONS	74/75	1PT=2	FTN 4.5+46.1	14/27/75	3.12.83.	PAGE
6	130	CONTINUE	SPEED(I)=MSPEED(I)	MONS	50	
		IF (MULTI(I).EQ.0.0) GO TO 140		MONS	61	
		MULTI(I)=MULTI(I)		MONS	62	
	140	CONTINUE		MONS	63	
		IF (MULTI(I).EQ.0.0) GO TO 200		MONS	64	
		LIFETIME=LIFETIME		MONS	65	
	200	CONTINUE		MONS	66	
				MONS	67	
	C	COMPUTE SEAT MILES, RPMs, AND FUEL BURN PER AIRCRAFT OF EACH		MONS	68	
	C	TYPE FOR THE CURRENT YEAR		MONS	69	
	C			MONS	70	
		GO 300		MONS	71	
		SEATM(I,YR)=SPEED(I)*MULTI(I)*SEATS(I)		MONS	72	
		RPM(I,YR)=SEATM(I,YR)/YE		MONS	73	
		FUELBURN(I,YR)=SEATM(I)*SEATM(I,YR)		MONS	74	
	300	CONTINUE		MONS	75	
		RETURN		MONS	76	
		END		MONS	77	
75						
79						

```

1      C      SUPPLEMENTARY BUYS (YR)
2      C      THIS SUBROUTINE COMPUTES THE NUMBER OF DEMAND DEMANDS, AND THE
3      C      NUMBER OF DAYS THAT CAN BE GENERATED BY THE CURRENT FLEET, AND
4      C      IF CURRENT FLEET DOES NOT MEET THE DEMAND, PURCHASES THE
5      C      ADDITIONAL AIRCRAFT REQUIRED.
6      C
7      C      INTERPOLATE YR, IN, OUT, YEAR
8      C      DIMENSION TYPE(10), YRINTRO(10), SEATS(10), SPEED(10),
9      C      UTIL(10), LIFE(10), LIFETIME(10)
10     C      DIMENSION PLOTS(10)
11     C      DIMENSION MARKET(10)
12     C      DIMENSION MONUYS(10), FUEL(10), MONRETRILL(10)
13     C      DIMENSION SCATMILL(10), FUEL(10), MONRETRILL(10), POPUL(10),
14     C      PEAL, MARKET, LIFE, LIFETIME, LOAD, CT, MONUYS, MONRETRILL,
15     C      COMMON /STATIST/ TYPE, YRINTRO, SEATS, SPEED, UTIL, LIFE, LIFETIME,
16     C      PLOTS
17     C      COMMON /RESULTS/ SEATINT, FUEL, RPM
18     C      COMMON /MARKET/ MARKET
19     C      COMMON /INDICES/ MONEXPL, IN, OUT
20     C      COMMON /POP/ MONUYS, MONRETRILL, POPUL
21     C      TOTAL=0.0
22     C
23     C      COMPUTE TOTAL RPMs AVAILABLE FOR THIS YEAR
24     C
25     C      ON 100 YR, IN, OUT
26     C      TOTAL=TOTAL+RPM(1), YR, POPUL(1), YR
27     C      CONTINUE
28     C
29     C      COMPUTE DIFFERENCE BETWEEN DEMAND AND AVAILABLE RPMs
30     C
31     C      RPMDIFF=MARKET(YR)-TOTAL
32     C      IF (RPMDIFF.LT.0.0) GO TO 999
33     C      YEAR=YR+1974.
34     C
35     C      DETERMINE WHICH TYPE OF AIRCRAFT IS TO BE PURCHASED
36     C
37     C      NO 200 YR, IN, OUT, J
38     C      J=INT(1+J)
39     C      IF ((YRINTRO(J).GT.YEAR).OR.(YRINTRO(J).EQ.0.0)) GO TO 200
40     C      IT=J
41     C      GO TO 320
42     C      CONTINUE
43     C      RETURN
44     C      CONTINUE
45     C
46     C      ADJUST POPULATION TO ACCOUNT FOR BUYS
47     C
48     C      MONUYS(1:10)=RPMDIFF/RPM(1:10)
49     C      POPUL(1:10)=POPUL(1:10)+MONUYS(1:10)*YR
50     C      IF (POPUL(1:10).LT.0.0) POPUL(1:10)=0.0
51     C      RETURN
52     C      CONTINUE
53     C
54     C      COMPUTE ACTUAL LOAD FACTOR FOR THE CASE OF NO-TO SUPPLY OF
55     C      SEAT-RIES

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C
AC      LOADCT=MARKET(YR)/TOTAL
        Y=AR-YR+1974
        WRITE (5,1) LOADCT,YEAR
        FORMAT(1E10,1F5.3)
        IF (LOADCT-1E10) GOTO 10
        PRINT *, 'WILL SATISFIED RPN REQUIREMENT FOR YEAR ',
        & Y
        PRINT *, 'DEFICTION'
        PRINT *, 'END'
        RIIVS 70
        RIIVS 50
        RIIVS 51
        RIIVS 52
        RIIVS 53
        RIIVS 54
        RIIVS 55
        RIIVS 56
  
```


1 CURVES 2
 2 CURVES 3
 3 CURVES 4
 4 CURVES 5
 5 CURVES 6
 6 CURVES 7
 7 CURVES 8
 8 CURVES 9
 9 CURVES 10
 10 CURVES 11
 11 CURVES 12
 12 CURVES 13
 13 CURVES 14
 14 CURVES 15
 15 CURVES 16
 16 CURVES 17
 17 CURVES 18
 18 CURVES 19
 19 CURVES 20
 20 CURVES 21
 21 CURVES 22
 22 CURVES 23
 23 CURVES 24
 24 CURVES 25
 25 CURVES 26
 26 CURVES 27

1 SURFQUITE CURVES
 2 THIS SURFQUITE SETS UP REVENUE PASSENGER MILES AND FUEL
 3 CONSUMPTION DATA FOR PLOTTING
 4
 5 INTERP YEND
 6 DIMENSION TOTLMTS(31),TOTLFL(31),TOTLRPM(31),TOTLPOP(31)
 7 DIMENSION RPM(30,31),FUELPRM(30,31),DUMMY(15)
 8 DIMENSION TOTLQV(31),TOTLPT(31)
 9 COMMON PLOTDATA/ RPM,FUELPRM
 10 COMMON ACCUMS/ TOTLMTS,TOTLFL,TOTLRPM,TOTLPOP,TOTLQV,TOTLPT
 11 COMMON /STARTER/ MDCRVS,DUMMY
 12
 13 INCREMENT NUMBER OF CURVES
 14 MDCRVS=MDCRVS+1
 15
 16 FOR EACH YEAR, LOAD ELEMENTS OF ARRAYS WITH RPMS FLOWN AND
 17 FUEL BURNED BY THE ASSOCIATED AIRCRAFT TYPE
 18
 19 DO 10 YRNO=1,32,1
 20 RPMS (MDCRVS,YRNO),TOTLRPM(YRNO)
 21 FUELPRM (MDCRVS,YRNO),TOTLFL(YRNO)
 22 CONTINUE
 23 RETURN
 24 END


```

1      C      SIMULATING PLOTTER (LUN)
      C      THIS SIMULATING PLOTS R-V-MUC PASSENGER MILES AND FUEL
      C      CONSUMPTION AIRCRAFT MARKET SHARE CURVES
      C
      C      DIMENSION YEARS(33),CURVE(33)
      C      DIMENSION RPTS(30,33),FUELRPN(30,33)
      C      DIMENSION MKTPOS(6),PLTPOS(31),PLTJFL(33)
      C      INTEGER YRNO
      C      COMMON /STARTER/ YRCPVS,MKTPOS(6),DP,PLTPOS,PLTFUEL,PERCENT
      C      COMMON /PLTDATA/ RPTS,FUELRPN
      C
12     C      PLOTS PERFORMS INITIALIZATION OF THE PLOTTING SOFTWARE
      C
15     C      CALL PLOTS(LUN)
      C      NO 10 YRNO=1,31,1
      C      YEAR=1974+YRNO
      C      YEARS(YRNO)-YEAR
      C      CONTINUE
      C
23     C      READ IN DATA FOR AXIS SCALING
      C
      C      READ (5,2) TOPFARM,NOMRKS,TOPDPMS,NOMRKS0
      C      FORMAT(1E10,4,11)
      C      IF (END(5)) 15,18
      C
25     C      IF NO DATA WAS READ IN, COMPUTE SCALE
      C
      C      CONTINUE
      C      MAYFARM=FUELRPN(NOMRKS,31)/1.E08+.5
      C      TOPFARM=1.E08+MAYFARM
      C      MAYRPTS=RPTS(NOMRKS,31)/1.E11+.5
      C      TOPRPTS=1.E11+MAYRPTS
      C      NOMRKS=4
      C      NOMRKS0=40
      C      CONTINUE
      C      XSC=(2005.-1975.)/2.
      C      YC=1975.
      C      YSC=TOPFARM/5.1
      C
40     C      LINES PLOTS AND LABELS A LINEAR AXIS
      C
      C      CALL LINAXS (1,1,0,0,7,0,125,-1,31,5,1975,20,0,4,4,MYEAR)
      C      CALL LINAXS (1,1,0,0,7,0,125,1, NOMRKS,1,0,0,0,0,0,3,1,1,
      C      X104FUELRPN(0))
      C
45     C      PLOT HEADING
      C
      C      CALL SYMPL(1,7,0,5,1,0,4,54FORCAST 'U.S. AIRLINE FUEL USAGE DISTRI
      C      BUTION,3,0,5)
      C      CALL NUMBER(4,0,7,25,12,0,PERCENT,0,0,1)
      C      CALL SYMPL(4,3,0,7,25,12,0,1747EVR= RPM GPNATH,3,0,15)
      C      CALL PLOT (1,0,1,0,-3)
      C
50     C      PLOT MARKET NAME
      C
      C      CALL SYMPL (3,25,0,50PLTFUEL(1),YSC+.125,0,225,MKTTYPE(1),0,0,6)

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SUBROUTINE SETUP	76/76	OPT=2	STN 4.6+46C	04/27/79	09.12.31	PAGE 1
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1	C	SUBROUTINE SETUP(LUN1)	2
	C	THIS SUBROUTINE INITIALIZES THE PLOTTING SOFTWARE	3
	C	ON LOGICAL UNIT LUN1	4
5	C	CALL PLOTS (LUN1)	5
		RETURN	6
		END	7
			8
			9

Appendix D

**LISTING OF PROGRAM SOURCE CODE FOR THE
AIRFRAME MANUFACTURER MODULE**


```

1      C      SUBROUTINE REVENUE
2
3      C
4      C      THIS SUBROUTINE COMPUTES INCOME IN A PER MONTH BASIS. ASSUMED
5      C      PAYMENT SCHEDULE IS THAT AIRLINES WILL PAY 5% IN ORDER, 70% ON
6      C      DELIVERY AND 25% SPREAD OVER MONTHS BETWEEN ORDER AND DELIVERY
7      C
8      C
9      C      DIMENSION ORDER(372),DELIVER(400)
10     C      DIMENSION INCOME(372),ACCINCO(372)
11     C      INTEGER TOTAL,ORITHPL,DEITHPL
12     C      INTEGER ORDER,DELIVER
13     C      REAL INCOME
14     C      COMMON /INCOME/ INCOME
15     C      COMMON /ACCINCO/ ACCINCO
16     C      COMMON /ORDER/ ORDER
17     C      COMMON /DELIVER/ DELIVER
18     C      COMMON /PRICEO/ PRICEO
19     C      COMMON /TOTAL/ TOTAL
20     C
21     C      INITIALIZE ARRAYS TO ZERO
22     C
23     C      ON 100 TOTAL=1,372,1
24     C      INCOME(1)=0.0
25     C      ACCINCO(1)=0.0
26     C      CONTINUE
27     C
28     C      FOR EACH AIRCRAFT TO BE PRODUCED, ADD REVENUES FROM ORDER
29     C      PAYMENT, PAYMENTS BETWEEN ORDER AND DELIVERY, AND FINAL
30     C      PAYMENT ON DELIVERY.
31     C
32     C      ON 300 ITPL=1,TOTAL,1
33     C      INCOME(ORDER(ITPL))=INCOME(ORDER(ITPL))+.25*PRICEO
34     C      INCOME(DELIVER(ITPL))=INCOME(DELIVER(ITPL))+.70*PRICEO
35     C      ORITHPL=ORDER(ITPL)+1
36     C      DEITHPL=DELIVER(ITPL)-1
37     C      ON 200 MONTHPAY=ORITHPL-DEITHPL,1
38     C      INCOME(MONTHPAY)=INCOME(MONTHPAY)+.25/(DEITHPL-ORITHPL+1)*PRICEO
39     C      CONTINUE
40     C      SUM=0.0
41     C
42     C      CALCULATE THE CUMULATIVE REVENUES FOR EACH MONTH
43     C
44     C      ON 400 MONTH=1,372,1
45     C      SUM=SUM+INCOME(MONTH)
46     C      ACCINCO(MONTH)=SUM
47     C      CONTINUE
48     C      RETURN
49     C      END

```

```

1      SUBROUTINE CASHFLW
2
3      THIS SUBROUTINE DETERMINES INDIVIDUAL MONTHLY CASHFLOW AND
4      CUMULATIVE MONTHLY CASHFLOW FOR ESTIMATED COSTS AND
5      REVENUES
6
7      DIMENSION CASHFLW(1372), ACCASHF(1372)
8      DIMENSION CUST(1372), ACCCUST(1372)
9      DIMENSION RDTTECMP(1372), COMPCST(1372), INCOME(1372)
10     REAL INCOME
11     COMMON /RDTTECMP/ RDTTECMP
12     COMMON /COMPCST/ COMPCST
13     COMMON /INCOME/ INCOME
14     COMMON /ACCCUST/ ACCCUST
15     COMMON /CUST/ CUST
16     COMMON /ACCASHF/ ACCASHF
17     CALL SUBROUTINES TO SPREAD INDIVIDUAL RATE, PRODUCTION COMPONENT
18     COSTS AND REVENUES BY MONTH
19
20     CALL RATE
21     CALL REVENUE
22     CALL COMPCOS
23
24     SUM RATE AND PRODUCTION COST FACTORS PER MONTH AND
25     CUMULATIVE BY MONTH
26
27     SUM=0
28     DO 30 MNT4=1,1372,1
29     SUM1=0.0
30     DO 10 ITH=1,5,1
31     SUM1=SUM1+RDTTECMP(ITH,MNT4)
32     CONTINUE
33     SUM2=0.0
34     DO 20 ITH=1,4,1
35     SUM2=SUM2+COMPCST(ITH,MNT4)
36     CONTINUE
37     COST(MNT4)=SUM1+SUM2
38     INCOME(MNT4)=SUM1+SUM2
39     ACCCUST(MNT4)=SUM
40     CONTINUE
41
42     CALCULATE CASHFLOW EACH MONTH AND CUMULATIVE BY MONTH
43
44     DO 40 MNT4=1,1372,1
45     CASHFLW(MNT4)=INCOME(MNT4)-COST(MNT4)
46     ACCASHF(MNT4)=ACCASHF(MNT4)+CASHFLW(MNT4)
47     CONTINUE
48     RETURN
49     END
50
51
52
53
54

```

16/27/77 09.12.77

CTN 405460

SUBROUTINE INDO 7-776 OPT=2

```

1      SUBROUTINE INDO(M)
2      THIS SUBROUTINE CALCULATES INTERNAL RATE OF RETURN FOR
3      ESTIMATED CASHFLOW
4
5      DIMENSION CASHF(1372),R1MC(1372),PV(31),R0C(31)
6      DIMENSION ACCASHF(372)
7      REAL K,KR,KR1
8      COMMON /CASHFLO/ CASHFLO
9      COMMON /ACCASHF/ ACCASHF
10     PARAMETER (INITIALIZATION)
11
12     KR=0.
13     KR1=0.
14     KR=0.
15     KR1=0.
16     KR=0.
17     KR1=0.
18     KR=0.
19     KR1=0.
20     R1MC=0.
21     IF (ACCASHF(372).LT.0.0) GO TO 16
22     R1MC=0.
23     DELTA=0.001
24     GO TO 20
25     CONTINUE
26     R1MC=0.
27     DELTA=0.001
28
29     MAIN PROGRAM LOOP POINT. CYCLE UNTIL ITERATION STOP CRITERIA
30     ARE MET. CALCULATE DISCOUNTED PRESENT VALUE SUM OF CASHFLOW
31     USING ESTIMATED RATE OF RETURN VALUE.
32
33     CONTINUE
34     DO 30 I=1,31
35     R1MC(I)=0.
36     R1MC(I)=(1.0+R1MC(I))**(-I)
37     CONTINUE
38     DO 40 J=1,372
39     R1MC(J)=R1MC(J)+CASHFLO(J)
40     CONTINUE
41     TS=1
42     DO 42 I=1,31
43     IF (R1MC(I).EQ.0.0) GO TO 42
44     TS=I
45     GO TO 44
46     CONTINUE
47     CONTINUE
48     IL=31
49     DO 46 I=1,31
50     I=32-I
51     IF (R1MC(I).EQ.0.0) GO TO 45
52     IL=I
53     GO TO 49
54     CONTINUE
55     CONTINUE
56     R1MC(I)=(TS+IL)/2
57     R1MC(I)=R1MC(I)
58

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J6/27/73 09.12.31

FTN 4.5+45C

74776 OPT=2

COMPUTIME INDO

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CONTINUE
NL=IL-TC+1
SUMPTMCI(1)
NCF(1)=SUM
DO 50 I=2,NL+1
NCF(I)=NCF(I)+PV(I-1)
SUM=SUM+NCF(I)
CONTINUE
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END

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ORIGINAL PAGE 10
OF POOR QUALITY

```

      SUBROUTINE PLANT
      THIS SUBROUTINE DETERMINES OPTIMAL PRODUCTION, DELIVERY, AND
      ORDER SCHEDULES FOR EACH NEW AIRPLANE TYPE BASED ON DEMAND
      SCHEDULE
      INTEGER LTIM, FIRSTPP, LIFETIME, PROD1, F01
      INTEGER P0702, TS, TTS, EN01, EN02
      INTEGER DELSCHL
      INTEGER ORDER, DELIVER, PRODUCT, PRODSCHL
      INTEGER TACCHN
      INTEGER MAXP, MAXAC
      INTEGER ARRPT, BRPT, KICKOFF, DISCREP
      REAL ACCUM, SUMDIF, AVGRIF
      DIMENSION KICKOFF(29), DTSCREP(29)
      DIMENSION DEMAND(31), PRODUCT(400), PRODSCHL(396), DELIVER(400)
      DIMENSION DELSCHL(396), LTIME(372), ORDER(400)
      COMMON /TTS/ TTS
      COMMON /PRODSCHL/ PRODSCHL
      COMMON /PRODUCT/ PRODUCT
      COMMON /ORDER/ ORDER
      COMMON /DELIVER/ DELIVER
      COMMON /TOTAL/ TOTAL
      COMMON /MAXP/ MAXP
      COMMON /LIFETIME/ LIFETIME
      COMMON /STARTUP/ TS
      COMMON /DEMAND/ DEMAND
      INITIALIZATION OF BASIC PARAMETERS
      DATA LTIME(1), I=1, 372/ 1372024 /
      MAXAC=4000
      TR=2
      LR=LIFETIME
      NTRY=-2
      ON 4 1=1, 29, 1
      KICKOFF(I)=0
      DISCREP(I)=1
      CONTINUE
      MAIN PROGRAM LOOP, DIVIDES TOTAL DEMAND PERIOD INTO ALL
      2-PART COMBINATIONS TO FIND OPTIMAL PRODUCTION RATES
      CONTINUE
      ON 2, 2 99001=19, LB, 1
      FIRSTSTP=ARRPT-1
      TOTAL I=0
      ON 10 1=1, FIRSTSTP
      TOTAL I=TOTAL I+DEMAND(I)
      CONTINUE
      TOTAL=TOTAL+.67
      WRITE (5, 3) TOTAL
      FORMAT( ' TOTAL DEMAND FOR FIRST PERIOD = ', I10)
      SUMDIF=TOTAL I/1.2+(FIRSTSTP)
      PROD1=PROD1+.67
      EN01=FIRSTSTP+.1
      TOTAL 2=0

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34/27/79 19.12.81

CTN 4.6460

SUBROUTINE PLANT 75/75 OPT=2

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60      DO 21 I=1,1,1,LIFETIM
          TOTAL2=TOTAL2+DEMAND(I)
          CONTINUE
          TOTAL2=TOTAL2+.67
          PRDND2=TOTAL2/(12.0*(LIFETIM-FIRSTPR))
          PRDND2=PRDND2+.67
          WRITE (6,4) TOTAL2
          FORMAT(4X,TOTAL DEMAND FOR SECOND PERIOD = ,F10.0)
          MAXD=MAXC(PRND1,PRND2)
          TOTAL=TOTAL1+TOTAL2
          TOTAL=TOTAL+.67
          WRITE (6,5) TOTAL
          FORMAT(4X,TOTAL DEMAND FOR THIS AIRCRAFT = ,F10.0)
          IF(TOTAL.LE.MAXAC) GO TO 21
          WRITE(6,22) MAXAC
          22  FORMAT(1X,*,TOTAL DEMAND EXCEEDS ARRAY MAXIMUM OF ,F15)
          STOP
          21  TOTAL1=12*PRND1+FIRSTPR
              TOTAL2=TOTAL-TOTAL1
              WRITE (6,5) PRND1,PRND2
              FORMAT(4X,PRODUCTION RATE FOR 1ST + 2ND PERIOD RESPECTIVELY IS = ,
                  F15.0,F15.0)
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175      IF(AVGDEL(1,0,0,0) .AVGDEL=50)
176      TTS=TTS+AVGDEL
177      COMPUTE MINMAX VALUE FOR THIS PRODUCTION PERIOD SUBROUTINE
178      MINMAX=MAX((MAXDEL(TARS(IMAX)),TARS(IMIN)),MAX((TARS(IMAX2)),TARS(
179      TARS(IMIN2)))
180      TMAX=0
181      TMIN=0
182      TMAX2=0
183      TMIN2=0
184      NTRY=1
185      GO TO 25
186      C124 CONTINUE
187      KICKOFF(ARKPT-1)=TTS
188      DISCREP(ARKPT-1)=MINMAX
189      NTRY=NTRY+1
190      END OF MAIN PROGRAM LOOP
191      C
192      CONTINUE
193      C303 CONTINUE
194      C
195      EXAMINE MINMAX VALUES TO FIND OPTIMUM PRODUCTION
196      SCHEDULE SUBROUTINE AND OFFSET (SMALLEST MAXIMUM DISCREPANCY)
197      MINMAX=372
198      ON 403 GOTO 19,19,1
199      IF (DISCREP(ARKPT-1).GE.MINMAX) GO TO 403
200      MINMAX=DISCREP(ARKPT-1)
201      ARKPT=ARKPT+1
202      TTS=KICKOFF(ARKPT-1)
203      CONTINUE
204      T=ARKPT
205      LB=ARKPT
206      NTRY=NTRY+2
207      RECYCLE MAIN LOOP ONCE TO CALCULATE OPTIMUM PRODUCTION
208      SCHEDULE WITH OFFSET
209      GO TO 100
210      C
211      CLEANUP CODE AFTER OPTIMUM PRODUCTION SCHEDULE FOUND. PRINT
212      STARTUP TIME AND COMPUTE ORDER SCHEDULE (24 MONTHS BEFORE
213      DELIVERY)
214      GO TO 100
215      C
216      CONTINUE
217      WRITE (6,121) TTS
218      FORMAT(1X,TIME FOR START-UP OF PRODUCTION = *,13)
219      ON 600 IT=PL-1,IT=PL,1
220      ORDER(IT*PL)=DELIVER(IT*PL)-TTS)
221      CONTINUE
222      FORMAT(13)
223      T=IT*PL-372 GO TO 601
224      WRITE(6,603) TTS
225      FORMAT(10X,MONTH FOR STARTUP OF PP FROM (*,13,*) 1* BEYOND END
226      1 OF 1 YEAR PERIOD)
227      603
228      1 OF 1 YEAR PERIOD)
229      1 OF 1 YEAR PERIOD)
230      1 OF 1 YEAR PERIOD)

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PLANT 173
PLANT 174
PLANT 175
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PLANT 228
PLANT 229

COMPONENT PLANT	7/75	OPT-2	FTM 400450	4/27/70	72.12.31	DATE	S
231	C TOP 501 TRIPPOSCHE (372) 50001 GO TO 6.2 WHITE (6.6.24) TOTAL PRODUCTION (372) 404 FORTUITOUS TOTAL DEMAND CALCULATED (0.15.00) CANNOT BE PRODUCED IN 1 31 YEAR PERIOD, /00 TOTAL PRODUCED IN THIS PERIOD IS 0.15.7 20 THIS NEW TOTAL WILL BE USED IN PRODUCTION COSTS FOR PERIOD TOTAL DEMAND: 372 602 BOTTOM	PLANT	231				
232		PLANT	232				
233		PLANT	233				
		PLANT	234				
		PLANT	235				
		PLANT	236				
		PLANT	237				
		PLANT	238				

```
1      C      SUBROUTINE INTDGR(9)
      C
      C      THIS SUBROUTINE PROCESSES MONTHLY ACCUMULATED COSTS, CALLS DATE OF
      C      OPTION CALCULATION AND PRINTS INCURRED COSTS, AND CASHFLOWS
      C      TABLE WHEN PRINT FLAG SET.
      C
      DIMENSION ROTECMP(15,372),COMPST(11,372)
      DIMENSION ACRTES(15,372),ACCPST(11,372)
      DIMENSION ROTELBL(11,372),ACCTELBL(11,372)
      DIMENSION INCOME(1372),ACCTELBL(1372),CASHFLO(1372)
      REAL INCOME
      INTEGER YR,YEAR
      COMMON /ROTECMP/ ROTECMP
      COMMON /COMPST/ COMPST
      COMMON /INCOME/ INCOME
      COMMON /ACCTELBL/ ACCTELBL
      COMMON /CASHFLO/ CASHFLO
      COMMON /ACCPST/ ACCPST
      COMMON /ROTELBL/ ROTELBL
      COMMON /ACCTELBL/ ACCTELBL
      COMMON /PRINT/ IP
      C
      C      CALCULATE ACCUMULATED MONTHLY FUNDS FOR DATE AND PRODUCTION COSTS
      C
      ON 20 YR=1,372
      SUM=0.0
      ON 100 MONTH=1,372
      SUM=SUM+ROTECMP(11,MONTH)
      ACCTELBL(11,MONTH)=SUM
      CONTINUE
      ON 400 YR=1,372
      SUM=0.0
      ON 300 MONTH=1,372
      SUM=SUM+COMPST(11,MONTH)
      ACCTELBL(11,MONTH)=SUM
      CONTINUE
      ON 500 YR=1,372
      SUM=0.0
      ON 300 MONTH=1,372
      SUM=SUM+COMPST(11,MONTH)
      ACCTELBL(11,MONTH)=SUM
      CONTINUE
      C
      CALL SUBROUTINE TO COMPUTE INTERNAL RATE OF RETURN FOR
      ESTIMATED CASHFLOW
      C
      CALL TMR(9)
      WRITE (6,2) R
      FORMAT(6,'R',F10.2)
      IF (IP.EQ.1) GO TO 700
      C
      WHEN REQUESTED, PRINT TABLE OF INDIVIDUAL MONTHLY INCOME
      COST AND CASHFLOW FIGURES FOR EACH YEAR FOR 1973-2025
      C
      ON 500 YR=1,372
      YR=12+YR-1
      YEAR=1974+YR
      WRITE (6,4)
      FORMAT(6,'Y',F10.2)
      CONTINUE
      C
      ON 500 YR=1,372
      YR=12+YR-1
      YEAR=1974+YR
      WRITE (6,5)
      FORMAT(6,'Y',F10.2)
      CONTINUE
      C
```

[illegible]

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				ACQUISITION 174	

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OF POOR QUALITY

PAGE

06/27/79 09.12.31

FIN 6.5+660

76/76 001-2

1	SHARQUINE ACCOST	ACCOST	2
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D-21

50	C	CEHNDL	COMPLEXITY FACTOR FOR LOADING AND HANDLING	ACOST	50
	C	CEHNDL	COMPLEXITY FACTOR HYDRAULIC SYSTEM	ACOST	50
	C	CEHNDL	COMPLEXITY FACTOR INSTRUMENT SYSTEM	ACOST	51
	C	CEHNDL	COMPLEXITY FACTOR LIGHTING GEAR SYSTEM	ACOST	52
	C	CEHNDL	COMPLEXITY FACTOR ENGINE MANIFOLDS	ACOST	53
	C	CEHNDL	COMPLEXITY FACTOR PASSENGER ACCOMMODATIONS	ACOST	54
	C	CEHNDL	COMPLEXITY FACTOR PNEUMATIC SYSTEM	ACOST	55
	C	CEHNDL	COMPLEXITY FACTOR AUXILIARY POWER SYSTEM	ACOST	56
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	C	CEHNDL	FLIGHT TEST OPERATION COST	ACOST	100
	C	CEHNDL	FLIGHT TEST OPERATION COST, INPUT VALUE	ACOST	101
	C	CEHNDL	FLIGHT TEST AIRCRAFT SPARES COST	ACOST	102
	C	CEHNDL	FLIGHT TEST VEHICLES COST	ACOST	103
	C	CEHNDL	CUMULATIVE TOTAL AIRPLANE COSTS FOR 404 UNITS	ACOST	104
	C	CEHNDL	CUMULATIVE TOTAL AIRPLANE COSTS FOR 40-14 UNITS	ACOST	105
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	C	CEHNDL	CUMULATIVE AVERAGE UNIT AIRPLANE COST	ACOST	109
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	C	CEHNDL	GROUND TEST VEHICLES	ACOST	111
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	C	POWER	- INDICATOR FOR QUANTITY MATRIX	ACOST	124
129	C	TO	- INITIAL FLIGHT CREW TRAINING COST FOR MV AIRPLANES	ACOST	125
	C	TO	- TOTAL AIRCRAFT PRODUCTION COSTS FOR MV AIRPLANES	ACOST	126
	C	TO	- INDICATOR FOR LANDING GEAR COMPONENT BREAKDOWN	ACOST	127
	C	TO	- INDICATOR	ACOST	128
	C	LEARN	- AIRFRAME LEARNING CURVE, PERCENT	ACOST	129
	C	LEARN	- AVIONIC LEARNING CURVE, PERCENT	ACOST	130
130	C	LEARN	- ENGINE LEARNING CURVE, PERCENT	ACOST	131
	C	LEARN	- MAXIMUM DESIGN FLIGHT MACH NUMBER FOR ENGINES	ACOST	132
	C	LEARN	- MISCELLANEOUS EQUIPMENT COST	ACOST	133
	C	LEARN	- INDICATOR	ACOST	134
139	C	NCREW	- NUMBER IN FLIGHT CREW, PER AIRPLANE	ACOST	135
	C	NCREW	- NUMBER OF POSITIONS ON LEARNING CURVE	ACOST	136
	C	NCREW	- AT LEAST 1 POSITION, (< 00 = 5)	ACOST	137
	C	NCREW	- NUMBER FLIGHT TEST VEHICLES	ACOST	138
	C	NCREW	- NUMBER OF GROUND TEST VEHICLES	ACOST	139
140	C	NCREW	- NUMBER OF CONCEPT FORMULATION CONTRACTORS	ACOST	140
	C	NCREW	- NUMBER OF CONTRACT DEFINITION ENGINEERS	ACOST	141
	C	NCREW	- NUMBER OF CONCEPT FORMULATION ENGINEERS	ACOST	142
	C	NCREW	- NUMBER OF CONTRACT DEFINITIONS ENGINEERS	ACOST	143
	C	NCREW	- NUMBER OF YEARS FOR CONCEPT FORMULATION	ACOST	144
149	C	NCREW	- NUMBER OF YEARS FOR CONTRACT DEFINITIONS	ACOST	145
	C	NCREW	- TOTAL NUMBER FLIGHT CREW PERSONNEL TO BE TRAINED	ACOST	146
	C	NCREW	- NUMBER OPERATIONAL VEHICLES	ACOST	147
	C	NCREW	- NUMBER OF VEHICLE AT EACH POINT ON LEARNING CURVE	ACOST	148
	C	NCREW	-1 FOR FIRST UNIT COST	ACOST	149
150	C	NCREW	- TOTAL NUMBER FLIGHT TEST - OPERATIONAL VEHICLES	ACOST	150
	C	NCREW	- OPERATIONAL VEHICLES SPARES COST	ACOST	151
	C	NCREW	- PRODUCTION AIRFRAME SPARES	ACOST	152
	C	NCREW	- PRODUCTION ENGINE SPARES	ACOST	153
	C	NCREW	- TRAINING EQUIPMENT COST	ACOST	154
159	C	NCREW	- OPERATIONAL VEHICLES COSTS	ACOST	155
	C	NCREW	- PRODUCTION DEVELOPMENT COST TURN-JET ENGINES	ACOST	156
	C	NCREW	- PRODUCTION DEVELOPMENT COST TURN-JET ENGINES, INPUT VA	ACOST	157
	C	NCREW	- TOTAL NUMBER OF PASSENGERS	ACOST	158
160	C	NCREW	- TOTAL PRODUCTION SYSTEM COST, F.O.M.C.	ACOST	159
	C	NCREW	- TOTAL NUMBER OF VEHICLES MANUFACTURED IN QUANTITY MATRI	ACOST	160
	C	NCREW	- VEHICLE PRODUCTION RATE, NUMBER/MONTH	ACOST	161
	C	NCREW	- CONTRACTOR FEE, NOT - 6 PHASE	ACOST	162
	C	NCREW	- RESEARCH, DEVELOPMENT, TESTING - ENGINEERING COST	ACOST	163
	C	NCREW	- ENGINEERING LABOR RATE, \$/HR.	ACOST	164
	C	NCREW	- TOOLING LABOR RATE, \$/HR.	ACOST	165
169	C	NCREW	- TOTAL MANUFACTURING SUSTAINING COSTS FOR 404 UNITS	ACOST	166
	C	NCREW	- SUSTAINING ENGINEERING COSTS	ACOST	167
	C	NCREW	- SUSTAINING TOOLING COST	ACOST	168
	C	NCREW	- SUBSYSTEM DEVELOPMENT COST	ACOST	169
179	C	NCREW	- SUBSYSTEM DEVELOPMENT COST, INPUT VALUE	ACOST	170
	C	NCREW	- THRUST PER ENGINE - TPEREN (SEA LEVEL)	ACOST	171
	C	NCREW	- PRODUCTION AIRCRAFT TECHNICAL DATA COST	ACOST	172

175	C	TDP	* TOTAL TECHNICAL DATA COST	ACOST	173
	C	TMC	* TOTAL OF ALL MANUFACTURING COSTS FOR 100 UNITS	ACOST	174
	C	TOOLC	* COMPLEXITY FACTOR TOOLING	ACOST	175
	C	TOUERO	* TOTAL ENGINE THRUST OVER AIRPLANE TAKE OFF GROSS WEIGHT	ACOST	176
	C	TPEDEH	* THRUST IN POUNDS PER ENGINE	ACOST	177
	C	TROTE	* TOTAL RESEARCH, DEVELOPMENT, TOOLING & ENGINEERING COST	ACOST	178
	C	TROTEC	* ARRAY OF RDTF COST FACTORS	ACOST	179
	C	TRI	* INITIAL TRANSPORTATION COST	ACOST	180
	C	TST	* TOOLING AND SPECIAL EQUIPMENT COST	ACOST	181
	C	TTLCMP	* ARRAY OF PRODUCTION COST FACTORS	ACOST	182
	C	VMPX	* MAXIMUM VEHICLE SPEED, KNOTS	ACOST	183
	C	VA	* VEHICLE AMPR WEIGHT	ACOST	184
	C	WACS	* AIR CONDITIONING SYSTEM WEIGHT	ACOST	185
	C	WAEON	* AERODYNAMIC CONTROL SYSTEM WEIGHT	ACOST	186
	C	WAMTC	* ANTI-ICING SYSTEM WEIGHT	ACOST	187
	C	WAVION	* AVIONICS SYSTEM WEIGHT	ACOST	188
	C	WAVINT	* TOTAL AVIONICS AND INSTRUMENT WEIGHT	ACOST	189
	C	WADNY	* FUSELAGE WEIGHT	ACOST	190
	C	WE	* AIRCRAFT EMPTY WEIGHT	ACOST	191
	C	WELCAD	* ELECTRIC POWER CONVERSION & DISTRIBUTION SYSTEM WEIGHT	ACOST	192
	C	WEMP	* EMPENNAGE WEIGHT	ACOST	193
	C	WENACC	* ENGINE ACCESSORIES WEIGHT	ACOST	194
	C	WENG5	* ENGINES TOTAL WEIGHT	ACOST	195
	C	WFUSYS	* FUEL SYSTEM WEIGHT	ACOST	196
	C	WFRUIT	* TOTAL FUEL WEIGHT	ACOST	197
	C	WGROSS	* AIRCRAFT GROSS TAKE-OFF WEIGHT = WC + WTD	ACOST	198
	C	WHANDL	* LOAD AND HANDLING SYSTEM WEIGHT	ACOST	199
	C	WHYCAD	* HYDRAULIC POWER CONVERSION & DISTRIBUTION SYSTEM WEIGHT	ACOST	200
	C	WMNST	* INSTRUMENT SYSTEM WEIGHT	ACOST	201
	C	WLG	* ALIGHTING GEAR SYSTEM WEIGHT	ACOST	202
	C	WLGCON	* ALIGHTING GEAR CONTROLS WEIGHT	ACOST	203
	C	WIGSTP	* ALIGHTING GEAR STRUCTURE WEIGHT	ACOST	204
	C	WLGTRS	* TIRE WEIGHT	ACOST	205
	C	WLGUHL	* WHEELS AND BRAKE WEIGHT	ACOST	206
	C	WMACEL	* ENGINE MACHINES WEIGHT	ACOST	207
	C	WPACCO	* PASSENGER ACCOMMODATIONS (AND EQUIPMENT) WEIGHT	ACOST	208
	C	WPAYL	* PAYLOAD WEIGHT	ACOST	209
	C	WPMCAD	* PNEUMATIC POWER AND DISTRIBUTION SYSTEM WEIGHT	ACOST	210
	C	WPOMW	* AUXILIARY POWER SYSTEM WEIGHT	ACOST	211
	C	WPPROV	* CREW SIZE RELATED SUBSYSTEM DEVELOPMENT COST FACTOR	ACOST	212
	C	WTREVS	* THRUST REVERSER WEIGHT	ACOST	213
	C	WWMG	* WING WEIGHT	ACOST	214
	C	WAVN	* AVIONICS DEVELOPMENT FACTOR, NOMINAL VALUE = C.10	ACOST	215
	C	WVCTTY	* FINAL ASSEMBLY-CHECK OUT COST FRACTION, NOMINAL VALUE	ACOST	216
	C	Y	* MISCELLANEOUS EQUIPMENT DEVELOPMENT FACTOR, VALUE = 1.	ACOST	217
	C	Z	* AIRFRAME PRODUCTION LEARNING CURVE COST FACTOR	ACOST	218
	C	ZA	* AVIONICS PRODUCTION LEARNING CURVE COST FACTOR	ACOST	219
	C	ZETA	* AIRFRAME LEARNING CURVE EXPONENT	ACOST	220
	C	ZETA1	* AVIONICS LEARNING CURVE EXPONENT	ACOST	221
	C	ZETAP	* ENGINE LEARNING CURVE EXPONENT	ACOST	222
	C	ZP	* ENGINE PRODUCTION LEARNING CURVE COST FACTOR	ACOST	223
				ACOST	224
				ACOST	225
				ACOST	226
				ACOST	227
				ACOST	228
				ACOST	229

REAL LEARN, LEARNP, IV, MEO, IT, MACH
 REAL NVENP, NGP, NFA, MV, MCREW, MPL, NUNF, LEAPNA
 REAL NNOENP, NNOHNS, NNOENP, NNOENP, NNOENP, NNOENP
 DIMENSION TROTEC 131, TTLCMP (11)
 COMMON /TROTEC / TROTEC


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345      T      = 0
      TCUM      = 0
      Z      = 0

      C MANUFACTURING COST BREAKDOWN

350      TPEREN=1
      TOVER=1
      TOVER=1
      VOTIE(5,20)=TOVER*EN
      SIO FORMAT(CTOVER=0.012,5,5X,EN = 0, E12.5)

      C LEARNING CURVE EXPONENT
      Z ZETA=1.0+(ALOG(0.01)/LEARN)/ALOG(2.0)

      C
      IF (LEARN.EQ.0.0) ZETA = ZETA
      IF (LEARN.EQ.0.0) GO TO 9
      ZETA = 1.0 + (ALOG(0.01)/LEARN)/ALOG(2.0)
      Z ZETAP=1.0+(ALOG(0.01)/LEARN)/ALOG(2.0)

360      C FIRST OF TWO MAJOR PROGRAM LOOPS. ONE CYCLE FOR EACH
      C POSITION ON LEARNING CURVE.

      C
      GO TO 999 NO2,NDATA

365      C
      ON 3332 J=1,37
      3332 C(J)=J.

      Z = NVEHIN)*ZETA
      ZA = NVEHIN)*ZETA
      ZP = NVEHIN)*ZETAP
      APPROV = NCREW * 500. - 500.

370      C
      AIRPLANE STRUCTURE

375      C
      1. WING GROUP
      CWING = 35000. * WING**0.451 * Z * CFMING
      2. TAIL GROUP
      CTMP = 10230. * WEMP**0.451 * Z * CFEMP
      3. BODY GROUP
      CBODY=56100.*WBODY**0.451*Z*CFBODY
      4. ALIGHTING GEAR GROUP STRUCTURE
      FLG = 10430. * WLG**0.541 * Z * CFLG
      5. NACELLE GROUP
      CNACEL=56130.*WNACEL**0.451*Z*CFNAC
      6. PROPULSION GROUP
      ENGINES
      CTJ = 2770.*T**0.60*Z*CFENG
      IF (CTJ.NF.1.0) CTJ = CTJ
      CENG = CTJ * EN

380      C
      THRUST REVERSER
      CTREVS = 2800.0 * WTRVS**0.766 * Z * CFTRREV

385      C
      ENGINE ACCESSORIES
      CFENAC = 1090.0 * WENACC * Z * CFENAC
      FUEL SYSTEM
      CFUSYS = 51.9 * WUSYS * Z * CFFUSY
      PROPULSION GROUP TOTAL

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POOR QUALITY


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515 C 7 END CUMULATIVE PRINTTYPE A/C, 9 END SUPERSONIC PRODUCTION A/C
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33 IF (IPRGN.EQ.0)
 34 *TST=PT6.1994A+1.062*TMIC+1.E-6

575 C
 C
 C ATOMACT TYPE FLIGHT TEST PROGRAM
 C FLIGHT TEST OPERATIONS(FY0)
 34 FTD=90*NFV+1.10 VMAX+991.E-6 * WGRDSS+0.0
 C IF (FTDI .NE.0.1) FTD=FTDI

580 C
 C AGE(AGEP)
 C AGE=0.15*FV+(5*ANDE
 C TF (AGEPI .NE. 0.1) AGE= AGEPI

585 C
 C TECHNICAL DATA(TDP)
 C TDP=0.02*FV

590 C
 C GROUND TEST SPARES(GTS)
 C GTS = GTV + GTSPAR

595 C
 C FLIGHT TEST SPARES(FS)
 C FTS= FV + FVSPAR

599 C
 C INITIAL INVESTMENT(IV)
 C

600 C
 C OPERATIONAL VEHICLE(OV)
 C 120 CONTINUE

605 C
 C 125 CAVO = NVHF+ZETA + AMFG
 C CAVO = CAVION + NVHF+ZETA
 C CPO = PRPU + NVHF+ZETA
 C 130 NV = (CAVO + CAGD + CPO) * (1.0 + XFASSY)

609 C
 C AGE
 C AGE0 = 0.15 + NV
 C TF (AGE0 .NE. 0.1) AGE0 = AGE01

613 C
 C SPARES(NS)
 C NSA = AFSPAD + (CAVO + CAVO)
 C NSP = PRPU + ((NVHF + (1.0+ENSPAD)))*ZETA - NVHF+ZETA
 C NS = NSA + NSP
 C IF (IMPS.EQ.1) NS = J.

615 C
 C FACILITIES(FAC) INPUT
 C FAC = FAC1

619 C
 C SUSTAINING ENGINEERING(SEE)
 C SE = ADDE + (NVHF+0.20 -1.0)
 C IF (IMPS.EQ.1) SE=1.0*OV

623 C
 C SUSTAINING TOOLING(ST)
 C ST = TST + ((NVHF) ** 0.14 - 1.0)
 C SN TO 176

625 C
 C TECHNICAL DATA(TDN)
 C 176 TNN = J+J2 + JV
 C

ORIGINAL PAGE IS
 OF POOR QUALITY

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639	C	MISCELLANEOUS EQUIPMENT(MEQ)	ACCOST	639
		MEQ = 500 + MEQ * 10.001 - (C6)	ACCOST	639
		IF (1705.EQ. 1) MEQ = 0	ACCOST	639
639	C	TRAINING EQUIPMENT (TE)	ACCOST	639
		TE = 1.442E-1 * NV + MVIF * (1.4525)	ACCOST	639
		IF (1705.EQ. 1) TE = 0	ACCOST	639
639	C	INITIAL TRAINING (IT)	ACCOST	639
		IT = MEQ * 0.5	ACCOST	639
		IF (1705.EQ. 1) IT = 0	ACCOST	639
640	C	INITIAL TRANSPORTATION (TI)	ACCOST	640
		TI = 0.005 * (DV + OS + MEQ + OT + AGE)	ACCOST	640
640	C	TV = NV + OS + FAC + ST + SE + TDN + MEQ + OT + IT + TRI + AGE	ACCOST	640
645	C	NAVIC RESEARCH, DEVELOPMENT, AND ENGINEERING COSTS	ACCOST	645
		NOTE = CF + CD + DDOL + SUMSYS + AD + PATJ + GTV + GTS + FTS + TST	ACCOST	645
		++ FTM + AGE + TOP	ACCOST	645
650	C	DOFEE = DOFEE + FEE	ACCOST	650
		TOTAL RESEARCH, DEVELOPMENT, TEST, AND EVALUATION COSTS	ACCOST	650
		TRDTE = DOFEE + DOFEE	ACCOST	650
		IF (ICUM * GT. 1) GO TO 190	ACCOST	650
655	C	CONTRACTOR FEE	ACCOST	655
		AOFEE = IV * FEE	ACCOST	655
660	C	TOTAL PRODUCTION COSTS	ACCOST	660
		AO = IV + AOFEE	ACCOST	660
		NS = GTV + GTS + FTS + TST + ST + SE + TDN + AGE + TOP	ACCOST	660
		IF (ICUM * EQ. 1) GO TO 196	ACCOST	660
190		CAVUCA = AFG + 0.012 * ZETA - 1.0	ACCOST	660
		CAVUCA = CAVINM + 0.012 * ZETA - 1.0	ACCOST	660
		CAVUCA = PRNU + 0.012 * ZETA - 1.0	ACCOST	660
		FWICA = (CAVUCA + CAVUCA + CAVUCA) * (1.0 + FEAASY) + (1.0 + FEE)	ACCOST	660
		CAVCT = AFG + 7	ACCOST	660
		CAVCT = CAVINM + 7	ACCOST	660
		CAVCT = PROP + 7	ACCOST	660
		FVCT = (CAVCT + CAVCT + CAVCT) * (1.0 + FEAASY) + (1.0 + FEE)	ACCOST	660
		IF (ICUM * EQ. 1) GO TO 195	ACCOST	660
		CAVCTI = AFG + (1.0 - 1.0) * ZETA	ACCOST	660
		CAVCTI = CAVINM + (1.0 - 1.0) * ZETA	ACCOST	660
		CAVCTI = PRNU + (1.0 - 1.0) * ZETA	ACCOST	660
		FVCTI = (CAVCTI + CAVCTI + CAVCTI) * (1.0 + FEAASY) + (1.0 + FEE)	ACCOST	660
		FVUC = FVCT - FVCTI	ACCOST	660
195		IF (ICUM * EQ. 1) FVUC = FVCT	ACCOST	660
		SC = 1.14 - NV + (1.0 + FEE)	ACCOST	660
		TMC = FVCT + TRDTE + SC	ACCOST	660
		AO = TMC / 0	ACCOST	660
196		CALL COST	ACCOST	660
		T = T + 1	ACCOST	660
		IF (1444 * GT. MEV) NV = 0 - MEV	ACCOST	660
		MYME = 0	ACCOST	660
		IF (1444 * GT. MEV) NV = 0 - MEV	ACCOST	660


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WRITE(6,50J) (C(24,M),M=1,NDATA)
WRITE(6,570J) (C(23,M),M=1,NDATA)
WRITE(6,59J) (C(22,M),M=1,NDATA)
WRITE(6,71J) (C(22,M),M=1,NDATA)
WRITE(6,710J) (C(20,M),M=1,NDATA)
WRITE(6,630J) (C(18,M),M=1,NDATA)
WRITE(6,62J) (C(15,M),M=1,NDATA)
WRITE(6,72J) (C(13,M),M=1,NDATA)
WRITE(6,72J) (C(12,M),M=1,NDATA)
35 FORMAT(1H, 32X,
144HREARDOON OF CUMULATIVE MANUFACTURING COST,
220HILLIONS OF DOLLARS
73 800 FORMAT(1H, 52HBODY STRUCTURE
15F14.4)
801 FORMAT(1H, 52HENGINE
15F14.4)
802 FORMAT(1H, 52HEMPENNAGE
15F14.4)
809 FORMAT(1H, 52HNUMBER OF VEHICLES
15F14.4)
2102 FORMAT(1H, 52HENGINE GEAR
15F14.4)
2000 FORMAT(1H, 52HPOPULATION SYSTEM
134 (, F11.4, 1H), 4(24 (, F11.4, 1H))
2076 FORMAT(1H, 52H ENGINE ACCESSORIES
15F14.4)
2902 FORMAT(1H, 52H ENGINES
15F14.4)
3000 FORMAT(1H, 52H THRUST REVERSER
15F14.4)
3102 FORMAT(1H, 52HMACELLES, PODS, PYLONS, SUPPORTS
15F14.4)
3300 FORMAT(1H, 52HPNEUMATIC
15F14.4)
3602 FORMAT(1H, 52H FUEL SYSTEM
15F14.4)
4202 FORMAT(1H, 52HAEODYNAMIC CONTROLS (SURFACE CONTROLS)
15F14.4)
4902 FORMAT(1H, 52HAUXILIARY POWER SOURCE
15F14.4)
5402 FORMAT(1H, 52HELECTRICAL POWER CONVERSION AND DISTRIBUTION
15F14.4)
5900 FORMAT(1H, 52H EQUIPMENT
15F14.4)
5702 FORMAT(1H, 52H INSTRUMENTATION
134 (, F11.4, 1H), 4(24 (, F11.4, 1H))
5902 FORMAT(1H, 52H INSTALLATION
15F14.4)
6002 FORMAT(1H, 52HMAIN CONDITIONING
15F14.4)
6102 FORMAT(1H, 52HANTI-ICING
15F14.4)
6202 FORMAT(1H, 52HLOAD AND HANDLING
15F14.4)
6302 FORMAT(1H, 52H INSTALLATION
15F14.4)

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COSTO 113
COSTO 114
COSTO 115

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115 15F14.4)
66.1 FORMATT14.524PASSENGER ACCOMMODATIONS
15F14.4)
711. FORMATT14.524HAUTONICS
134 (6, 511.4, 14), 4124 (6, 511.4, 14))
7103 FORMATT14.524 EQUIPMENT
15F14.4)
7201 FORMATT14.524VEHICLE TOTAL
15F14.4)
7205 FORMATT14.524FINAL ASSEMBLY AND CHECKOUT
15F14.4)
C
1 WRITE (5,12)100IE,ADD,CF,CD,POEL
192 FORMATT14.524CONSTITUTIONS OF GALLARSI/
1414 RESEARCH,DEVELOPMENT,TEST,AND EVALUATION,77X,F12.2/
214 10X,434HAUTONICS DESIGN AND ENGINEERING DEVELOPMENT,37X,F12.2/
314 20X,194CONCEPT FORMULATION,31X,F12.2/
414 20X,194CONTRACT DEFINITION,31X,F12.2/
5 14 20X,224HAUTONICS ENGINEERING,33X,F12.2)
WRITE(6,193) SUBSYS
193 FORMATT
14 10X,354SUBSYSTEMS DEVELOPMENT , 45X, F12.2)
WRITE(6,194) AO, POTJ, NS
194 FORMATT
614 10X,244HAUTONICS DEVELOPMENT,60X, F12.2/
714 10X,224CONSTITUTION DEVELOPMENT,58X, F12.2/
814 10X,194DEVELOPMENT SUPPORT, 61X, F12.2)
WRITE(6,200) NG, GTV
200 FORMATT
114 20X,224GROUND TEST VEHICLES (6,6.1, 14), 23X,F12.2)
WRITE (6,301) GTS, FTS,TST, FTO, AGEF, TOP, ROPFE
WRITE (6,440) C(13,1), C(10,1), C(11,1), C(12,1)
WRITE (6,435) C(13,1)
WRITE (6,443) AO, VUMF, OV
301 FORMATT
214 20X,194GROUND TEST SPARES,32X,F12.2/
414 20X,184FLIGHT TEST SPARES, 32X, F12.2/
514 20X,344MODELING AND SPECIAL TEST EQUIPMENT,16X,F12.2/
714 20X,224FLIGHT TEST OPERATIONS, 28X,F12.2/
114 20X,504GROUND SUPPORT EQUIPMENT
15F12.2/
214 20X,144TECHNICAL DATA,36X,F12.2/
414 10X, 344EEF, 77X, F12.2//)
445 FORMATT 14 ,
3 204AIRCRAFT PRODUCTION, 58X, F12.2/
414 10X,224OPERATIONAL VEHICLES (6,6.1, 14), 51X, F12.2)
435 FORMATT
414 30X, 304FINAL ASSEMBLY AND CHECKOUT , 141, F12.2, 141//)
440 FORMATT (14 ,
134,274MANUFACTURING--FIRST UNIT,23X,14(F12.2,141//
414 30X,144HAUTONICS ,16X,14(F12.2,141//
714 30X,204HAUTONICS ORCIPMENT,14(F12.2,141//
914 30X 224CONSTITUTION DEVELOPMENT,EX,14(F12.2,141//
WRITE(5,403)
170

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FTM 4.06460C

00T=2

SUBROUTINE COST00

305 AFAC = SE ST, ACEN, TOR, MEO, NT, IT, TPI,

0 ADPE
0 CORMAT

175 514 10X, 64SPACES, 74X, F12.2/
514 10X, 134FACILITIES, 70X, F12.2/
514 10X, 224SUSTAINING ENGINEERING, 58X, F12.2/
714 10X, 134SUSTAINING TOOLING, 62X, F12.2/
114 10X, 34GROUND SUPPORT EQUIPMENT, 50X, F12.2/
214 10X, 14TECHNICAL DATA, 66X, F12.2/
314 10X, 234MISCELLANEOUS EQUIPMENT, 57X, F12.2/
414 10X, 104TRAINING EQUIPMENT, 62X, F12.2/
514 10X, 154INITIAL TRAINING, 64X, F12.2/
614 10X, 224INITIAL TRANSPORTATION, 59X, F12.2/
714 10X, 34FEE, 77X, F12.2//

TOTAL=TRITE+AO

AVIAC = TOTAL/AVIAC

WRITE (6,415) TOTAL, NVME, AVIAC

415 FORMAT(1H2, 10HTOTAL COST, 107X, F12.2//

141HTOTAL NUMBER OF FLIGHT VEHICLES PRODUCED, 77X, F12.2//

22740AVERAGE UNIT AIRPLANE COST, 91X, F12.2)

WRITE (5,416)

416 FORMAT(1H1)

700 IF (11000 .GT. 1) GO TO 735

WRITE HEADINGS ONLY FOR SECOND CALL FROM ACCOST

WRITE (6,710)

WRITE (6,720) W, MACH, WGRDSS, NV, PATE, NEV, PPI, FEEI

WRITE (6,730)

710 FORMAT (14, 40X, AIRPLANE COST VERSUS QUANTITY *//)
720 FORMAT (14, 15X, AMPR WEIGHT (LBS), F10.2, 8X, AMACH NO.,
1 F7.2, 4X, TAKE-OFF GROSS WT. (LBS), F10.2// 5X, AMO. AIRCRAFT,
2F8.0, 4X, PRODUCTION RATE AC/MO, F9.2, 4X, TEST AC, F5.0, 4X,
3PROGRAM TYPE, 2X, AU, 4X, PERCENT PROFIT, F8.1//)
730 FORMAT (14, 12X, AVERAGE MANUFACTURING COSTS,
1 T75, 8X, D T = E, T40, SUSTAINING,
2 T101, AVERAGE//
3 T2, QUANTITY, T13, AIRFRAME, T24, PRODUCTION,
4 T37, AVIATION, T40, UNIT COST, T42, CUMULATIVE, T77,
5 ACOSTS, TOR, ACOSTS, T103, ACOST//)

WRITE COST VERSUS QUANTITY DATA FOR ALL CALLS AFTER THE
FIRST (ACCOST)

735 WRITE (6,740) 10F11, CAPUCA, CPUCA, CAUACA, EVICA, TONTE,
1 SC, AP
740 FORMAT (14, 12, 16, T11, F10.4, T22, F10.4, T33, F10.4,
1747, F8.2, T61, F10.2, T74, F10.2, T87, F10.2, T90, F10.2)

503 RETURN
END

COST00 173
COST00 174
COST00 175
COST00 176
COST00 177
COST00 178
COST00 179
COST00 180
COST00 181
COST00 182
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COST00 227

10.12.31

FTN 4-6-66C

76/76 OPT-2

SUBROUTINE UNIT04

```

1      C      SUBROUTINE UNIT04
      C      THIS SUBROUTINE COPIES SOME OF THE CARD INPUT TO DISK UNITS
      C      FUTURE USE
5      C      DIMENSION CARD(8)
      C      CONTINUE
      C      READ (5,11) CARD
      C      IF (EOF(11)) 90,20
      C      CONTINUE
10     C      TITLE INFORMATION IS WRITTEN ON UNIT 4 FOR USE IN SUBROUTINE
      C      CDSPE
      C
15     C      WRITE (4,11) CARD
      C      FORMAT(8A12)
      C      GO TO 10
      C      CONTINUE
20     C      CONTINUE
      C      READ (5,11) CARD
      C      IF (EOF(11)) 100,40
      C      CONTINUE
      C
25     C      PRICE INFORMATION IS WRITTEN ON UNIT 6 FOR USE IN SUBROUTINE
      C      IMPLANT
      C
30     C      WRITE (6,11) CARD
      C      GO TO 30
      C      CONTINUE
      C      REWIND 6
      C      RETURN
      C      END

```

UNIT04 2
UNIT04 3
UNIT04 4
UNIT04 5
UNIT04 6
UNIT04 7
UNIT04 8
UNIT04 9
UNIT04 10
UNIT04 11
UNIT04 12
UNIT04 13
UNIT04 14
UNIT04 15
UNIT04 16
UNIT04 17
UNIT04 18
UNIT04 19
UNIT04 20
UNIT04 21
UNIT04 22
UNIT04 23
UNIT04 24
UNIT04 25
UNIT04 26
UNIT04 27
UNIT04 28
UNIT04 29
UNIT04 30
UNIT04 31
UNIT04 32
UNIT04 33
UNIT04 34

Appendix E

**LISTING OF PROGRAM SOURCE CODE FOR THE
AIR CARRIER MODULE**

ORIGINAL PAGE IS
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```
1  PROGRAM OPLIFE (INPUT,OUTPUT,TAPE5=INPUT,TAPE6=OUTPUT)
C  PROGRAM OCFROI (INPUT, OUTPUT, TAPE5=INPUT, TAPE6=OUTPUT)
C  OCF/ROI ECONOMIC MODEL
C
5  MAIN CONTROL FOR AIR CARRIER MODULE
C
C  DIMENSION NAME (122)
C  LEVEL 2,
X  AREV, CSEB, CF, CSINTL, CSAREV, CSCF, CSINTN, CSDEPR,
10 1CSEB, CSEB, DCF, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR,
2CSEB, DCF, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR,
31, INCTAX, INCTAX, INCTAX, INCTAX, INCTAX, INCTAX, INCTAX,
4PRIN, PV, PV, PV, PV, PV, PV, PV, PV, PV, PV, PV, PV, PV,
5SEB, SINTX, SINTX, SINTX, SINTX, SINTX, SINTX, SINTX, SINTX,
6SSTDEP, SDEP, SDEP, SDEP, SDEP, SDEP, SDEP, SDEP, SDEP,
7COST, PRICE, PRICE, PRICE, PRICE, PRICE, PRICE, PRICE,
8BVALUE
C  DIMENSION AREV(25,100),CF(25),COST(25,100),
10 1DCFX(25),DEPR(25,100),ECLIFE(25,100),ECLIFE(100),
2INCTAX(25,100),INCTAX(25,100),INCTAX(25,100),OPCOST(25,100),
3PRIN(25,100),PV(25),RES(100),RATE(100),SALVAG(25,100),
4SDEP(25,100),YNTRST(25,100),
5SAREV(25),SSALVAG(25),SINTX(25),SPRIN(25),SOPCST(25),SDEPR(25),
6SSTDEP(25),SEB(25),SYNTR(25),SEB(25),SINTX(25),SNTERN(25),
7DEPR(25),PRICE(25,100),CSCF(25),
8SEB(25),SINCOM(25),CSCF(25,100),BVALUE(25,100),SCG(25),
9CGTX(25,100),SCGTX(25)
C  COMMON/IN/AREV,
10 1CSEB, CSEB, CF, CSINTL, CSAREV, CSCF, CSINTN, CSDEPR,
2CSEB, DCF, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR,
31, INCTAX, INCTAX, INCTAX, INCTAX, INCTAX, INCTAX, INCTAX,
4PRIN, PV, PV, PV, PV, PV, PV, PV, PV, PV, PV, PV, PV, PV,
5SEB, SINTX, SINTX, SINTX, SINTX, SINTX, SINTX, SINTX, SINTX,
6SSTDEP, SDEP, SDEP, SDEP, SDEP, SDEP, SDEP, SDEP, SDEP,
7COST, PRICE, PRICE, PRICE, PRICE, PRICE, PRICE, PRICE,
8BVALUE
C  REAL INTINV,INCTAX,NTEARN
C
C  SET INITIAL AIRCRAFT PRICE AND CALL INPUTS
C
C  PRICE(1,1)=500000.
C  CALL INPUTS(NAME)
C  DO 31 N=1,NAC
45  WRITE (6,32) (INTINV(I,N),I=1,IYEAR)
32  FORMAT (1, INTINV=,10E12.4)
31  CONTINUE
C
C  MAIN LOOP POINT - CYCLE MAIN PROGRAM FOR SUCCESSIVE AIRCRAFT
C  PRICES
C
C  WE HAVE ASSUMED THE RATIOS 2.0 AND 1.5 USED IN THE NEXT TWO LINES
50 39 INTINV(1,1)=PRICE(1,1)/2.0
COST(1,1)=1.5*INTINV(1,1)
C
C  CALL REPAY, DEPR, AND NETSUB FOR EACH YEAR UNDER STUDY.
C  THEN CALL REST OF SUBROUTINES FOR THIS CYCLE
C
```

34/26/73 09.41.54

FTN 4.60463

PROGRAM OPLIFE 76/76 OPT-1

```

60      DO 2 I=1,IYEAR
        CALL REPAY
        CALL DEPSUB
        CALL NETSUB
        2 CONTINUE
        CALL SUM
        CALL CFSUB
        CALL TAX
        CALL CFSUB
        CALL DCFSUB
        CALL OUTPUT
        C
        C INCREMENT AIRCRAFT PRICE AND RECYCLE UNTIL PRICE LIMIT IS REACHED
        C
        PRICE(1)=PRICE(1)+2500000.
        IF(PRICE(1).GT.30000000.) GO TO 29
        GO TO 39
        29 STOP
        END
75
76      OPLIFE 59
77      OPLIFE 60
78      OPLIFE 61
79      OPLIFE 62
80      OPLIFE 63
81      OPLIFE 64
82      OPLIFE 65
83      OPLIFE 66
84      OPLIFE 67
85      OPLIFE 68
86      OPLIFE 69
87      OPLIFE 70
88      OPLIFE 71
89      OPLIFE 72
90      OPLIFE 73
91      OPLIFE 74
92      OPLIFE 75
93      OPLIFE 76
94      OPLIFE 77

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```

1      SUBROUTINE INPUTS(NAME)
C      THIS SUBROUTINE CAUSES THE INITIALIZATION OF BASIC PARAMETERS AND
C      THE DIRECT AND INDIRECT OPERATING COST VALUES
C
5      DIMENSION
2MINFL(25),DOC(100),IO(1,100),DOC(25,100),IOC(25,100)
REAL INTINV,INCTX,NTERN
DIMENSION NAME(122)
LEVEL 2
X AREV,CF,CSAREV,CSCF,CSDCF,CSDPR,CSEBT,CSEBT,CSEBT,CSEBT,CSEBT,CSEBT,
1CSNTR,CSPCT,CSPRM,CSTPR,DCP,DEPR,DEPR,DEPR,DEPR,DEPR,DEPR,DEPR,DEPR,
21,INCTX,INTINV,IYEAR,MAC,NTERN,OPCOST,PRIM,PV,R,SALVAG,SAREV,
3SDEPR,SEBIAT,SEBT,SINCTX,SINTX,SMTERN,SOPCT,SPRIM,SSALVG,
4SSTDEP,STDEP,SYN,RS,YNTRST,COST,PRICE,RES,RATE,THRATE,BVALUE
DIMENSION AREV(25,100),CF(25),CST(25,100),DCP(25),DEPR(25,100),
1EBIAT(25,100),EBT(25,100),ECLIFE(100),INCTX(25,100),
2INTINV(25,100),NTERN(25,100),OPCOST(25,100),PRIM(25,100),
3PV(25),RES(100),RATE(100),SALVAG(25,100),STDEP(25,100),
4YNTRST(25,100),SAREV(25),SSALVG(25),SINTX(25),SPRIM(25),
5SOPCT(25),SDEPR(25),SSTDEP(25),SEBIAT(25),SYNTRST(25),SEBT(25),
6SINCTX(25),SMTERN(25),DEPR(25),PRICE(25,100),CSDCF(25),
7SEXP(25),SINCON(25),CG(25,100),BVALUE(25,100),SCG(25),
8CGTX(25,100),SCGTX(25)
COMMON/IN/AREV,CF,CSAREV,CSCF,CSDCF,CSDPR,CSEBT,CSEBT,CSEBT,CSEBT,
1CSNTR,CSPCT,CSPRM,CSTPR,DCP,DEPR,DEPR,DEPR,DEPR,DEPR,DEPR,DEPR,DEPR,
2ECLIFE,INCTX,INTINV,IYEAR,MAC,NTERN,OPCOST,PRIM,PV,R,
3SALVAG,SAREV,SDEPR,SEBIAT,SEBT,SINCTX,SINTX,SMTERN,SOPCT,SPRIM,
4SSALVG,SSTDEP,STDEP,SYNTRST,COST,PRICE,CSEBT,THRATE,
5BVALUE
LEVEL 2, DOC,IOC
COMMON/DUC/DOC
COMMON/IOC/IOC
C      INITIALIZE BASIC AIR CARRIER MODULE PARAMETERS
C
C      IYEAR=15
C      MAC=1
C      THRATE=0.48
C      WE HAVE ASSUMED THE RATIOS 2.0 AND 1.5 USED IN THE NEXT 2 LINES
C      INTINV(1,1)=PRICE(1,1)/2.5
C      COST(1,1)=1.5*INTINV(1,1)
C      DO 20 I=1,IYEAR
C      MINFL(I)=1.0
C      AREV(I,1)=7500000.
C      CONTINUE
C      DO 11 M=1,MAC
C      DO 10 I=2,IYEAR
C      COST(I,M)=0.
C      INTINV(I,M)=0.
C      PRICE(I,M)=PRICE(I-1,M)+IMFL(I)
C      PRICE(I,M)=PRICE(I-1,M)+IMFL(I)
C      CONTINUE
C      DO 31 M=1,MAC
C      CONTINUE
C      WRITE (6,32) (INTINV(I,M),I=1,IYEAR)
C      FORMAT (6,10E12.6)
32

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INPUTS 2
 INPUTS 3
 INPUTS 4
 INPUTS 5
 INPUTS 6
 INPUTS 7
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 INPUTS 55
 INPUTS 56
 INPUTS 57
 INPUTS 58

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FTN 4.04460

SUBROUTINE INPUTS 76/76 OPT=1

```

31 CONTINUE
DO 30 M=1,NAC
ECONOMIC LIFE OF EACH AIRCRAFT ASSUMED TO BE 15 YEARS
ECLIFE(M)=15.
RES(M)=0.15
KRATE(M)=0.1
CONTINUE
30
C
C CALL SUBROUTINES TO INPUT AND CALCULATE DIRECT AND INDIRECT
C OPERATING EXPENSES
C
C CALL DIRECT(DOCL,NAC)
C CALL INDIR(IOC1,NAC)
C
C APPLY INFLATION FACTOR TO DIRECT AND INDIRECT COSTS, THEN SUM
C THEM INTO OPCOST AND PRINT THE SUM
C
DO 50 M=1,NAC
DOCL(M)=DOCL(M)
IOC1(M)=IOC1(M)
DO 40 I=2,IYEAR
DOCL(I,M)=DOCL(I-1,M)*HINFL(I)
IOC1(I,M)=IOC1(I-1,M)*HINFL(I)
CONTINUE
40
50 CONTINUE
DO 60 M=1,NAC
DO 70 I=1,IYEAR
OPCOST(I,M)=DOCL(I,M)+IOC1(I,M)
CONTINUE
60 CONTINUE
WRITE(6,91)
FORMAT (2E10.0 OPCOST(6) *)
DO 80 M=1,NAC
WRITE(6,91) (OPCOST(I,M),I=1,IYEAR)
51 FORMAT(1H *10F12.0)
80 CONTINUE
RETURN
END

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FTM 4.6460

76/76 OPT-1

SUBROUTINE DIRECT

```

C      RATE OF CLIMB-SEA LEVEL-RCSL, RATE OF CLIMB-M-RCH
C      CLT=20H/(RCSL+RCH)*60.0)
C      DESCENT TIME - DEST
C      ASSUME THAT DESCENT TIME EQUALS CLIMB TIME
C      DEST=CLT
C
C      CYCLE POINT FOR EACH STAGE UNDER STUDY (SOME STAGE LENGTH
C      DEPENDENT CALCULATIONS)
C      DO 10 I=1,MSL
C      CRUISE TIME - CRT(I)
C      CRTI=((SL(I)+0.03*SL(I))-((CLS*CLT+DESS*DEST))/CRS
C      IFSL(I).GT.1400.) CRTI=((SL(I)+0.02*SL(I)+20.0)-
C      1(CLS*CLT+DESS*DEST))/CRS
C      GROUND MANEUVER TIME
C      AIR MANEUVER TIME
C      BLOCK TIME - BT(I)
C      BTI)=GMT+CLT+DEST+CRTI)+AMT
C      BS(I)=SL(I)/BT(I)
C      BLOCK FUEL COMPUTATION
C      FL=0.1,F2=0.92 ARE DATA POINTS ON THE RANGE-PAYLOAD DIAGRAMS
C      FL=0.1 IS THE MAX. PAYLOAD POINT AND F2=0.2 IS THE MAX. FUEL POINT
C      BLOCK FUEL - BF(I)
C      BF(I)=WGR*0.01*((F1-F2)/(02-01))+SL(I)*0.033*((F1-F2)/(02-01))+CMS
C      BF(I)=11200.
C      BF(I)=19800.
C      BF(I)=20000.
C      BF(I)=34300.
C
C      FLIGHT TIME COMPUTATION
C      FLT(I)=BT(I)-GMT
C      FLIGHT TIME - FLT(I)
C
C      CALCULATION OF THE OPERATING COSTS
C      FLYING OPERATIONS COSTS
C      FLIGHT CREW COSTS
C      CFC(I)=(WGR*0.0E-5*FCK)/BS(I)
C      FUEL AND OIL COSTS
C      CFO(I)=1.02*1 COFL*BF(I)+NRENG*0.133* COIL*BT(I)/SL(I)
C      INSURANCE COSTS
C      CII)=0.001*BT/(U*BS(I))
C      TOTAL FLYING OPERATIONS COSTS
C      CFP(I)=CFC(I)+CFO(I)+CII(I)
C      DIRECT MAINTENANCE - FLIGHT EQUIPMENT COSTS
C      LABOR - AIRFRAME
C      CLAI)=BL*(FHAL*FLT(I)+FCAL)/SL(I)
C      CMAT(I)=IFHAM*FLT(I)+FCAM)/SL(I)
C      LABOR - ENGINE
C      CLEI)=BL*(FHEL*FLT(I)+FCEL)/SL(I)
C      CMAT(I)=IFHEM*FLT(I)+FCEN)/SL(I)
C      MAINTENANCE BURDEN
C      CMB(I)=1.0*(CLA(I)+CLE(I))
C      TOTAL DIRECT MAINTENANCE COSTS
C      CMII)=CLAI(I)+CMAT(I)+CLE(I)+CHE(I)+CMB(I)
C      DEPRECIATION - FLIGHT EQUIPMENT
C      COI(I)= 0.

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115 C COST PER AIRCRAFT MILE
      CAM(I)=CFOP(I)*CM(I)+CD(I)
116 DIRECT
117 DIRECT
118 C COST PER FLIGHT HOUR
      CFM(I)=CAM(I)*BS(I)*OT(I)/FLT(I)
119 DIRECT
120 C COST PER BLOCK HOUR
      CBM(I)=CAM(I)*BS(I)
121 DIRECT
122 C DIRECT OPERATING COST PER AVAILABLE SEAT MILE
      CASH(I)=CAM(I)/NRSEAT
123 DIRECT
124 CONTINUE
125 WRITE (6,904)
126 FORMAT (0.0,0.0 PER AIRCRAFT MILE *)
127 DIRECT
128 C LOOP POINT TO WRITE DIRECT OPERATING COSTS PER BLOCK HOUR AFTER
      COSTS PER AIRCRAFT MILE ARE WRITTEN
129 DIRECT
130 C
      WRITE (6,911) XNAME
131 WRITE (6,97) H
132 WRITE (6,91) (SL(I),I=1,MSL)
133 WRITE (6,92) (OS(I),I=1,MSL)
134 WRITE (6,93) (OT(I),I=1,MSL)
135 WRITE (6,94) (FLT(I),I=1,MSL)
136 WRITE (6,95) (CAT(I),I=1,MSL)
137 WRITE (6,96) (OP(I),I=1,MSL)
138 WRITE (6,100) UP, RL
139 WRITE (6,101) (CFC(I),I=1,MSL)
140 WRITE (6,102) (CFOP(I),I=1,MSL)
141 WRITE (6,103) (CI(I),I=1,MSL)
142 WRITE (6,104) (CFOP(I),I=1,MSL)
143 WRITE (6,105) (CLA(I),I=1,MSL)
144 WRITE (6,106) (CMA(I),I=1,MSL)
145 WRITE (6,107) (CLE(I),I=1,MSL)
146 WRITE (6,108) (CME(I),I=1,MSL)
147 WRITE (6,109) (CMB(I),I=1,MSL)
148 WRITE (6,110) (CM(I),I=1,MSL)
149
150 C SKIP REST OF WRITE STATEMENTS WHEN WRITING COSTS PER BLOCK HOUR
151 C
      IFIX=11212,214,214
212 WRITE (6,112) (CAM(I),I=1,MSL)
      WRITE (6,113) (CFM(I),I=1,MSL)
      WRITE (6,114) (CBM(I),I=1,MSL)
      WRITE (6,115) (CASH(I),I=1,MSL)
213 FORMAT (0.0,0.0 STAGE LENGTH (MILES)
           0.10F8.0)
91 FORMAT (0.0,0.0 BLOCK SPEED (MPH)
           0.10F8.0)
92 FORMAT (0.0,0.0 BLOCK TIME (HOURS)
           0.10F8.3)
93 FORMAT (0.0,0.0 FLIGHT TIME (HOURS)
           0.10F8.3)
94 FORMAT (0.0,0.0 CRUISE TIME (HOURS)
           0.10F8.3)
95 FORMAT (0.0,0.0 CRUISE FUEL (POUNDS)
           0.10F8.0)
96 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
97 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
98 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
99 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
100 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
101 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
102 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
103 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
104 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
105 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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106 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
107 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
108 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
109 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
110 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
111 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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           0.10F8.0)
117 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
118 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
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           0.10F8.0)
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122 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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123 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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131 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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139 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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140 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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141 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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142 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
143 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
144 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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145 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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146 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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147 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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149 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
150 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
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           0.10F8.0)
152 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
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153 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
154 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
155 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
156 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
157 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
158 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
159 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
160 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
161 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
162 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
163 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
164 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
165 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
166 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
167 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
168 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
169 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
170 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
171 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)
172 FORMAT (0.0,0.0 CRUISE ALTITUDE (FEET)
           0.10F8.0)

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1 0.10F6.3)
106 FORMAT( MATERIAL AIRFRAME 0.10F6.3)
107 FORMAT( LABOR ENGINES 0.10F6.3)
108 FORMAT( MATERIAL ENGINES 0.10F6.3)
109 FORMAT( MAINT. BURDEN 0.10F6.3)
110 FORMAT( TOTAL MAINTENANCE 0.10F6.3)
111 FORMAT( TOTAL DIRECT OPERATING COST/100, 0.10F6.3)
112
113 AIRCRAFT MILE 0.10F6.3)
114 FORMAT( 1/FLIGHT HOUR 0.10F6.3)
115 FORMAT( 1/BLOCK HOUR 0.10F6.3)
116
117 COST PER CRUISE MILE FOR THE AIRCRAFT
118 CACH=(CAM(2)+SL(2))-CAM(1)+SL(1)/(SL(2)-SL(1))
119
120 COST PER TAKEOFF FOR THE AIRCRAFT
121 CTO=(CAM(1)-CACM)+SL(1)
122
123 COST PER CRUISE MILE PER SEAT
124 CSCM=CACM/MRSEAT
125
126 COST PER TAKEOFF PER SEAT
127 CSTO=CTO/MRSEAT
128
129 WRITE (6,116) CTO,CACM
130 WRITE (6,117) CSTO,CSCM
131
132 FORMAT (00.0 COST PER AIRCRAFT TRIP = 0.0F7.2.0 PLUS 0.0F6.2,
133 10/MILE)
134
135 FORMAT (00.0 COST PER SEAT TRIP = 0.0F5.2.0 PLUS 0.0F6.4.0/MILE)
136
137 CALCULATION OF THE OPERATING COSTS PER BLOCK HOUR
138 K=1
139
140 SECOND CYCLE FOR STAGE LENGTH DEPENDENT CALCULATIONS
141
142 FLYING OPERATIONS COSTS
143 FLIGHT CREW COSTS
144 DO 500 I=1,NSL
145 CFC(I)=(MGR(3.0E-5)+FCK)
146 FUEL AND OIL COSTS
147 CFC(I)=11.02*(CFL(1)+HREN(1)+0.135*COIL(1))/SL(I)+BS(I)
148
149 INSURANCE COSTS
150 CII=0.001*VT/U
151
152 TOTAL FLYING OPERATIONS COSTS
153 CFOP(I)=CFC(I)+CFC(I)+CII(I)
154
155 DIRECT MAINTENANCE - FLIGHT EQUIPMENT COSTS
156 LABDR - AIRFRAME
157 CLAI=(IRL*(FHAL+FLT(I)+FCAL)/SL(I)+BS(I)
158
159 MATERIAL - AIRFRAME
160 CHAI=(FHAM+FLT(I)+FCAM)/SL(I)+BS(I)
161
162 LABOR - ENGINE
163 CLE=(IRL*(FHEL+FLT(I)+FCEL)/SL(I)+BS(I)
164
165 MATERIAL - ENGINE
166 CHE=(FHER+FLT(I)+FCEM)/SL(I)+BS(I)
167
168 MAINTENANCE BURDEN
169 CMH(1)=0.0*(CLAI)+CLE(I)
170
171 TOTAL DIRECT MAINTENANCE COSTS
172 CMH=(CLAI)+CHAI)+CLE(I)+CHE(I)+CMH(I)
173
174 FIRST STAGE LENGTH SHOULD BE FROM "FD BOOK
175 DOC(1)=CMH(1)+CFOP(1)+U
176
177 CONTINUE
178
179 WRITE (6,130)
180
181 FORMAT (0.0,0.0 PER BLOCK HOUR)
182
183
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04/26/79 09.41.54

FTM 4.6+460

SUBROUTINE DIRECT 76/76 OPT=1

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230      C      TRANSFER TO WRITE COSTS PER CLOCK HOUR
      C
      C      GO TO 505
      C      MPLANE=MPLANE+1
      C      IF(MPLANE.GT.M1) GO TO 23
      C
      C      TRANSFER UNTIL ALL AIRCRAFT HAVE BEEN PROCESSED
      C
      C      GO TO 909
      C      RETURN
      C      END
240
230      DIRECT
231      DIRECT
232      DIRECT
233      DIRECT
234      DIRECT
235      DIRECT
236      DIRECT
237      DIRECT
238      DIRECT
239      DIRECT
240      DIRECT
241      DIRECT

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09.01.54

04/26/79

FTN 4.6+460

76/76 OPT=1

SUBROUTINE INDIR

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60      WRITE(6,RES)
        READ(5,ADV)
        WRITE(6,ADV)
        READ(5,MAINT)
        WRITE(6,MAINT)
        READ(5,GENADH)
        WRITE(6,GENADH)
        READ(5,GENADH)
        WRITE(6,GENADH)
        C BLOCK TIME COMPUTATION TB(I)
        C CLIMB TIME-CLT TO ALTITUDE-H
        C RATE OF CLIMB-SEA LEVEL-RCSL, RATE OF CLIMB-H-RCH
        C CLT=2*H/(RCSL+RCH)*60.0)
        C DESCENT TIME - DEST
        C ASSUME THAT DESCENT TIME EQUALS CLIMB TIME
        C DEST-CLT
        C
        C CYCLE POINT FOR EACH STAGE UNDER STUDY (SOME STAGE LENGTH
        C DEPENDENT CALCULATIONS)
        C
        C DO 10 I=1,MSL
        C   SL(I)=DIS(I)
        C   CRUISE TIME - CRT(I)
        C   CRT(I)=(SL(I)+0.015*SL(I))-(CLS*CLT+DESS*DEST))/CRS
        C   IF(SL(I)-67.1400.) CRT(I)=(SL(I)+0.02*SL(I)+20.0)-
        C   1(CLS*CLT+DESS*DEST))/CRS
        C   GROUND MANEUVER TIME
        C   GMT=0.16
        C   AIR MANEUVER TIME
        C   AMT=0.16
        C   BLOCK TIME - TB(I)
        C   TB(I)=GMT+CLT+DEST+CRT(I)+AMT
        C   BLOCK SPEED - SB(I)
        C   SB(I)=SL(I)/TB(I)
        C   FLIGHT TIME COMPUTATION
        C   FLIGHT TIME - FLT(I)
        C   FLT(I)=TB(I)-GMT
        C   DERIVATION OF THE ELEMENTS OF THE IOC EQUATION
        C   STEWARDESS EXPENSE: FIRST CLASS
        C   C6(I)= A15+GSTEW*TB(I)
        C   STEWARDESS EXPENSE COACH
        C   C7(I)= A15+GSTEW*TB(I)
        C   TOTAL STEWARDESS EXPENSE
        C   C8(I)= C6(I)+C7(I)
        C   FOOD EXPENSE FIRST CLASS
        C   C9(I)=A16+FS*FLP(I)+FDDDR*TB(I)
        C   FOOD EXPENSE COACH
        C   C10(I)=A16+CS*CLF(I)+TB(I)
        C   TOTAL FOOD EXPENSE
        C   C11(I)= C9(I)+C10(I)
        C   OTHER PASSENGER IN FLT EXPENSE FIRST CLASS
        C   C12(I)= A17+FS*FLP(I)+JIS(I)+CF(I)
        C   OTHER PASSENGER IN FLT EXPENSE COACH
        C   C13(I)=A17+CS*CLF(I)+JIS(I)+CF(I)
        C   TOTAL OTHER PASSENGER IN FLT EXPENSE
        C   C14(I)=C12(I)+C13(I)
        C   AIRCRAFT SERVICING EXPENSE, LINE SERVICING
        C   C15(I)=A18+OPT(I)
        C   AIRCRAFT SERVICING EXPENSE, CONTROL

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FTN 4.6+463

SUBROUTINE INDIR 76/76 DPT=1

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1 C26(I)+C27(I)+C30(I)+C31(I)+C32(I)+C33(I)+C34(I)+C35(I)
C FIRST STAGE LENGTH SHOULD BE FROM AED BOOK
C IOL(PLANE)-C36(I)/T8(I)
C INDIRECT COST PER AIRCRAFT MILE
C CAN(I)-C36(I)/SL(I)
C INDIRECT COST PER FLIGHT HOUR
C CM(I)-C36(I)/FLT(I)
C INDIRECT COST PER BLOCK HOUR
C CM(I)-C36(I)/T8(I)
C INDIRECT COST PER AVAILABLE SEAT MILE
C CASH(I)-CAN(I)/MRSEAT
C CONTINUE
10
C
C WRITE OUTPUT TABLE OF INDIRECT OPERATING COSTS PER BLOCK HOUR
C FOR EACH STAGE FOR EACH AIRCRAFT
C
WRITE(6,512) XNAME
WRITE(6,511) (SL(I),I=1,MSL)
WRITE(6,511) (SM(I),I=1,MSL)
WRITE(6,511) (FT(I),I=1,MSL)
WRITE(6,511) (FLT(I),I=1,MSL)
WRITE(6,511) (OPT(I),I=1,MSL)
WRITE(6,511) (CF(I),I=1,MSL)
WRITE(6,511) (CLF(I),I=1,MSL)
WRITE(6,511) (FLF(I),I=1,MSL)
WRITE(6,511) (C1(I),I=1,MSL)
WRITE(6,511) (C2(I),I=1,MSL)
WRITE(6,511) (C4(I),I=1,MSL)
WRITE(6,511) (C6(I),I=1,MSL)
WRITE(6,511) (C11(I),I=1,MSL)
WRITE(6,511) (C14(I),I=1,MSL)
WRITE(6,511) (C15(I),I=1,MSL)
WRITE(6,511) (C16(I),I=1,MSL)
WRITE(6,511) (C17(I),I=1,MSL)
WRITE(6,511) (TAS(I),I=1,MSL)
WRITE(6,511) (C22(I),I=1,MSL)
WRITE(6,511) (C23(I),I=1,MSL)
WRITE(6,511) (TTS(I),I=1,MSL)
WRITE(6,511) (C26(I),I=1,MSL)
WRITE(6,511) (C27(I),I=1,MSL)
WRITE(6,511) (TRS(I),I=1,MSL)
WRITE(6,511) (C31(I),I=1,MSL)
WRITE(6,511) (C32(I),I=1,MSL)
WRITE(6,511) (TAP(I),I=1,MSL)
WRITE(6,511) (C33(I),I=1,MSL)
WRITE(6,511) (C34(I),I=1,MSL)
WRITE(6,511) (C36(I),I=1,MSL)
WRITE(6,511) (CAN(I),I=1,MSL)
WRITE(6,511) (CFM(I),I=1,MSL)
WRITE(6,511) (CASH(I),I=1,MSL)
91 FORMAT(10X, 'PASSENGER TRIP DISTANCE(MILES)')
92 FORMAT(10X, 'BLOCK SPEED (MPH)')
93 FORMAT(10X, 'BLOCK TIME (HOURS)')

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FTM 4.6+463

SUBROUTINE INDIA 76776 OPT=1

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230 94 FORMAT(10, FLIGHT TIME (HOURS))
95 FORMAT(10, DEP/PAX TRIP (FLIGHT BASIS))
96 FORMAT(10, PASSENGER TRIP CIRCULARITY FACTOR)
97 FORMAT(10, COACH LOAD FACTOR)
100 FORMAT(10, FIRST CLASS ADAD FACTOR)
101 FORMAT(10, FLYING OPERATIONS(LESS RENTALS) EXP
235 102 FORMAT(10, MAINTENANCE EXPENSE FLY EQUIPMENT
103 FORMAT(10, RENTALS FLT EQUIP
104 FORMAT(10, PASSENGER IN FLY EXPENSE)
105 STEPS
107 FORMAT(10, FOOD
108 FORMAT(10, OTHER
109 FORMAT(10, AIRCRAFT SERVICING EXP 0/
110 LINE SERVICE
111 FORMAT(10, CONTROL
112 FORMAT(10, LANDING FEES
245 113 FORMAT(10, TOTAL AIRCRAFT SERVICING
114 FORMAT(10, TRAFFIC SERVICING EXP 0/
115 PAX - BAGGAGE
116 CARGO
117 CARGO
118 TOTAL TRAFFIC SERVICING
119 TOTAL TRAFFIC SERVICING
120 TOTAL TRAFFIC SERVICING
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SUBROUTINE INDIR 76776 OPT-1 PAGE 6

FTN 4.6+469 34/26/79 09.41.54

20 RETURN
END

INDIR 287
INDIR 288

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1      SUBROUTINE REPAY
C
C      CALCULATE REPAYMENT SCHEDULE FOR EACH AIRCRAFT
C
5      DIMENSION AREV(25,100),CF(25),COST(25,100),
10      1DCF(25),DEPR(25,100),EBIAT(25,100),EAT(25,100),ECLIFE(103),
15      2INCTAX(25,100),ININ(25,100),MTEARM(25,100),OPCOST(25,100),
20      3PRIN(25,100),PV(25),RES(103),RATE(100),SALVAG(25,100),
25      4STDEP(25,100),YTRAST(25,100),
30      5SAREV(25),SSALV(25),SINTIN(25),SOPCST(25),SDEPR(25),
35      6SSTDEP(25),SEBIAT(25),SYNTR(25),SEBT(25),SINCTR(25),SINTERM(25),
40      7DEPRE(25),PRICE(25,100),CSOCF(25),
45      8SEXP(25),SINCOM(25),CG(25,100),BVALUE(25,100),SCG(25),
50      9CGTX(25,100),SCGT(25),YRPPAY(25,100)
55      LEVEL 2,
60      X AREV, CF, CSAREV, CSCF, CSOCT, CSDEPR,
65      1CSEBIT, CSEBT, CSINTI, CSINTX, CSOCT, CSOCT, CSOCT,
70      2CSYNTR, DCF, DEPR, DEPRE, EBIAT, EBT, ECLIFE,
75      3I, INCTAX, ININ, IYEAR, MAC, MTEARM, OPCOST,
80      4PRIN, PV, R, SALVAG, SAREV, SDEPR, SEBIAT,
85      5SEBT, SINCTX, SINTIN, SINTERM, SOPCST, SPRIN,
90      6SSTDEP, STDEP, SYNTR, YTRAST,
95      7COST, PRICE, RES, RATE, THRATE,
100      8BVALUE
105      LEVEL 2, YRPPAY
110      COMMON/IN/AREV, CF, CSAREV, CSCF, CSOCT, CSOCT, CSOCT,
115      1CSEBIT, CSEBT, CSINTI, CSINTX, CSOCT, CSOCT, CSOCT,
120      2CSYNTR, DCF, DEPR, DEPRE, EBIAT, EBT, ECLIFE,
125      3I, INCTAX, ININ, IYEAR, MAC, MTEARM, OPCOST,
130      4PRIN, PV, R, SALVAG, SAREV, SDEPR, SEBIAT,
135      5SEBT, SINCTX, SINTIN, SINTERM, SOPCST, SPRIN,
140      6SSTDEP, STDEP, SYNTR, YTRAST,
145      7COST, PRICE, RES, RATE, THRATE,
150      8BVALUE
155      COMMON/YRPPAY/YRPPAY
160      REAL INTIN,INCTAX,MTEARM
165      MAIN LOOP POINT FOR EACH AIRCRAFT - SKIP PROCESSING WITHIN LOOP
170      EXCEPT FOR FIRST YEAR CRAFT IS PURCHASED
175      DO 90 J=1,MAC
180      IF (COST(I,J) .LE.0.0) GO TO 90
185      PCOST=COST(I,J)
190      CALCULATE TOTAL ANNUAL PAYMENTS (PRINCIPAL PLUS INTEREST)
195      YRPPAY(I,J)=(PCOST*RATE(J))*((1+RATE(J))**ECLIFE(J))/
200      1*((1+RATE(J))**ECLIFE(J))-1)
205      MLIFE=ECLIFE(J)
210      NM=1+MLIFE-1
215      DO 90 M=1,MN
220      CALCULATE INTEREST AND PRINCIPAL PAYMENT SCHEDULE
225      YNTRST(M,J) = PCOST*RATE(J)
230      PRIN(M,J) = YRPPAY(I,J) - YNTRST(M,J)
235      PCOST=PCOST-PRIN(M,J)

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SUBROUTINE REPA		76/76	OPT-1	FTN 4-6+463	04/26/79	09.41.54	PAGE	2
60								
	50	CONTINUE						
	C				REPA	59		
	C	END OF MAIN PROGRAM LOOP			REPA	60		
	C				REPA	61		
		90	CONTINUE		REPA	62		
			RETURN		REPA	63		
			END		REPA	64		
					REPA	65		

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FTM 4.6+400

76/76 OPT=1

SUBROUTINE DEPSUB

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CALCULATE DEPRECIATION SCHEDULE FOR EACH AIRCRAFT - DOUBLE
 DECLINING BALANCE METHOD FOR FIRST HALF OF ECONOMIC LIFE AND
 STRAIGHT LINE METHOD FOR BALANCE OF ECONOMIC LIFE

DIMENSION AREV(25,100),CF(25),COST(25,100),
 10CF(25),DEPR(25,100),EBIAT(25,100),EST(25,100),ECLIFE(100),
 2INVTAX(25,100),INTINV(25,100),NTEARM(25,100),OPCOST(25,100),
 3PRM(25,100),PV(25),RES(100),RRATE(100),SALVAGE(25,100),
 4STDEP(25,100),YNTAST(25,100),
 5SAREV(25),SSALVG(25),SYNTIN(25),SPRIM(25),SOPCST(25),SDEPR(25),
 6SSTOEP(25),SEBIAT(25),SYNTR(25),SEBT(25),SINCTN(25),SNTERM(25),
 7DEPRE(25),PRICE(25,100),CSDCF(25),
 8SERP(25),SINCOM(25),CG(25,100),BVALUE(25,100),SCG(25),
 9CGTX(25,100),SCGTN(25)

LEVEL 2,
 X
 1CSEBIT, AREV, CF, CSAREV, CSCF, CSNTRM, CSOCT, CSDEPR, CSDEPR,
 2CSYNTR, DCF, DEPR, DEPREC, EBIAT, EBT, ECLIFE, CSNTRM, CSNTRM,
 3I, INCNTX, INTNV, IYEAR, MAC, NTEARM, OPCOST, DEPR, DEPR,
 4PRIN, PV, R, SALVAG, SAREV, SDEPR, SEBIAT, DEPR, DEPR,
 5SEBT, SINCTX, SYNTIN, SNTERM, SNTERM, SNTERM, SNTERM, SNTERM,
 6SSTOEP, STDEP, SYNTR, SYNTR, SYNTR, SYNTR, SYNTR, SYNTR,
 7COST, PRICE, RES, RRATE, TRATE, TRATE, TRATE, TRATE, TRATE,
 8BVALUE
 COMMON/IN/AREV, CF, CSAREV, CSCF, CSNTRM, CSOCT, CSDEPR, CSDEPR,
 1CSEBIT, CSEBT, CSINTI, CSINTX, CSNTRM, CSOCT, CSNTRM, CSNTRM,
 2CSYNTR, DCF, DEPR, DEPREC, EBIAT, EBT, ECLIFE, CSNTRM, CSNTRM,
 3I, INCNTX, INTNV, IYEAR, MAC, NTEARM, OPCOST, DEPR, DEPR,
 4PRIN, PV, R, SALVAG, SAREV, SDEPR, SEBIAT, DEPR, DEPR,
 5SEBT, SINCTX, SYNTIN, SNTERM, SNTERM, SNTERM, SNTERM, SNTERM,
 6SSTOEP, STDEP, SYNTR, SYNTR, SYNTR, SYNTR, SYNTR, SYNTR,
 7COST, PRICE, RES, RRATE, TRATE, TRATE, TRATE, TRATE, TRATE,
 8BVALUE
 REAL INTNV,INCTAX,NTEARM

MAIN LOOP POINT FOR EACH AIRCRAFT - DO DOUBLE DECLINING BALANCE
 METHOD COMPUTATIONS FIRST

DO 92 J=1,NAC
 RATE = 2.0*(1-RES(J))/ECLIFE(J)
 IECLIF=ECLIFE(J)/2.
 BVALUE(1,J) = PRICE(1,J)
 IF(PRICE(1,J) < 0.0) GO TO 92.
 MM=ECLIFE(J)
 DO 20 K=1,MM
 II=K
 DEPR(II,J)=BVALUE(II,J)*RATE
 BVALUE(II+1,J)=BVALUE(II,J)-DEPR(II,J)
 LL=IECLIF+1
 DO 30 M=LL,MM
 DEPR(M,J)=0.0
 BVALUE(M,J)=BVALUE(M-1,J)
 CONTINUE
 30 CONTINUE

PERFORM STRAIGHT LINE METHOD CALCULATIONS


```

C
60      STLIFE=ECLIFE(J)-IECLIF,
C      ASSUME SALVAGE VALUE SAME AS FOR MLIFE
M=ECLIFE(J)-1
DO 75 L=1,M
    SALVAGE(L,J)=0.0
75      CONTINUE
65      MLIFE=ECLIFE(J)
    SALVAGE(MLIFE,J)=PRICE(1,J)*RES(J)
    I1=IECLIF+1
DO 55 M=I1,MLIFE
    STDEP(M,J)=(BVALUE(II,J)-SALVAGE(MLIFE,J))/STLIFE
    M=M+1
    BVALUE(M,J)=BVALUE(M,J)-STDEP(M,J)
55      CONTINUE
C      ASSUME STRAIGHT LINE DEPRECIATION IS ZERO UP TO II YEARS
DO 65 N=1,IECLIF
    STDEP(N,J)=0.
65      CONTINUE
C
C      END OF MAIN PROGRAM LOOP
92      CONTINUE
    RETURN
    END

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DEPSUB 59
DEPSUB 60
DEPSUB 61
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76776 OPT-1

SUBROUTINE NETSUB

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COMPUTE EARNINGS AND TAXES
 DIMENSION AREV(25,100),CF(25),COST(25,100),
 10CF(25),DEPR(25,100),EBIAT(25,100),EBT(25,100),ECLIFE(100),
 21INCTAX(25,100),INTINV(25,100),MTEARN(25,100),OPCOST(25,100),
 3PRIN(25,100),PV(25),RES(100),RRATE(100),SALVAG(25,100),
 4STDEP(25,100),YNTAST(25,100),
 5SAREV(25),SSALV(25),SININ(25),SPRIN(25),SOPCST(25),SDEPR(25),
 65STDEP(25),SEIAT(25),SYNTRN(25),SEBT(25),SINCTM(25),SINTERM(25),
 7DEPREC(25),PRICE(25,100),CSDCF(25),
 8SEXP(25),SINCON(25),CG(25,100),BVALUE(25,100),SCG(25),
 9CGTX(25,100),SCGTN(25)
 LEVEL 2,
 X
 10SEBIT, AREV, CF, CSAREV, CSCF, CSNTRN, CSDCF, CSDEPR, NETSUB
 20SYNTR, DCF, CSINTX, CSINTX, CSNTRN, CSOPCT, CSOPCT, CSOPCT, NETSUB
 31, INCTAX, INTINV, INTINV, EBIAT, EBT, ECLIFE, NETSUB
 4PRIN, PV, R, SALVAG, SAREV, MTEARN, OPCOST, NETSUB
 5SEBT, SINCTX, SININ, SININ, SAREV, SDEPR, SEIAT, NETSUB
 65STDEP, STDEP, SYNTR, SYNTR, SINTERM, SDEPR, SEIAT, NETSUB
 7COST, PRICE, RES, RATE, THRATE, THRATE, THRATE, NETSUB
 8BVALUE
 COMMON/IN/AREV, CF, CSAREV, CSCF, CSNTRN, CSDCF, CSDEPR, NETSUB
 10SEBIT, CSEBT, CSINTX, CSINTX, CSNTRN, CSOPCT, CSOPCT, CSOPCT, NETSUB
 20SYNTR, DCF, DEPR, DEPR, EBIAT, EBT, ECLIFE, NETSUB
 31, INCTAX, INTINV, INTINV, EBIAT, EBT, ECLIFE, NETSUB
 4PRIN, PV, R, SALVAG, SAREV, MTEARN, OPCOST, NETSUB
 5SEBT, SINCTX, SININ, SININ, SAREV, SDEPR, SEIAT, NETSUB
 65STDEP, STDEP, SYNTR, SYNTR, SINTERM, SDEPR, SEIAT, NETSUB
 7COST, PRICE, RES, RATE, THRATE, THRATE, THRATE, NETSUB
 8BVALUE
 REAL INTINV,INCTAX,MTEARN
 MAIN PROGRAM LOOP POINT FOR EACH AIRCRAFT - WRITE DEPRECIATION AND
 EARNINGS WHEN CALCULATED
 DO 94 J=1,NAC
 WRITE (6,53) DEPR(I,J)
 53 FORMAT (0 DEPR= 0,10E12.4)
 WRITE (6,61) STDEP(I,J)
 61 FORMAT (0 STDEP= 0,10E12.4)
 EBIAT(I,J)=AREV(I,J)-OPCOST(I,J)-DEPR(I,J)-STDEP(I,J)
 EBT(I,J)=EBIAT(I,J)-YNTAST(I,J)
 WRITE (6,20) EBIAT(I,J)
 20 FORMAT (0 EBIAT= 0,10E12.4)
 WRITE (6,30) EBT(I,J)
 30 FORMAT (0 EBT= 0,10E12.4)
 INCTAX(I,J)=(EBT(I,J)/THRATE)
 MTEARN(I,J)=EBT(I,J)-INCTAX(I,J)
 94 CONTINUE
 RETURN
 END

SUBROUTINE SUM

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FTM 4.6+60

SUBROUTINE SUM 76/76 OPT=1

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60      DO 20 N=1,IYEAR
          SSALVG(N)=0.0
          DO 20 J=1,MAC
              SSALVG(N)=SALVAG(N,J)+SSALVG(N)
          20 CONTINUE
          C
          C      INITIAL INVESTMENT
          C
          DO 30 N=1,IYEAR
              SINTIN(N)=0.0
              DO 30 J=1,MAC
                  SINTIN(N)=INTIN(N,J)+SINTIN(N)
              30 CONTINUE
          21      WRITE (6,21) (SINTIN(N),N=1,IYEAR)
              FORMAT(4 SINTIN=*,10E12.4)
              CSINTI=0.0
              DO 32 N=1,IYEAR
                  CSINTI=SINTIN(N)+CSINTI
              32 CONTINUE
          C
          C      PRINCIPAL PAYMENTS
          C
          DO 40 N=1,IYEAR
              SPRIN(N)=0.0
              DO 40 J=1,MAC
                  SPRIN(N)=PRIN(N,J)+SPRIN(N)
              40 CONTINUE
          42      CSPRIN=0.0
              DO 42 N=1,IYEAR
                  CSPRIN=SPRIN(N)+CSPRIN
              42 CONTINUE
          C
          C      OPERATING COSTS
          C
          DO 50 N=1,IYEAR
              SOPCST(N)=0.0
              DO 50 J=1,MAC
                  SOPCST(N)=OPCOST(N,J)+SOPCST(N)
              50 CONTINUE
          52      CSOPCT=0.0
              DO 52 N=1,IYEAR
                  CSOPCT=SOPCST(N)+CSOPCT
              52 CONTINUE
          C
          C      DEPRECIATION
          C
          DO 60 N=1,IYEAR
              SDEPR(N)=0.0
              DO 60 J=1,MAC
                  SDEPR(N)=DEPR(N,J)
              60 CONTINUE
          53      WRITE (6,53) DEPR(N,J)
              FORMAT(4 DEPR=*,10E12.4)
              SDEPR(N)=DEPR(N,J)+SDEPR(N)
          60 CONTINUE
          DO 70 N=1,IYEAR
              SSTDEP(N)=0.0
              DO 70 J=1,MAC
                  SSTDEP(N)=STDEP(N,J)
          70 CONTINUE

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SUBROUTINE SUM	76/76	OPT=1
115	61	FORMAT (* STDEP= *,10E12.4) SSTDEP(N)=STDEP(N,J)+SSTDEP(N) 70 CONTINUE DO 75 M=1,IYEAR DEPREC(N)=DOEP(N)+SSTDEP(N) WRITE (6,71) DEPREC(N) 71 FORMAT (* DEPREC= *,10E12.4) 75 CONTINUE CSOEPN=0.0 DO 77 M=1,IYEAR CSOEPN=DEPREC(M)+CSOEPN 77 CONTINUE C C C EARNINGS BEFORE INTEREST AND TAXES C
130		DO 80 M=1,IYEAR SEBINT(M)=0.0 DO 80 J=1,NAC SEBINT(M)=EBINT(M,J)+SEBINT(M) 80 CONTINUE CSEBIT=0.0 DO 82 M=1,IYEAR CSEBIT=SEBINT(M)+CSEBIT 82 CONTINUE C C C INTEREST PAYMENTS C
145	90	DO 90 M=1,IYEAR SYNTRS(M)=0.0 DO 90 J=1,NAC SYNTRS(M)=YNTNST(M,J)+SYNTRS(M) 90 CONTINUE CSYNTR=0.0 DO 92 M=1,IYEAR CSYNTR=SYNTRS(M)+CSYNTR 92 CONTINUE C C C EARNINGS BEFORE TAXES C
155	100	DO 100 M=1,IYEAR SEBT(M)=0.0 DO 100 J=1,NAC SEBT(M)=EBTN(J)+SEBT(M) 100 CONTINUE CSEBT=0.0 DO 102 M=1,IYEAR CSEBT=SEBT(M)+CSEBT 102 CONTINUE C C C INCOME TAX C
165		DO 110 M=1,IYEAR SINCTX(M)=0.0 DO 110 J=1,NAC SINCTX(M)=INCTAX(M,J)+SINCTX(M) 110 CONTINUE CSINCTX=0.0 C
170		

SUBROUTINE	SUM	76/76	DPT-1	FTN 4.6*460	04/26/79	09-41.54	PAGE	4
175	DO 112 N=1, IYEAR CSINTX=SIINCT(N)*CSINTX 112 CONTINUE C C NET EARNINGS				SUN	173		
180	DO 120 N=1, IYEAR SINTER(N)=0.0 DO 120 J=1, MAC SINTER(N)=N*EARN(N, J)*SINTER(N) 120 CONTINUE CSINTX=0.0				SUN	174		
185	DO 122 N=1, IYEAR CSINTX=SINTER(N)*CSINTX 122 CONTINUE RETURN END				SUN	175		
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1      SUBROUTINE CFSUB
C
C      COMPUTE THE CASHFLOW FOR EACH YEAR UNDER STUDY
C
5      DIMENSION AREV(25,100),CF(25),COST(25,100),
10     10CF(25),DEPR(25,100),EBIAT(25,100),EBT(25,100),ECLIFE(100),
15     2INCTAX(25,100),INTINV(25,100),NTEARN(25,100),OPCOST(25,100),
20     3PRIN(25,100),PV(25),RES(100),RATE(100),SALVAG(25,100),
25     4SDEP(25,100),YNTRST(25,100),
30     5SAREV(25),SSALV(25),SINTIN(25),SPRIN(25),SOPCST(25),SDEPR(25),
35     6SSSTDEP(25),SEBIAT(25),SYNTRS(25),SEBT(25),SINCTX(25),SINTER(25),
40     7DEPRECI(25),PRICE(25,100),CSOCF(25),
45     8SEXP(25),SINCOM(25),CG(25,100),NVALUE(25,100),SCG(25),
50     9CGTX(25,100),SCGTX(25)
55     LEVEL 2,
60     X
65     AREV, CF, CSOCF, CSDEPR,
70     1CSEBIT, CSEBT, CSINTI, CSMTX, CSOPCT, CSDEPR,
75     2CSYNTR, DCF, DEPR, CSINTX, CSMTX, CSOPCT, CSDEPR,
80     3I, INCTAX, INTINV, IYEAR, NTEARN, ECLIFE,
85     4PRIN, PV, R, SALVAG, SAREV, SDEPR, SEBIAT,
90     5SSEBT, SINCTX, SINTIN, SYNTRS, SOPCST, SPRIN, SSALV,
95     6SSSTDEP, SDEP, SYNTRS, YNTRST, TXRATE,
100    7COST, PRICE, RES, TXRATE,
105    8BVALUE
110    COMMON/IN/AREV, CF, CSOCF, CSDEPR,
115    1CSEBIT, CSEBT, CSINTI, CSMTX, CSOPCT, CSDEPR,
120    2CSYNTR, DCF, DEPR, CSINTX, CSMTX, CSOPCT, CSDEPR,
125    3I, INCTAX, INTINV, IYEAR, NTEARN, ECLIFE,
130    4PRIN, PV, R, SALVAG, SAREV, SDEPR, SEBIAT,
135    5SSEBT, SINCTX, SINTIN, SYNTRS, SOPCST, SPRIN, SSALV,
140    6SSSTDEP, SDEP, SYNTRS, YNTRST, TXRATE,
145    7COST, PRICE, RES, TXRATE,
150    8BVALUE
155    REAL INTINV,INCTAX,NTEARN
160    CALCULATE CASHFLOW FOR EACH YEAR
165    DO 50 N=1,IYEAR
170    CF(N)=SAREV(N)+SSALV(N)-SINTIN(N)-SPRIN(N)-SOPCST(N)
175    1-SYNTRS(N)-SINCTX(N)
180    50 CONTINUE
185    WRITE CALCULATED CASHFLOW
190    WRITE (6,60) (CF(N),N=1,IYEAR)
195    FORMAT(' CASH FLOW = ',10E12.4)
200    CSCF=0.0
205    COMPUTE CUMULATIVE TOTAL CASHFLOW
210    DO 100 N=1,IYEAR
215    CSCF=CSCF+CF(N)+CSCF
220    100 CONTINUE
225    RETURN
230    ENO

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1      SUBROUTINE DCFSUB
2
3      CASHFLOW
4      DIMENSION AREV(25,100),CF(25),COST(25,100),
5      1DCF(25),DEPR(25,100),EBIAT(25,100),EBT(25,100),ECLIFE(100),
6      2INCTX(25,100),INTINV(25,100),NTEARN(25,100),OPCOST(25,100),
7      3PRIN(25,100),PV(25),RES(100),RRATE(100),SALVAG(25,100),
8      4SDEPR(25,100),YNTRST(25,100),
9      5SAREV(25),SSALVAG(25),SENTIN(25),SOPRIN(25),SOPCST(25),SDEPR(25),
10     6SSTDEPR(25),SEBIAT(25),SYNTRIS(25),SEBT(25),SINCTX(25),SNTEARN(25),
11     7DEPRECI(25),PRICE(25,100),CSDCF(25),
12     8SEXP(25),SINCOM(25),CG(25,100),BVALUE(25,100),SCG(25),
13     9CGTX(25,100),SCGTX(25)
14     LEVEL 2.
15
16     X AREV, CF, CSAREV, CSINTX, CSDEPR, CSOPCT, CSOPCT, CSDEPR,
17     1CSEBT, CSEBT, CSINTI, CSINTX, CSINTX, CSINTX, CSINTX, CSINTX, CSINTX,
18     2CSYNTR, DCF, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR,
19     3I, INCTX, INTINV, INTINV, INTINV, INTINV, INTINV, INTINV, INTINV, INTINV,
20     4PRIN, PV, A, SALVAG, SAREV, SAREV, SAREV, SAREV, SAREV, SAREV,
21     5SSEBT, SINCTX, SENTIN, SYNTRIS, SYNTRIS, SYNTRIS, SYNTRIS, SYNTRIS, SYNTRIS,
22     6SSTDEPR, STDEP, STDEP, STDEP, STDEP, STDEP, STDEP, STDEP, STDEP, STDEP,
23     7COST, PRICE, RES, RATE, TRATE, TRATE, TRATE, TRATE, TRATE, TRATE,
24     8BVALUE
25     COMMON/IM/AREV, CF, CSAREV, CSINTX, CSINTX, CSINTX, CSINTX, CSINTX, CSINTX,
26     1CSEBT, CSEBT, CSINTI, CSINTX, CSINTX, CSINTX, CSINTX, CSINTX, CSINTX,
27     2CSYNTR, DCF, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR, DEPR,
28     3I, INCTX, INTINV, INTINV, INTINV, INTINV, INTINV, INTINV, INTINV, INTINV,
29     4PRIN, PV, A, SALVAG, SAREV, SAREV, SAREV, SAREV, SAREV, SAREV,
30     5SSEBT, SINCTX, SENTIN, SYNTRIS, SYNTRIS, SYNTRIS, SYNTRIS, SYNTRIS, SYNTRIS,
31     6SSTDEPR, STDEP, STDEP, STDEP, STDEP, STDEP, STDEP, STDEP, STDEP, STDEP,
32     7COST, PRICE, RES, RATE, TRATE, TRATE, TRATE, TRATE, TRATE, TRATE,
33     8BVALUE
34     REAL INTINV, INCTX, NTEARN
35
36     INITIALIZE LOOP AND TEST PARAMETERS
37
38     REAL K,KK,KKK
39     KK=0.
40     KKK=0.
41     RR=0.
42     RRR=0.
43     ICNT=0
44
45     SET STARTING VALUE FOR R
46
47     IF (ICSCF.LT.0.0) GO TO 53
48     R=0.01
49     GO TO 40
50     R=-0.01
51
52     MAIN ITERATION CYCLE POINT - CALCULATE DISCOUNTED CASHFLOW SUM
53
54     DO 49 A=1,IYEAR
55     48 PV(M)=(1.+R)**(-M)
56     SUM=CF(1)
57
58     DCFSUB
59     DCFSUB
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SUBROUTINE DCFSUB 76/75 OPT=1

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        DCFI(1)=CF(1)
        DO 500 N=2,IYEAR
        DCFI(N)=CFI(1)+PV(N-1)
        SUM=SUM+DCF(N)
        500 CONTINUE
        70 K=ABS(SUM)-9000.
        C
        C IF CURRENT AND SECOND TO LAST SUMS ARE THE SAME, VALUES NEEDED
        C HAVE BEEN FOUND
        C
        IF(KK.EQ.K) GO TO 10
        KK=KK
        RRR=RR
        KK=K
        RR=R
        ICNT=ICNT+1
        C
        C WRITE VALUES OF R AND K EVERY TENTH ITERATION CYCLE
        C
        IF(MOD(ICNT,10).NE.0) GO TO 75
        WRITE(6,101) R,K
        75 CONTINUE
        C
        C VALUES NEEDED HAVE BEEN FOUND IF K IS NEAR ZERO OR NEGATIVE
        C
        IF(K.LE.1.E-32) GO TO 20
        C
        C INCREMENT R FOR NEXT ITERATION CYCLE
        C
        IF (SUM) 60,20,30
        ASSUME DELTA R
        R=R-0.001
        GO TO 80
        30 R=R+0.001
        C
        C TRANSFER TO PERFORM NEXT ITERATION CYCLE
        C
        80 GO TO 40
        C
        C SELECT PROPER VALUE FOR R WHEN DONE
        C
        10 CONTINUE
        IF(KK=KK) 120,130,130
        120 R=RR
        GO TO 20
        130 R=RR
        C
        C CALCULATE DISCOUNTED CASHFLOW SUM AND WRITE FINAL VALUES FOR
        C R AND K
        C
        20 CSDCF(1)=CF(1)
        DO 100 N=2,IYEAR
        CSDCF(N)=DCF(N)+CSDCF(N-1)
        100 CONTINUE
        WRITE(6,101) R,K
        101 FORMAT(2F20.7)
        RETURN

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SUBROUTINE DCFSUB 7676 OPT-1

115

END

PTN 4.6+460

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PAGE 3

DCFSUB 116


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105 FORMAT(1H0,'TOTALS',4X,F13.2,3X,F13.2,2X,F13.2,3X,F13.2,3X,F13.2,3X,F13.2,
13X,F13.2,3X,F13.2)
C
C  WRITE SECOND OUTPUT PAGE
C
C  WRITE (6,200)
200 FORMAT(1H1, 4X,0X,'EARNINGS',10X,'INCOME',11X,'NET',11X,
1X,'NET CASH',5X,'DISCOUNT',6X,'DISCOUNTED',16X,'NPV')
WRITE (6,201)
201 FORMAT(1H, 11X,'BEFORE TAX',11X,'TAX',10X,'EARNINGS',10X,'FLOW',
10X,'FACTOR',8X,'CASH FLOW')
WRITE (6,202) A
202 FORMAT(1H, 14X,'(S)',13X,'(S)',12X,'(S)',13X,'(S)',7X,
1X,'(S)',14X,'(S)',20X,'(S)')
N=1
WRITE(6,309) N,SEBT(M),SINCTX(M),SNTERM(M),CF(M),OCF(M),CSDCF(M)
309 FORMAT(1H0,14,F18.2,3F16.2,3X,9H 1.0000,F17.2,F22.2)
DO 30 N=2,1YEAR
NN=N-1
WRITE(6,400) N,SEBT(M),SINCTX(M),SNTERM(M),CF(M),PV(MN),OCF(M),
1CSDCF(M)
400 FORMAT(1H0,14,6X,E12.2,3F16.2,F12.4,F17.2,F22.2)
30 CONTINUE
WRITE (6,204)
204 FORMAT(1H0,10X,12(1H-),4X,12(1H-),4X,12(1H-),4X,12(1H-),17X,
112(1H-))
WRITE (6,205) SEBT,CSINTX,CSNTAM,CS,F,CSDCF(1YEAR)
205 FORMAT(1H0,'TOTALS',3X,E13.2,F16.2,3X,E13.2,F16.2,12X,F17.2)
RETURN
END

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1      SUBROUTINE TAX
C
C      CALCULATE EFFECTS OF LOSS CARRIED BACKWARDS AND FORWARDS.
C      ALSO CAPITAL GAINS TREATMENT
C
5      DIMENSION AREV(25,100),CF(25),COST(25,100),
10      1OCF(25),DEPR(25,100),EBIAT(25,100),EBT(25,100),ECLIFE(100),
15      2INCTAX(25,100),INTINV(25,100),NTEARN(25,100),OPCOST(25,100),
20      3PRIM(25,100),PV(25),RES(100),RATE(100),SALVAG(25,100),
25      4STDEP(25,100),VINTRST(25,100),
30      5SAREV(25),SSALVAG(25),SINTIN(25),SPRIN(25),SOPCST(25),SDEPR(25),
35      6STDEP(25),SEBIAT(25),SYNTRS(25),SEBT(25),SINCTX(25),SINTERN(25),
40      7DEPREC(25),PRICE(25,100),CSOCP(25),
45      8SEXP(25),SINCOM(25),CG(25,100),BVALUE(25,100),SCG(25),
50      9CGTX(25,100),SCGTX(25)
55      LEVEL 2,
X      AREV, CF, CSAREV, CSCF, CSOCP, CSDEPR,
1      CSEBT, CSINTI, CSINTX, CSMTN, CSOCT, CSPRIN,
2      DCF, DEPR, DEPREC, EBIAT, EBT, ECLIFE,
3      INCTAX, INTINV, IYEAR, MAC, NTEARN, OPCOST,
4      PV, R, SALVAG, SAREV, SDEPR, SEBIAT,
5      SINCTX, SINTIN, SINTERN, SOPCST, SPRIN, SSALVAG,
6      STDEP, SYNTRS, VINTRST, TXRATE,
7      TCOST, PRICE, RES,
8      BVALUE
9      LEVEL 2,CG,CGTX
10     COMMON/IN/AREV,
15     1CSEBT, CSEBT, CSINTI, CSCF, CSOCP, CSDEPR,
20     2CSYNTR, DCF, DEPR, DEPREC, CSINTX, CSMTN, CSOCT, CSPRIN,
25     3, INCTAX, INTINV, IYEAR, MAC, NTEARN, OPCOST,
30     4PRIN, PV, R, SALVAG, SAREV, SDEPR, SEBIAT,
35     5SEBT, SINCTX, SINTIN, SINTERN, SOPCST, SPRIN,
40     6STDEP, STDEP, SYNTRS, VINTRST, TXRATE,
45     7COST, PRICE, RES,
50     BVALUE
55     COMMON/CG/CG
60     COMMON/CGTX/CGTX
65     REAL INTINV,INCTAX,NTEARN
70     EVALUATING LOSSES AND CARRYING THEM FORWARD AND BACKWARDS,
75     DETERMINING CAPITAL GAINS TAX AND READJUSTING NET EARNINGS,
80     AND INCOME TAX APPROPRIATELY
85     PARAMETER INITIALIZATION
90     DD 20, N=1, IYEAR
95     SEMP(N)=SOPCST(N)+DEPREC(N)+SYNTRS(N)
100    SINCOM SARE AS SEBT
105    SINCOM(N)=SAREV(N)-SEXP(N)
110    CONTINUE
115    20
120    C
125    C      START OF LOOP TO CALCULATE LOSS CARRIED BACKWARDS AND FORWARD.
130    C      ROUTINE STOPS IF A YEAR WITH NET POSITIVE INCOME IS FOUND FOR
135    C      EACH AIRCRAFT
140    C
145    DD 400 N=1, IYEAR
150    IF(SINCOM(N).GT.0.0) GO TO 300
155    ALOSS=SINCOM(N)

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	C	WE CAN CARRY THE LOSS BACK 3 YEARS - FORWARD 5 YEARS AND MODIFY SEBT	TAX
60	C	IFIN.LT.2) GO TO 200	TAX
	C	IFIN.EQ.2) GO TO 25	TAX
		IFIN.EQ.3) GO TO 35	TAX
		IFIN.GE.4) GO TO 45	TAX
65	C	SPECIAL CARRY BACK CASE - YEAR=2	TAX
	C	K=N-1	TAX
	25	IF (SINCOM(K).LE.0.0) GO TO 14	TAX
		SINCOM(K)-SINCOM(K)+ALOSS	TAX
		ALOSS-SINCOM(K)	TAX
70	14	IF(ALOSS-GE.0.0)GO TO 300	TAX
		IF(ALOSS-LT.0.0) GO TO 200	TAX
75	C	SPECIAL CARRY BACK CASE - YEAR=3	TAX
	C	M=N-2	TAX
	35	L=M-1	TAX
		DO 90 K=M,L	TAX
		IF (SINCOM(K).LE.0.0) GO TO 1000	TAX
		SINCOM(K)-SINCOM(K)+ALOSS	TAX
		ALOSS-SINCOM(K)	TAX
80	1000	IF(ALOSS-GE.0.0)GO TO 300	TAX
	50	CONTINUE	TAX
		IF(ALOSS-LT.0.0) GO TO 200	TAX
85	C	GENERAL CARRY BACKWARD CASE	TAX
	C		TAX
	45	M=N-3	TAX
		L=M-1	TAX
		DO 90 K=M,L	TAX
		IF (SINCOM(K).LE.0.0) GO TO 2000	TAX
		SINCOM(K)-SINCOM(K)+ALOSS	TAX
		ALOSS-SINCOM(K)	TAX
95	2000	IF(ALOSS-GE.0.0)GO TO 300	TAX
	60	CONTINUE	TAX
		IF(ALOSS-LT.0.0) GO TO 200	TAX
	C	LOSS CARRY FORWARD PORTION	TAX
	C		TAX
	200	M=N+1	TAX
100		IF(M-15)16,17,17	TAX
	16	L=M+5	TAX
		IF(L-15)18,19,19	TAX
	17	M=L+YEAR	TAX
	19	L=L+YEAR	TAX
	18	DO 70 K=M,L	TAX
		IF(SINCOM(K).LE.0.0) CJ TO 70	TAX
		SINCOM(K)-SINCOM(K)+ALOSS	TAX
		ALOSS-SINCOM(K)	TAX
		IF(SINCOM(K)-LT.0.0) SINCOM(K)=0.0	TAX
	70	CONTINUE	TAX
	400	CONTINUE	TAX
110	C	LOSS TREATMENT COMPLETED - UPDATE EARNINGS AND TAXES	TAX
	C		TAX

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SUBROUTINE TAX 76/76 DP:01

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115      C      300      DO 90 N=1,IYEAR
      SEBT(N)=SINCOM(N)
      SINCTX(N)=0.0
      CONTINUE
120      80      DO 90 N=1,IYEAR
      SINTERM(N)=SEBT(N)
      IF (SEBT(N).LT.0.0) GO TO 90
      SINTERM(N)=SEBT(N)-SEBT(N)*XRATE
      SINCTX(N)=SINCTX(N)+SEBT(N)*XRATE
      CONTINUE
125      90      DO 99 L=2,IYEAR
      M=L-1
      DEPREC(L)=DEPREC(L)+DEPREC(M)
      CONTINUE
130      99      CALCULATE CAPITAL GAINS TREATMENT
      C
      C      LONG TERM CAPITAL GAINS TAX
      C      DO 11 J=L,MAC
      M=ECLIFE(J)-1
      DO 75 L=1,M
      C      ASSUME SALVAGE VALUE ZERO
      SALVAG(L,J)=0.0
      CONTINUE
140      75      M=ECLIFE(J)
      SALVAG(L,J)=PRICE(L,J)*RES(J)
      DO 11 M=1,IYEAR
      SSALVG(N)=0.0
      SSALVG(N)=SALVAG(M,J)+SSALVG(N)
      CONTINUE
145      11      CAPITAL GAINS=CG
      C      77      DO 111 L=1,IYEAR
      DO 111 J=L,MAC
      CG(L,J)=0.0
      CG(IYEAR,J)=PRICE(L,J)*RES(J)
      CGTX(L,J)=0.0
      IF (CG(L,J).LE.0.0) GO TO 333
      IF (CG(L,J).GE.25000.0) GO TO 65
      IF (CG(L,J).LT.25000.0) GO TO 44
      CGTAX=0.30
      66      GO TO 3
      44      CGTAX=0.22
      GO TO 3
      333      CGTAX=0.0
      3      CGTX(L,J)=CGTAX*CG(L,J)
      CG(L,J)=CG(L,J)-CGTX(L,J)
      CONTINUE
150      111      C
      C      UPDATE CUMULATIVE SUMS
      C
      C      DO 90 N=1,IYEAR
      SCG(N)=0.0
      SCGTX(N)=0.0
      DO 90 J=L,MAC
      WRITE(6,8) CG(N,J)
      FORMAT(8,8) CG(N,J) E12.4)
170      8

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TAX 171
TAX 172

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76/76 OPT=1

SUBROUTINE TAX

```

175      SCG(N)=CG(N,J)+SCG(N)
          SCGTX(N)=CGTX(N,J)+SCGTX(N)
          CONTINUE
          DO 95 N=1,IYEAR
180      SINTER(N)=SINTER(N)+SCG(N)
          SINTX(N)=SINTX(N)+SCGTX(N)
          CONTINUE
          CSEBT=0.0
          DO 102 N=1,IYEAR
185      CSEBT=CSEBT(N)+CSEBT
          CSNTAN=0.0
          DO 122 N=1,IYEAR
          CSMTAN=SMTER(N)+CSYTR4
190      CSINTX=0.0
          CSINTX=CSINTX(N)+CSINTX
          CONTINUE
          RETURN
          END

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TAX 192
TAX 193

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Appendix F

COSMIC SOFTWARE SUBMITTAL INFORMATION

Appendix F

COSMIC SOFTWARE SUBMITTAL INFORMATION

Abstract

The Analysis of the Benefits and Costs of Aeronautical Research and Technology (ABC-ART) models have been developed by NASA for use in analyzing the economic feasibility of applying advanced aeronautical technology to future civil aircraft. The methodology is composed of three major modules: Fleet Accounting Module, Airframe Manufacturer Module, and Air Carrier Module.

The Fleet Accounting Module is used to estimate the number of new aircraft required as a function of time to meet demand. This estimation is based primarily upon the expected retirement age of existing aircraft and the expected change in revenue passenger miles demanded. Fuel consumption estimates are also generated by this module. The Airframe Manufacturer Module is used to analyze the feasibility of manufacturing the new aircraft demanded. The module includes logic for production scheduling and estimating manufacturing costs. For a series of aircraft selling prices, a cash flow analysis is performed and a rate of return on investment is calculated. The Air Carrier Module provides a tool for analyzing the financial feasibility of an airline purchasing and operating the new aircraft. This module includes a methodology for computing the air carrier direct and indirect operating costs, performing a cash flow analysis, and estimating the internal rate of return on investment for a set of aircraft purchase prices.

The ABC-ART models are exercised in two distinct job steps. The Fleet Accounting Module and Airframe Manufacturer Module are run in one job step. The Air Carrier Module is run in a second job step. The modules are

designed for batch processing. Hardware requirements include a card reader, a printer, and disk storage. In addition, plotter output is generated by the Fleet Accounting Module. The software is currently programmed for a ZETA 230 plotter. The models have been run on a CDC 7600 with the SCOPE 2.1.3 operating system. CPU time is very modest for either job step. However, core requirements are substantial. The first job step requires 132K octal words, the second requires 154K octal words small core memory plus 144K octal words of large core memory.

Method of Solution

The methodology embedded in the ABC-ART models is described in detail in Volume I of this report (NASA CR-152278). The methodology for the Fleet Accounting, Airframe Manufacturer, and Air Carrier Modules are described in Sections II, III, and IV, respectively.

Computer Configuration Required

The ABC-ART models have been run on the NASA AMES CDC 7600, with the SCOPE 2.1.3 operating system. The only required input device is a card reader. Required output devices include a printer, a plotter, and disk storage. The job step which exercises the Fleet Accounting and Airframe Manufacturer Modules requires a printer, a plotter (software is provided for the ZETA 230 plotter), and four output disk files, two for storage of data generated for the plotter and two for temporary storage of input data. The job step that exercises the Air Carrier Module requires only a printer and a card reader.

Memory Required

On the CDC 7600 there are two types of memory, small core memory (SCM) and large core memory (LCM)*. The Fleet Accounting and Airframe Manufacturer Modules require 132K octal words of SCM and no LCM. The Air Carrier Module requires 154K octal words of SCM and 144K words of LCM to run. Variables were placed in LCM using the LEVEL 2 conversion for the CDC 7600.

Source Language

The ABC-ART models are programmed in CDC FORTRAN extended 4 language. The reference manual for this language is CDC Publication Number 84000009. The ZETA 230 plotter software is written in the same language, except for a single COMPASS routine (less than 1% of the code). COMPASS can be compiled using the CDC FORTRAN compiler.

User Instructions

Instructions for use of ABC-ART models are provided in Volume I, Section V of this report where a sample run of the models is described.

Implementation Instructions

The software package is being submitted in the five data sets. Each data set consists of a single tape file except for the fifth data set, which is composed of four tape files. These files are described below:

* On the NASA Ames computer system, 160K octal words of SCM and 1,200K octal words of LCM are available.

<u>Data Sets</u>	<u>Tape File</u>	<u>Contents</u>
1	1	FORTTRAN source code for the Fleet Accounting and Airframe Manufacturer Modules (see Appendices C and D).
2	2	FORTTRAN and COMPASS source code for the ZETA 230 plotter software.
3	3	FORTTRAN source code for the Air Carrier Module (see Appendix E).
4	4	Data input for the sample run of the Air Carrier Module (see Volume I, Table 41).
5	5,6,7,8	Data input for sample run of the Fleet Accounting and Airframe Manufacturer Modules (see Volume I, Table 40).

The tape containing these files is an unlabeled, 9-track tape in the EBCDIC character set with a density of 1600 BPI. The logical records are in card image format, 80 characters in length with no blocking. The tape was written on a CDC 7600 computer in SCOPE stranger format (i.e., F=S). No job control language is included in any of the files.

Program Timing

The CPU time required to run the ABC-ART models is quite modest. It will vary with the problem analyzed, the main factor being the number of new aircraft types. The sample run of the ABC-ART models, described in Volume I, Section V, took 22 and 4 CPU seconds for the Fleet Accounting and Airframe Manufacturer modules and Air Carrier Module, respectively. This time includes 5 and 3 CPU seconds, respectively, for compilation.

Accuracy of Results

The ABC-ART system does not contain mathematical or statistical routines that are very sensitive to word size. Single precision calculations on the CDC 7600 use a 60-bit word which provides a high degree of accuracy in computed results.

Sample Input and Output

Volume I of this report provides sample input and resulting output. In Section V, the sample problem is described, including a listing of the card input. Excerpts from the resulting printed and plotted output for the Fleet Accounting, Airframe Manufacturer and Air Carrier Modules are provided in Sections II, III, and IV, respectively.

Flowcharts

Flowcharts of the logic of each of the routines in the ABC-ART models have been provided in Volume I of this report. Flowcharts for the Fleet Accounting, Airframe Manufacturer, and Air Carrier Modules are provided in Sections II, III, and IV, respectively.

Program Listing

Listings of the FORTRAN code for the ABC-ART models are provided in Volume II of this report. Program listings for the Fleet Accounting, Airframe Manufacturer, and Air Carrier Modules are provided in Appendices C, D, and E, respectively.